

Reduced-order models for fluids, using balanced truncation and dynamically scaling modes

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Mechanical
and Aerospace
Engineering
PRINCETON

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- Students

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(channel flow)



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(airfoil separation)



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(free shear flow)



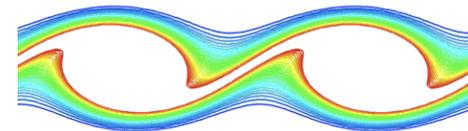
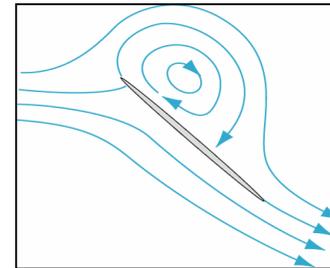
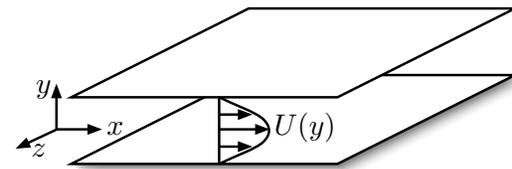
- Collaborators

- Yannis Kevrekidis (Princeton)
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 - Gilead Tadmor (Northeastern)
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Outline

- Approximate balanced truncation using POD
 - Importance of inner product for Galerkin projection
 - Balanced truncation
 - Method of snapshots
- Applications
 - Linearized channel flow
 - Separating flow past an airfoil
- Dynamically scaling POD modes
 - Free shear layer
 - Scaled basis functions
 - Template fitting
 - Equations for the shear layer thickness

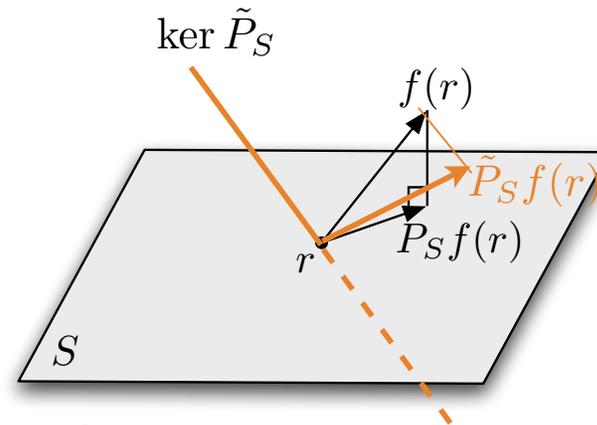


Galerkin projection

- Dynamics evolve on a high-dimensional space (or infinite-dim'l)
- Project dynamics onto a low-dimensional subspace S

$$\dot{x} = f(x) \quad x \in V$$

$$r \in S \subset V$$



- Define dynamics on the subspace by

$$\dot{r} = P_S f(r) \quad P_S : V \rightarrow S \text{ is a projection}$$

- Two choices:
 - choice of subspace
 - choice of inner product
(equivalently, choice of the nullspace for a non-orthogonal projection)



Energy-based inner products

- Reduced-order models can behave unpredictably

- Can even change stability type of equilibria

[Rempfer, Thoret. CFD 2000]

- Simple example: consider the system:

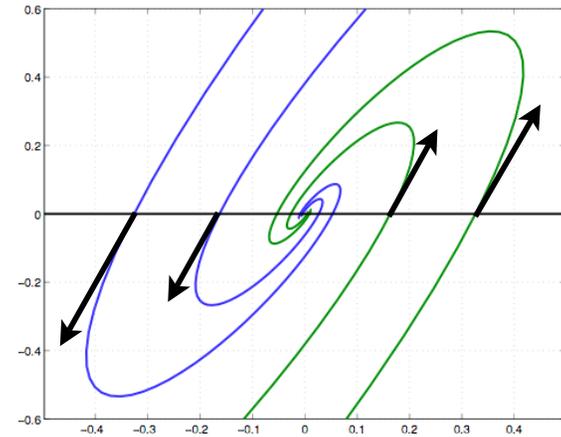
$$\frac{d}{dt} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} = \begin{pmatrix} 1 & -1 \\ 3 & -2 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix}$$

- Sink at the origin
- Projection onto x_1 axis is

$$\dot{x}_1 = x_1 \quad \text{unstable}$$

- Can at least fix this simple problem by changing the inner product used for the projection

- **Cute result:** If an orthogonal projection is used with an “energy-based” inner product, this will ensure stability of the origin
- Note: does not guarantee stability preserved for other equilibrium points, periodic orbits, etc.



