

Relaxation method based solvers for multifluid flows

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Headlines

1 Introduction

2 Mathematical models

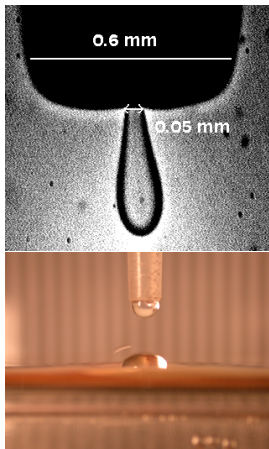
3 Numerical strategy

4 Parallel computing

5 Conclusion

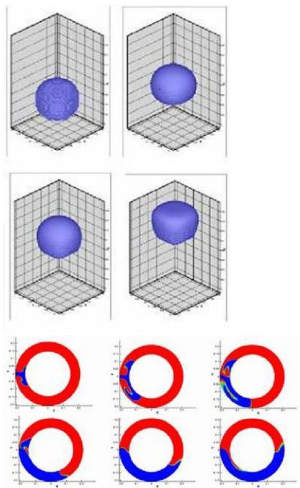
Physical context

- Two fluids
 - liquid or gas
 - different physical properties (EOS)
 - non-miscible fluids, separated by interfaces
 - surface tension force
 - viscous fluids
- Compressibility
 - acoustic wave propagation (interface deformation)
 - fluid volume variation
- Low Mach flows
 - low fluid velocities
 - pressure fluctuations almost negligible



Numerical context : FLUIDBOX

- 2D-3D
- unstructured meshes
- moving meshes
- finite volume and finite difference element methods
- cell vertex, cell centered formulation
- explicit and implicit formulation
- high order MUSCL technique
- Runge Kutta schemes
- upwind downwind triangles
- parallel computing



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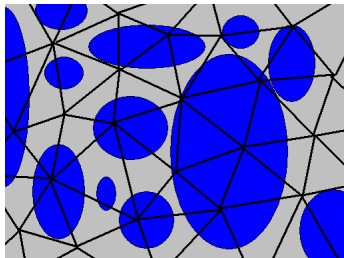
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Fluids unknowns

- volume fraction $\alpha_k = \frac{V_k}{V} \in [0, 1]$
- mass fraction $y_k = \frac{m_k}{m} \in [0, 1]$
- density $\rho_k = \frac{m_k}{V_k}$
- velocity u_k
- internal energy ϵ_k
- energy $E_k = \rho_k \left(\epsilon_k + \frac{1}{2} |u_k|^2 \right) > 0$
- enthalpy $H_k = \frac{E_k + p_k}{\rho_k}$
- temperature T_k
- entropy S_k
- sound speed c_k



Reduced model : Kapila(2000)

- Baer Nunziato model : u_k, p_k , velocity and pressure relaxation procedure
- asymptotic expansion : $u_k = u + \epsilon u'_k, p_k = p + \epsilon p'_k$

$$\begin{cases} \partial_t(\alpha_k) + u\partial_x(\alpha_k) + \beta_k\partial_x(u) = 0 \\ \partial_t(\mathbf{w}) + \partial_x(\mathbf{f}) = 0 \end{cases}$$

$$\mathbf{w} = \begin{pmatrix} \alpha_k \rho_k \\ \rho u \\ E \end{pmatrix}, \quad \mathbf{f} = \begin{pmatrix} \alpha_k \rho_k u \\ \rho u^2 + p \\ (E + p)u \end{pmatrix}, \quad \mathbf{v} = \begin{pmatrix} \rho_k \\ u \\ p \end{pmatrix}$$

- mixture density $\rho = \sum_k \alpha_k \rho_k$

- compaction term $\beta_k = \alpha_k \alpha_{k'} \frac{\rho_k c_k^2 - \rho_{k'} c_{k'}^2}{\alpha_{k'} \rho_k c_k^2 + \alpha_k \rho_{k'} c_{k'}^2}$

- sound speed $\frac{1}{\rho c^2} = \sum_k \frac{\alpha_k}{\rho_k c_k^2}$

- entropy $S = \sum_k y_k S_k$

'Stiffened gas' equation of states

- pressure : $p + \gamma P^\infty = (\gamma - 1)\rho(\epsilon - \epsilon^\infty)$

$$\frac{1}{\gamma - 1} = \sum_k \frac{\alpha_k}{\gamma - 1}, \quad \frac{\gamma P^\infty}{\gamma - 1} = \sum_k \frac{\alpha_k \gamma_k P_k^\infty}{\gamma_k - 1}, \quad \epsilon^\infty = \sum_k y_k \left(\epsilon_{k0} - \frac{p_{k0} + \gamma_k P_k^\infty}{\rho_{k0}(\gamma_k - 1)} \right)$$