[Rowley, T Colonius, RM Murray, Phys D 2004]

Energy-based inner products

- Consider a system with a stable equilibrium point at the origin:

$$\dot{x} = f(x) \quad f(0) = 0 \quad x \in \mathbb{R}^n$$

- Consider an inner product whose induced norm is a Liapunov function (“energy-based”):

$$\langle x, y \rangle = x^T Q y, \quad Q > 0 \quad \begin{array}{l} V(x) = x^T Q x \text{ is a Liapunov function} \\ \dot{V}(x) = 2x^T f(x) \leq 0, \quad \forall x \in U \end{array}$$

- Reduced-order dynamics given by orthogonal projection

$$\begin{array}{ll} r = Px & P^2 = P \\ \dot{r} = Pf(r) & \langle x, Py \rangle = \langle Px, y \rangle \quad QP = P^T Q \end{array}$$

- Then V is a Liapunov function for the reduced-order system:

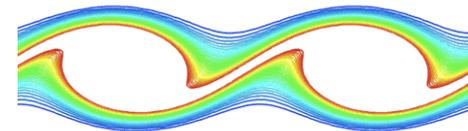
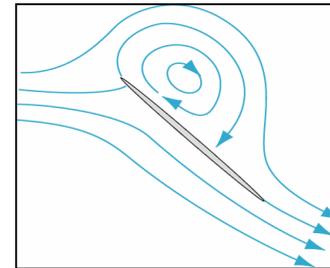
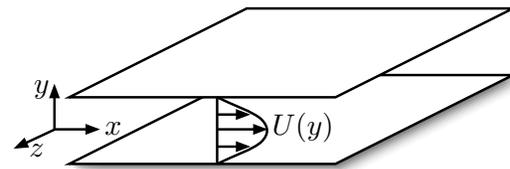
$$\begin{aligned} \dot{V}(r) &= 2r^T QP f(r) = 2r^T P^T Q f(r) = 2(P r)^T Q f(r) \\ &= 2r^T Q f(r) \leq 0 \end{aligned}$$

- So: if an energy-based inner product is used, the origin is stable for the reduced-order system, regardless of the subspace used for the projection



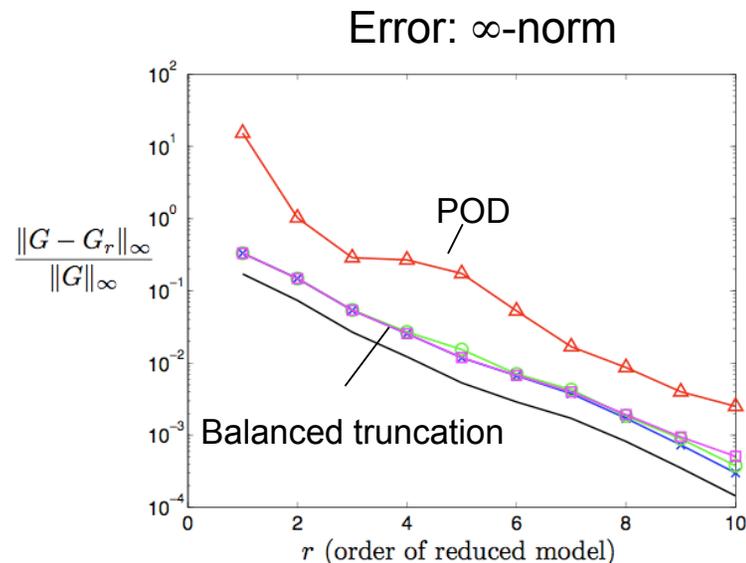
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Are POD modes optimal?

- POD modes are not optimal for Galerkin projection
 - POD determines a subspace that optimally captures the **energy** in a given dataset
 - These modes are usually **not optimal** for Galerkin projection
 - **Low-energy modes** can play an important role in the dynamics [Aubry, Holmes, Lumley, 1988; Smith 2002 PhD thesis, Princeton]
 - Can often do better with **balanced truncation** [Moore 1981]



Balanced truncation

- Why doesn't everybody use this?
 - Valid for stable, linear systems
 - Extensions for unstable systems [Jonckheere & Silverman 1983, Zhou 2001]
 - Extensions for nonlinear systems [Scherpen 1993, Lall, Marsden, Glavaski 1999]
 - Computationally expensive for large systems
 - n^3 computational time: $n > 10^5$ for typical fluids simulations
- Improvements for large systems
 - POD is tractable for large systems. Can we extend, e.g., the **method of snapshots**, to compute balancing transformations?
 - Based on earlier snapshot-based methods:
 - Lall, Marsden, & Glavaski, 1999
 - Willcox & Peraire, 2001



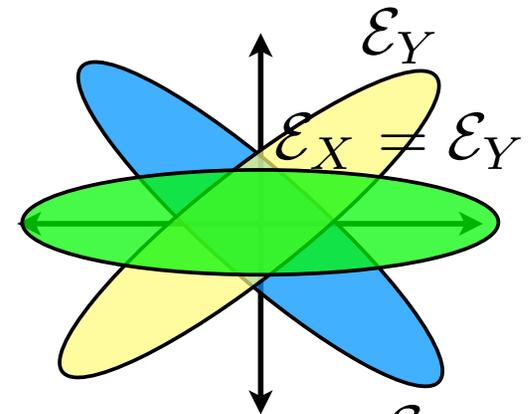
Overview of balanced truncation

- Start with a stable, linear input-output system

$$\dot{x} = Ax + Bu$$

What are you interested in capturing?

$$y = Cx$$



- Compute controllability and observability Gramians

$$X = \int_0^{\infty} e^{At} BB^* e^{A^*t} dt$$

$$Y = \int_0^{\infty} e^{A^*t} C^* C e^{At} dt$$

$$AX + XA^* + BB^* = 0$$

$$A^*Y + YA + C^*C = 0$$

States easily excited
by an input

States that have large influence
on the output

- Find a transformation T that simultaneously diagonalizes X and Y

$$x = Tz, \quad T^{-1}X(T^{-1})^* = T^*YT = \Sigma = \begin{bmatrix} \sigma_1 & & \\ & \ddots & \\ & & \sigma_n \end{bmatrix}$$

- Change coordinates, and truncate states that are least controllable/observable



Empirical Gramians

- Construct Gramians from impulse response data

- Not solving Liapunov equations
- For a single input: compute impulse-state response:

solution $\dot{x} = Ax, \quad x(0) = B$
 $x(t) = e^{At} B$

- The controllability Gramian is then $W_c = \int_0^{\infty} x(t)x(t)^T dt$
- Discretize in time, collect snapshots into a matrix:

$$X = \begin{bmatrix} | & & | \\ x(t_1) & \cdots & x(t_m) \\ | & & | \end{bmatrix}$$

- Then $W_c \approx XX^T$
- For observability Gramian, same procedure, but use adjoint equations $\dot{z} = A^*z \quad z(0) = C^*$
- For multiple inputs/outputs, same procedure, but do one impulse-response for each input/output

[Lall et al, 1999]



Method of snapshots

- POD: method of snapshots vs. direct method

$$X = \begin{bmatrix} | & & | \\ x_1 & \cdots & x_m \\ | & & | \end{bmatrix}$$

POD modes (direct method):

$$XX^T \varphi = \lambda \varphi \quad n \times n$$

POD modes (method of snapshots):

$$\varphi = Xc$$

$$X^T Xc = \lambda c \quad m \times m$$

[Sirovich, Q Appl Math 1987]

- method of snapshots more efficient when $m < n$.
- **Balanced truncation: method of snapshots**

- Empirical Gramians represented as $W_c = XX^T \quad n \times n$
 $W_o = YY^T$

- Find a balancing transformation with an SVD of $Y^T X \quad m_y \times m_x$

[Rowley, Int. J Bif Chaos, 2005]



Computing modes

- Snapshot matrices

$$X = \begin{bmatrix} | & & | \\ x(t_1) & \cdots & x(t_m) \\ | & & | \end{bmatrix}$$

Linearized snapshots

$$Y = \begin{bmatrix} | & & | \\ z(t_1) & \cdots & z(t_l) \\ | & & | \end{bmatrix}$$

Adjoint snapshots

- Compute SVD

$$Y^* X = U \Sigma V^*$$

- Obtain bi-orthogonal set of modes:

$$\Phi_r = \begin{bmatrix} | & & | \\ \varphi_1 & \cdots & \varphi_r \\ | & & | \end{bmatrix}$$