- temperature : $p + p_k^\infty = (\gamma_k - 1)\rho_k^{\gamma_k} C_{vk} \left(\frac{T_k}{\rho_k^{\gamma_k - 1}} - \frac{T_k^\infty}{\rho_{k0}^{\gamma_k - 1}} \right)$

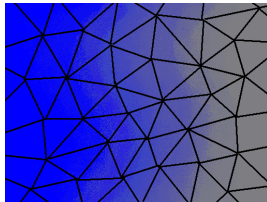
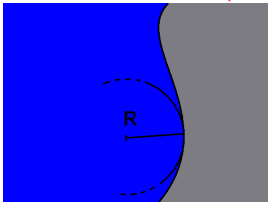
$$T_k^\infty = T_{k0} - \frac{p_{k0} + p_k^\infty}{\rho_{k0}(\gamma_k - 1)C_{vk}}$$

- sound speed : $\rho_k c_k^2 = \gamma_k (p + p_k^\infty)$, entropy : $S_k = (p + p_k^\infty)/(\rho_k^{\gamma_k})$

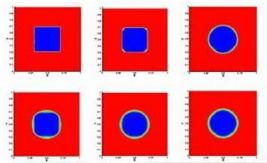
	γ	P^∞	ρ_0	P_0	ϵ_0	T_0	C_v
air	1.4	0	50	$1.0 \cdot 10^5$	$2.0 \cdot 10^5$	300	$1.0 \cdot 10^3$
water	4.4	$6.0 \cdot 10^8$	$1.0 \cdot 10^3$	$1.0 \cdot 10^5$	$6.17 \cdot 10^5$	220	$4.18 \cdot 10^3$

Capillary effects

Continuum Surface Force (CSF, Brackbill 1992)



- ϕ_k phase field α_k, y_k
- $n = \nabla \phi_k$ normal at the interface
- $\kappa = \nabla \cdot \left(\frac{n}{|n|} \right)$ curvature
- σ surface tension coefficient
- $F_S = -\sigma \kappa n$ surface tension force



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'hyper-consistent' schemes

Proposition

- when α_k is constant, the system becomes conservative :

$$\partial_t(\mathbf{w}) + \nabla \cdot (\mathbf{f}) = 0$$

→ the discretisation of the numerical flux \mathbf{f} must be in a conservative form.

- when the solution $\bar{\mathbf{w}}$ is such that $(\bar{\rho}_k, \bar{u}, \bar{p})$ are constant, the volume fraction and masses equations are redundant. This criterion leads to an 'hyper-consistent' discretisation of the non-conservative operator.

'hyper consistent' numerical schemes rewrite formally as :

$$\left\{ \begin{array}{l} \frac{(\alpha_k)_i^{n+1} - (\alpha_k)_i^n}{\Delta t} + (\mathbf{u} \cdot \nabla)_i^h \alpha_k + (\beta_k \cdot \nabla)_i^h \mathbf{u} = 0 \\ \frac{\mathbf{w}_i^{n+1} - \mathbf{w}_i^n}{\Delta t} + \frac{\phi_{i+1/2}^n - \phi_{i-1/2}^n}{\Delta x} = 0 \end{array} \right.$$

$$\rightarrow \phi_{i+1/2}^n = \frac{1}{2} [\mathbf{f}_{i+1}^n + \mathbf{f}_i^n - \mathbf{P}|\Lambda|\mathbf{P}^{-1}(\mathbf{w}_{i+1}^n - \mathbf{w}_i^n)]$$

$$\rightarrow (\mathbf{u} \cdot \nabla)_i^h \alpha_k = \frac{\phi_{i+1/2}^{\alpha_k \rho_k}(\alpha_k^n, \bar{\mathbf{w}}^n) - \phi_{i-1/2}^{\alpha_k \rho_k}(\alpha_k^n, \bar{\mathbf{w}}^n)}{\Delta x}$$

Pressure relaxation solver

$$\partial_t(\mathbf{W}_R) + \partial_x(\mathbf{F}_R) + B_R \partial_x(\mathbf{V}_R) = \frac{1}{\lambda} \mathbf{R}$$