Direct modes
linear combinations of
direct snapshots

$$\Psi_r = \begin{bmatrix} | & & | \\ \psi_1 & \cdots & \psi_r \\ | & & | \end{bmatrix}$$

Adjoint modes
linear combinations of
adjoint snapshots

$$\Phi = X V_r \Sigma_r^{-1/2}$$

$$\Psi = Y U_r \Sigma_r^{-1/2}$$

$$\Psi_r^* \Phi_r = I_r$$



Reduced-order models

- Original equations

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

- Form reduced-order model

- Do not need to transform entire state: just take first r modes

$$\dot{a} = \Psi_r^* A \Phi_r a + \Psi_r^* B u$$

$$y = C \Phi_r a + D u$$

- Extensions to nonlinear systems straightforward

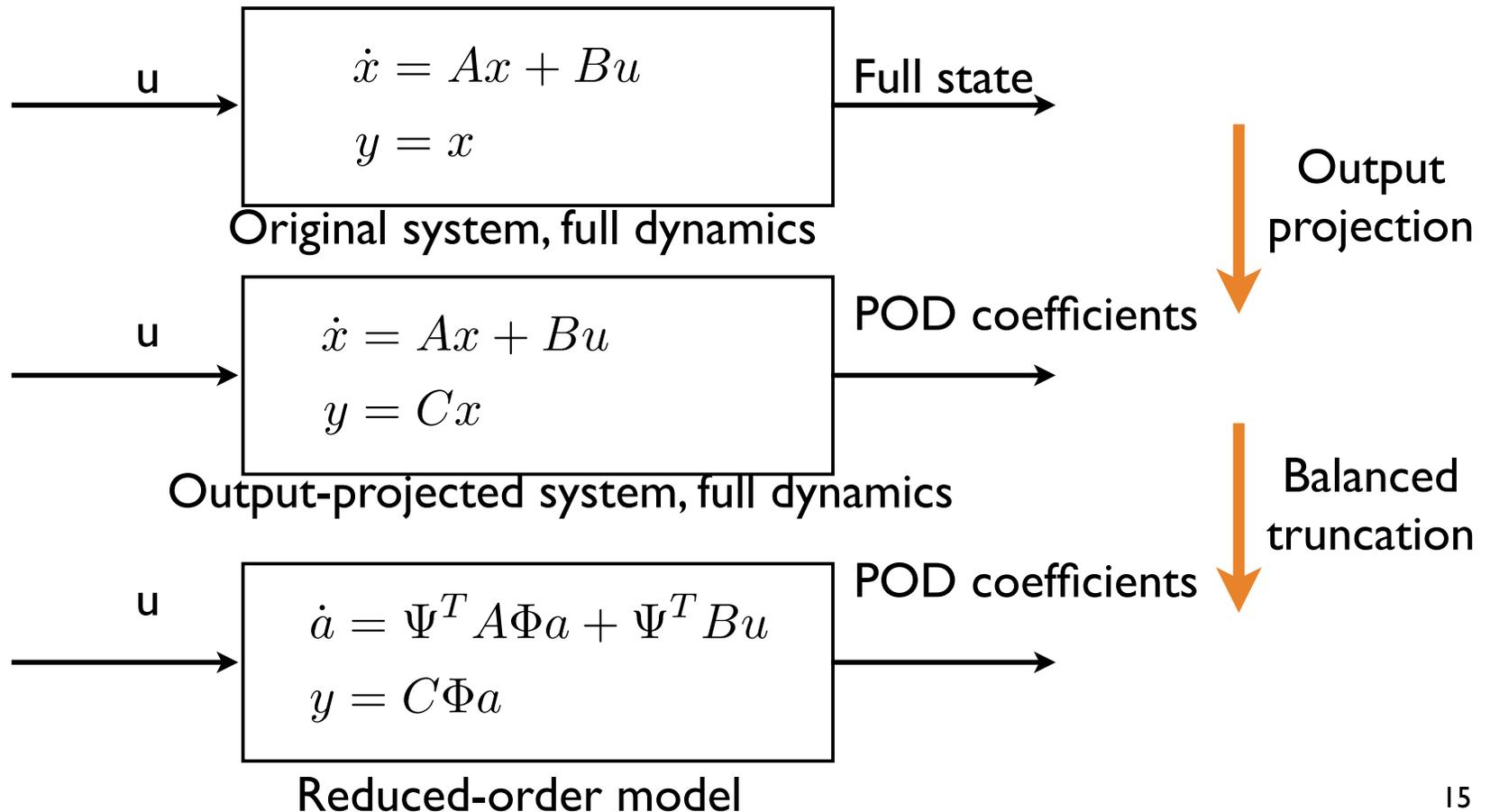
- For instance, compute modes for linearized system, project nonlinear dynamics