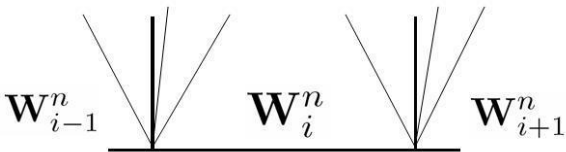
$$\mathbf{W}_R = \begin{pmatrix} \alpha_k \\ \alpha_k \rho_k \\ \rho \mathbf{u} \\ E \\ \pi \end{pmatrix}, \quad \mathbf{B}_R = \begin{pmatrix} \mathbf{u} & 0 & \tilde{\beta}_k & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & a^2/\rho & 0 & \mathbf{u} \end{pmatrix}, \quad \mathbf{F}_R = \begin{pmatrix} 0 \\ \alpha_k \rho_k \mathbf{u} \\ \rho \mathbf{u}^2 + \pi \\ (E + \pi) \mathbf{u} \\ 0 \end{pmatrix}$$

$$\mathbf{V}_R = \begin{pmatrix} \alpha_k \\ \rho_k \\ \mathbf{u} \\ \epsilon \\ \pi \end{pmatrix}, \quad \mathbf{R} = \begin{pmatrix} 0 \\ 0 \\ 0 \\ 0 \\ p - \pi \end{pmatrix}$$

- relaxation pressure $\pi \rightarrow p$
- relaxation time $\lambda \rightarrow 0$
- relaxation parameter $a > \rho c$
- approximate compaction $\tilde{\beta}^k = \alpha_k - \frac{y_k a_k^2}{a^2}$

wave	Riemann invariants					
λ_u	u			π		
$\lambda_{u \pm a/\rho}$	$u \pm a p$	$\pi \mp a u$	$\pi^2 - 2a^2 \epsilon$	y_k	$(\alpha_k - y_k a^2/a_k^2)/\rho$	a

1-D numerical implementation of relaxation schemes



exact solution :

$$\mathcal{V}_{i-1/2}(x) = \mathcal{V}\left(\frac{x - x_{i-1/2}}{\Delta t}, \mathbf{W}_{i-1}^n, \mathbf{W}_i^n\right)$$

Godunov method :

$$\Delta x \mathbf{W}_i^{n+1} = \int_{x_{i-1/2}}^{x_i} \mathcal{V}_{i-1/2}(x) dx + \int_{x_i}^{x_{i+1/2}} \mathcal{V}_{i+1/2}(x) dx$$

wave propagation form :

$$\mathbf{W}_i^{n+1} = \mathbf{W}_i^n - \frac{\Delta t}{\Delta x} \left[\left(\bar{\mathbf{A}}^- \Delta \mathbf{W}\right)_{i+1/2}^n + \left(\bar{\mathbf{A}}^+ \Delta \mathbf{W}\right)_{i-1/2}^n \right]$$

$$\left(\bar{\mathbf{A}}^- \Delta \mathbf{W}\right)_{i+1/2}^n = -\frac{1}{\Delta t} \int_{x_i}^{x_{i+1/2}} (\mathcal{V}_{i+1/2}(x) - \mathbf{W}_i^n) dx$$

$$\left(\bar{\mathbf{A}}^+ \Delta \mathbf{W}\right)_{i-1/2}^n = -\frac{1}{\Delta t} \int_{x_{i-1/2}}^{x_i} (\mathcal{V}_{i-1/2}(x) - \mathbf{W}_i^n) dx$$

'hyper-consistent' schemes properties

update solution :

$$\mathbf{W}_i^{n+1} = \mathbf{W}_i^n - \frac{\Delta t}{\Delta x} (\psi_{i+1/2} - \psi_{i-1/2})$$

'hyper consistency' formulation :

$$\psi_{i+1/2} = \phi_{i+1/2} + \left(\overline{\mathbf{B}}^- \Delta \mathbf{V}\right)_{i+1/2}^n, \quad \psi_{i-1/2} = \phi_{i-1/2} - \left(\overline{\mathbf{B}}^+ \Delta \mathbf{V}\right)_{i-1/2}^n$$

dissipative formulation :

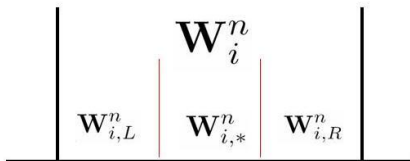
$$\psi_{i+1/2} = \frac{1}{2} \left[\mathbf{F}_i^n + \mathbf{F}_{i+1}^n - \left(|\overline{\mathbf{A}}| \Delta \mathbf{W}\right)_{i+1/2}^n - \left(|\overline{\mathbf{B}}| \Delta \mathbf{V}\right)_{i+1/2}^n \right] + \frac{1}{2} \left(\overline{\mathbf{B}} \Delta \mathbf{V}\right)_{i+1/2}^n$$

$$\psi_{i-1/2} = \frac{1}{2} \left[\mathbf{F}_{i-1}^n + \mathbf{F}_i^n - \left(|\overline{\mathbf{A}}| \Delta \mathbf{W}\right)_{i-1/2}^n - \left(|\overline{\mathbf{B}}| \Delta \mathbf{V}\right)_{i-1/2}^n \right] - \frac{1}{2} \left(\overline{\mathbf{B}} \Delta \mathbf{V}\right)_{i-1/2}^n$$

Proposition

There exists a matrix $\overline{\mathbf{B}}$ such that approximate solvers associated to 'hyper consistent' numerical schemes satisfy :

$$\sum_k (\delta \mathcal{V})_k = \Delta \mathbf{F} + \overline{\mathbf{B}} \Delta \mathbf{V}$$

2nd order space accuracy

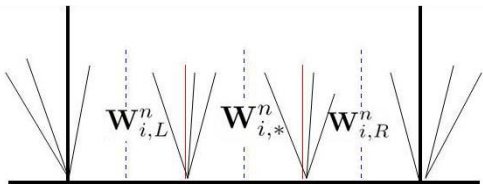
gradients construction :

$$\nabla \mathbf{W}_{i,R}^n = \text{Lim} \left(\frac{\mathbf{W}_{i+1} - \mathbf{W}_{i-1}}{2\Delta x}, \frac{(\Delta \mathbf{W})_{i+1/2}}{x} \right), \quad \nabla \mathbf{W}_{i,L}^n = \text{Lim} \left(\frac{\mathbf{W}_{i+1} - \mathbf{W}_{i-1}}{2\Delta x}, \frac{(\Delta \mathbf{W})_{i-1/2}}{\Delta x} \right)$$

states reconstruction :

$$\mathbf{W}_{i,R}^n = \mathbf{W}_i^n + \frac{\Delta x}{2} \nabla \mathbf{W}_{i,R}^n, \quad \mathbf{W}_{i,L}^n = \mathbf{W}_i^n - \frac{\Delta x}{2} \nabla \mathbf{W}_{i,L}^n$$

conservation property : $\mathbf{W}_i^n = \frac{1}{3}(\mathbf{W}_{i,L}^n + \mathbf{W}_{i,*}^n + \mathbf{W}_{i,R}^n)$

2nd order implementation

$$\mathbf{w}_i^{n+1} = \mathbf{w}_i^n - \frac{\Delta t}{\Delta x} [H_{i+1/2} - H_{i-1/2}]$$

second order flux formulation :

$$H_{i-1/2}^n = \mathbf{F}_{i,L}^n + \mathbf{B}_i^n \mathbf{W}_{i,L}^n + \frac{1}{\Delta t} \int_{x_{i-1/3}}^{x_{i-1/2}} \left[\mathcal{V} \left(\frac{x - x_{i-1/2}}{\Delta t}, \mathbf{W}_{i-1,R}^n, \mathbf{W}_{i,L}^n \right) - \mathbf{W}_{i,L}^n \right] dx$$

$$H_{i+1/2}^n = \mathbf{F}_{i,R}^n + \mathbf{B}_i^n \mathbf{W}_{i,R}^n - \frac{1}{\Delta t} \int_{x_{i+1/3}}^{x_{i+1/2}} \left[\mathcal{V} \left(\frac{x - x_{i+1/2}}{\Delta t}, \mathbf{W}_{i,R}^n, \mathbf{W}_{i+1,L}^n \right) - \mathbf{W}_{i,R}^n \right] dx$$

Second order

- second order time accuracy : default correction :

$$\text{prediction : } \mathbf{W}_i^{n+1/2} = \mathbf{W}_i^n - \frac{\Delta t}{a_i} \sum_{j \in \nu(i)} \phi_{ij,ji}^n$$

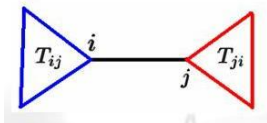
$$\text{correction : } \mathbf{W}_i^{n+1} = \frac{1}{2} \left(\mathbf{W}_i^{n+1/2} + \mathbf{W}_i^n - \frac{\Delta t}{a_i} \sum_{j \in \nu(i)} \phi_{ij,ji}^{n+1/2} \right)$$