$$\dot{x} = f(x) \quad \longrightarrow \quad \begin{aligned} x(t) &= \sum_{j=1}^r a_j(t) \varphi_j \\ \dot{a}_j(t) &= \langle \psi_j, f(x) \rangle \end{aligned}$$



Large numbers of outputs

- Often, we are interested in modeling the full state
 - If dimension is large, project output onto POD modes
 - POD gives optimally-close output-projected system (in 2-norm)



Approximate balanced truncation for large systems

- Method of snapshots enables one to compute approximate balanced truncations with cost similar to POD
 - One simulation for each control input, one adjoint simulation for each output
 - One SVD, (# direct snapshots) x (# adjoint snapshots)
 - If number of outputs is large, method for projection onto smaller-rank output
- Balanced truncation is just POD with respect to an inner product defined by the observability Gramian Y :

$$\langle x_1, x_2 \rangle_Y = x_1^T Y x_2$$

- Observability Gramian is always a Liapunov function => preserves stability!
- Obtain set of bi-orthogonal modes:

direct modes: $\{\varphi_1, \dots, \varphi_n\}$

adjoint modes: $\{\psi_1, \dots, \psi_n\}$

bi-orthogonal: $\langle \psi_i, \varphi_j \rangle = \delta_{ij}$

Galerkin:

$$\dot{x} = f(x)$$

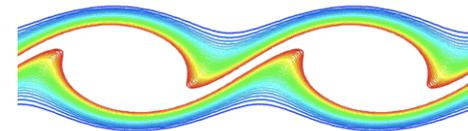
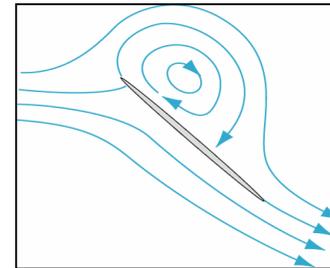
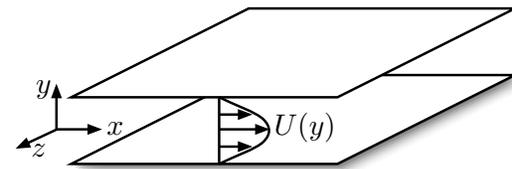
$$x(t) = \sum_j a_j(t) \varphi_j$$

$$\dot{a}_j(t) = \langle \psi_j, f(x) \rangle$$



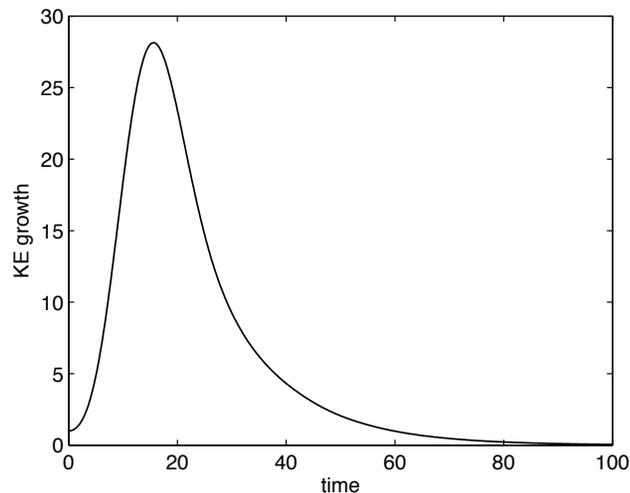
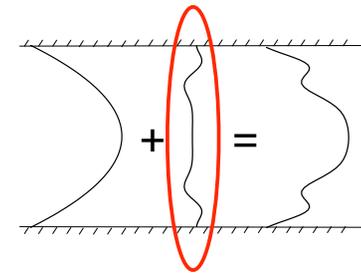
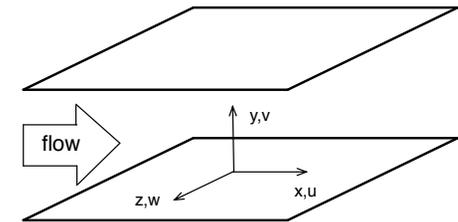
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Application: Linearized Channel Flow