- 2D second order space accuracy :
MUSCL technique :

$$\nabla \mathbf{W}_{ij} = \frac{1}{2} \left(\nabla \mathbf{W}_{ij}^{cent} + \nabla \mathbf{W}_{ij}^{up} \right) \cdot \boldsymbol{\eta}_{ij}$$

Van Albada Van Leer limiter :

$$\mathbf{W}_{ij} = \mathbf{W}_i + \frac{1}{2} \text{Lim}(\nabla \mathbf{W}_{ij}, \nabla \mathbf{W}_{ij}^{cent} \cdot \boldsymbol{\eta}_{ij})$$



Source terms

- splitting technique

$$\left\{ \begin{array}{l} \partial_t(\alpha_k) = 0 \\ \partial_t(\alpha_k \rho_k) = 0 \\ \partial_t(\rho \mathbf{u}) = \rho \mathbf{g} + F_S + \nabla \cdot \tau \\ \partial_t(E) = (\rho \mathbf{g} + F_S) \cdot \mathbf{u} + \nabla \cdot (\tau \mathbf{u}) \end{array} \right.$$

- \mathbf{g} gravity
- F_S surface force
- $\tau = \frac{1}{2} (\nabla \mathbf{u} + \nabla \mathbf{u}^T)$ viscosity stress tensor
- $\frac{1}{\mu} = \sum_k \frac{\alpha_k}{\mu_k}$ mixture viscosity coefficient

Low Mach preconditioning

- Low Mach number : $Ma = \frac{|\mathbf{u}|}{c} \ll 1$
- solutions spoiled by rounded errors coming from acoustic wave resolution
- Turkel pressure preconditioning method (rounded errors filter) :

$$\frac{1}{Ma^2} \partial_t(p) + \mathbf{u} \cdot \nabla(p) + \rho c^2 \nabla \cdot (\mathbf{u}) = 0$$

- preconditioned relaxation system :

$$\left\{ \begin{array}{l} \partial_t(\alpha_k) + \mathbf{u} \cdot \nabla(\alpha_k) + \tilde{\beta}_k \nabla \cdot (\mathbf{u}) = 0 \\ \partial_t(\alpha_k \rho_k) + \nabla \cdot (\alpha_k \rho_k \mathbf{u}) = 0 \\ \partial_t(\rho \mathbf{u}) + \nabla \cdot (\rho \mathbf{u}^2 + \pi) = 0 \\ \partial_t(E) + \nabla \cdot ((E + \pi) \mathbf{u}) = 0 \\ \partial_t(\pi) + \left(\mathbf{u} + \frac{a^R - a^L}{\rho} \right) \cdot \nabla(\pi) + \frac{a^R a^L}{\rho} \nabla \cdot (\mathbf{u}) = \frac{1}{\lambda} (p - \pi) \end{array} \right.$$

- L.D., waves : $\lambda^\pm \approx \frac{(M^2 + 1)u \pm \sqrt{(M^2 - 1)^2 u^2 + 4M^2 c^2}}{2}$

Implicit formulation

- implicit formulation

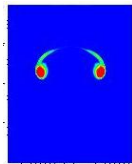
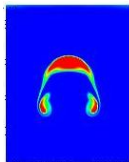
$$\left(1 + \frac{\Delta t}{a_i} \sum_{j \in \nu(i)} \mathbf{A}_{ij}^+ \cdot \boldsymbol{\eta}_{ij} \right) (\mathbf{W}_i^{n+1} - \mathbf{W}_i^n) + \sum_{j \in \nu(i)} \mathbf{A}_{ji}^+ \cdot \boldsymbol{\eta}_{ji} (\mathbf{W}_j^{n+1} - \mathbf{W}_j^n) = -\frac{\Delta t}{a_i} \psi_{ij}^n$$

- Acoustic scheme (Guillard, Murrone)

$$\mathbf{A}_{ij}^+ = \left(\frac{\partial \mathbf{F}_{ij}^*}{\partial \mathbf{W}_{ij}^*} + \mathbf{B}(\mathbf{W}_i^n) \right) \left(\frac{\partial \mathbf{W}_{ij}^*}{\partial \mathbf{W}_i^*} \right)$$

- Resolution based on iterative methods :

Jacobi
Gauss-Seidel
Gmres



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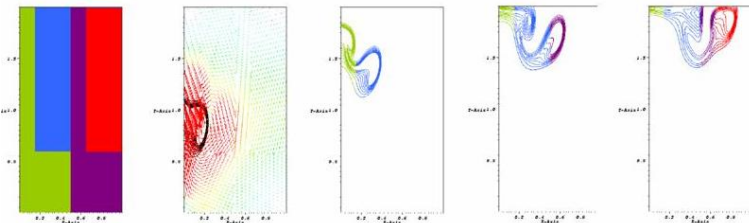
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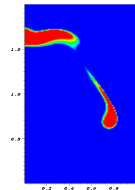
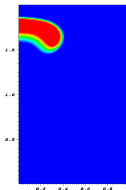
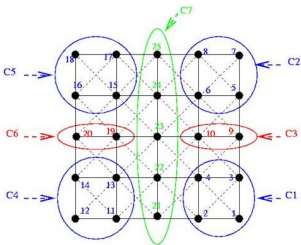
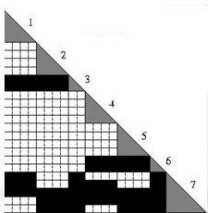
Fluidbox parallelisation

- parallelisation based on mesh partitioning
- block Gauss Seidel or block Jacobi preconditioners
- dedicated parallel assembly implementation
- partitioning step is performed by Metis
- overlap the frontiers of domain decomposition



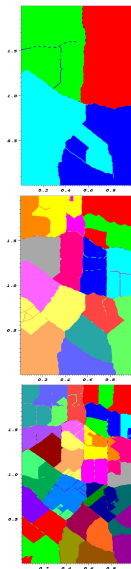
PASTIX : direct solver and graph partitioner

- PASTIX : parallel direct solver based on LU method
robust factorisation (ill conditioned matrix)
unsymmetric matrix with symmetric pattern
- PASTIX features
 - matrix ordering techniques based on graph partitioning :
 - incomplete Nested Dissection
 - approximate Minimum Degree method
 - benefits :
 - minimising the fill-in in the factorized matrix and the number of operations
 - maximising the independence of computations
 - block symbolic factorisation :
 - compute the block pattern of the factorized matrix (linear time complexities)
 - block data structure allow to use dense linear algebra subroutine (BLAS3)



PASTIX efficiency

Mesh Unknowns	101 × 201 121806	201 × 401 483606	401 × 801 1927206
4 processors			
PASTIX	1.41 10 ³	1.59 10 ⁴	— — —
Gauss-Seidel 10 ⁻³	4.77 10 ⁴	4.05 10 ⁵	3.38 10 ⁶
Gauss-Seidel 10 ⁻⁵	7.13 10 ⁴	5.57 10 ⁵	— — —
Gmres 10 ⁻³	5.95 10 ³	4.78 10 ⁴	4.18 10 ⁵
Gmres 10 ⁻⁵	5.22 10 ⁴	4.23 10 ⁵	— — —
16 processors			
PASTIX	6.10 10 ²	5.65 10 ³	6.17 10 ⁴
Gauss-Seidel 10 ⁻³	1.63 10 ⁴	1.55 10 ⁵	1.32 10 ⁶
Gauss-Seidel 10 ⁻⁵	2.46 10 ⁴	2.12 10 ⁵	1.53 10 ⁷
Gmres 10 ⁻³	2.07 10 ³	1.82 10 ⁴	1.32 10 ⁵
Gmres 10 ⁻⁵	1.80 10 ⁴	1.65 10 ⁵	1.84 10 ⁶
32 processors			
PASTIX	6.30 10 ²	5.71 10 ³	6.31 10 ⁴
Gauss-Seidel 10 ⁻³	8.15 10 ³	9.14 10 ⁴	1.46 10 ⁵
Gauss-Seidel 10 ⁻⁵	1.23 10 ⁴	1.26 10 ⁵	9.74 10 ⁵
Gmres 10 ⁻³	1.04 10 ³	1.07 10 ⁴	7.76 10 ⁴
Gmres 10 ⁻⁵	8.83 10 ³	9.67 10 ⁴	1.12 10 ⁵



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Conclusion

- Done
 - 'hyper consistent' numerical schemes (relaxation scheme)
 - second order extension
 - low Mach preconditioning
 - implicit formulation
 - parallel computing : Pastix
- In progress
 - 3D
 - comparison with experiments