- Plane channel flow with periodic boundary conditions
 - Goal: delay transition to turbulence using feedback control
 - Goal: improved understanding of transition mechanisms
 - **Focus: low-dimensional models of transition**
- Linear development of small perturbations
 - Transition not predicted correctly by linear stability analysis
 - Non-normality of the governing operator results in large transient growth of exponentially stable perturbations
 - **Large linear system with complex dynamic behavior**



Previous work:

Trefethen et al [Science, 1993]

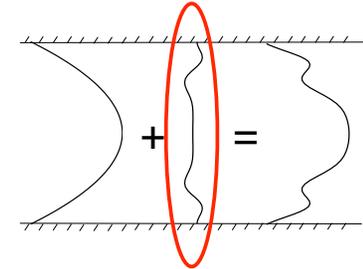
Farrell & Ioannou [96,96,01]

Schmid & Henningson [01]

Bamieh & Jovanovic [01,03]

Governing Equations

- Navier-Stokes equations linearized about a laminar profile
- Perturbation dynamics fully described by wall-normal velocity v and wall-normal vorticity η
- Clamped boundary conditions $v(\pm 1) = \frac{\partial v}{\partial y}(\pm 1) = 0$



Orr-Sommerfeld/Squire system

$$\frac{\partial}{\partial t} \begin{bmatrix} -\Delta & 0 \\ 0 & I \end{bmatrix} \begin{bmatrix} v \\ \eta \end{bmatrix} = \begin{bmatrix} L_{OS} & 0 \\ -U'\partial_z & L_{SQ} \end{bmatrix} \begin{bmatrix} v \\ \eta \end{bmatrix}$$

$$L_{OS} = U\partial_x\Delta - U''\partial_x - \frac{1}{Re}\Delta^2$$

$$L_{SQ} = -U\partial_x + \frac{1}{Re}\Delta$$

Adjoint system

$$\frac{\partial}{\partial t} \begin{bmatrix} -\Delta & 0 \\ 0 & I \end{bmatrix} \begin{bmatrix} v \\ \eta \end{bmatrix} = \begin{bmatrix} L_{OS}^* & U'\partial_z \\ 0 & L_{SQ}^* \end{bmatrix} \begin{bmatrix} v \\ \eta \end{bmatrix}$$

$$L_{OS}^* = -U\partial_x\Delta - 2U'\partial_x\partial_y - \frac{1}{Re}\Delta^2$$

$$L_{SQ}^* = U\partial_x + \frac{1}{Re}\Delta$$

$$\dot{x} = Ax + \underbrace{Bu_1}_{\text{actuation}} + \underbrace{Fu_2}_{\text{disturbances}}$$

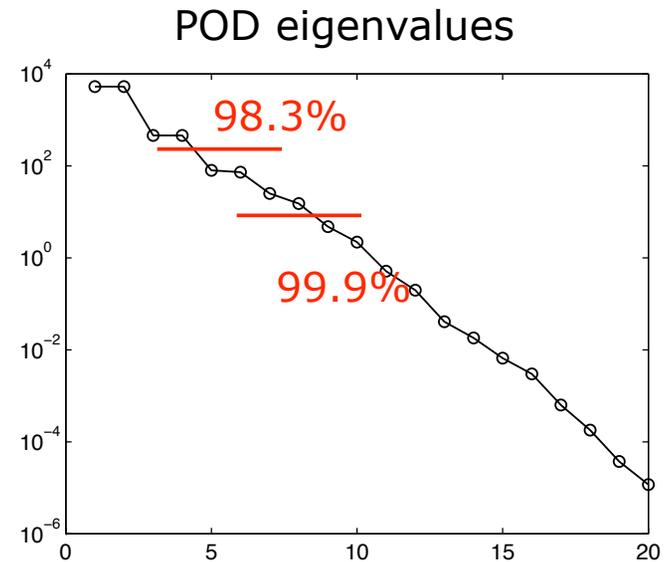
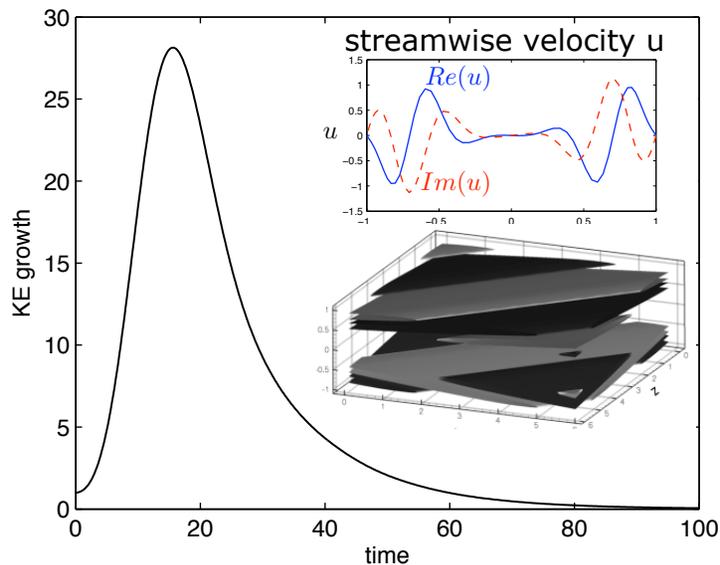
- System in standard state-space form with actuation and disturbances

Single-wavenumber perturbation - optimal

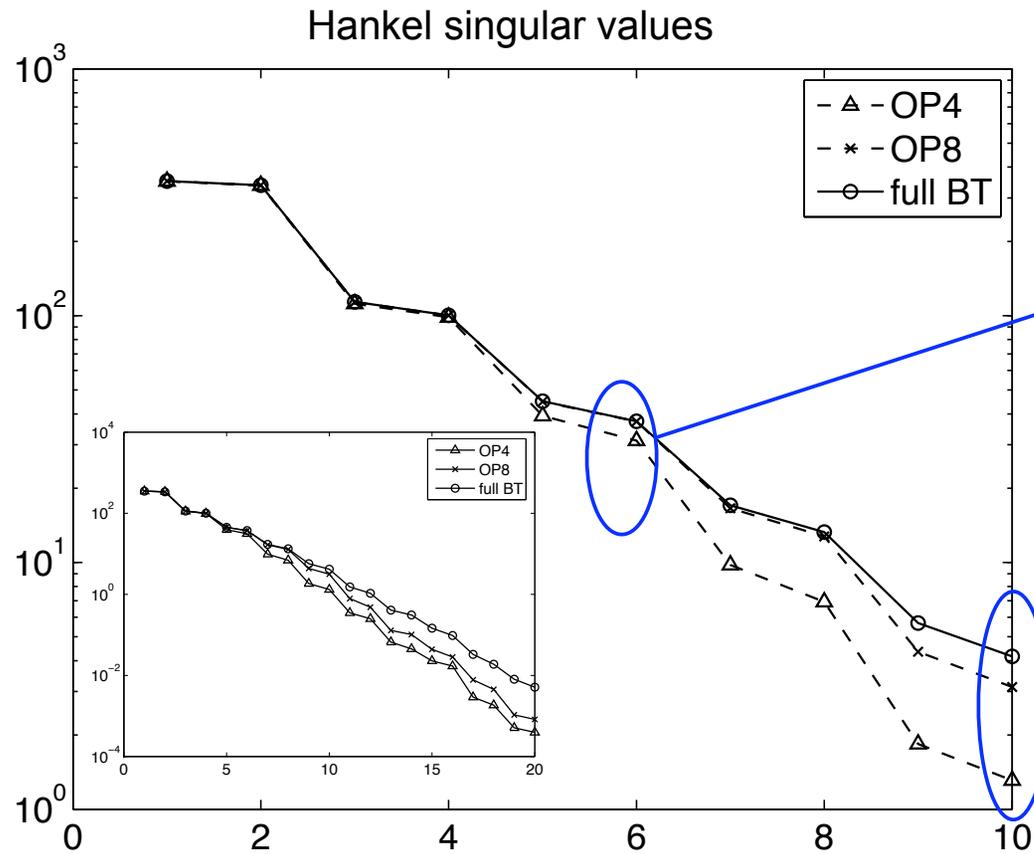
- Perturbations of the form

$$q = \hat{q}(y)e^{i\alpha x + i\beta z + \lambda t} \quad q = \begin{bmatrix} v \\ \eta \end{bmatrix}$$

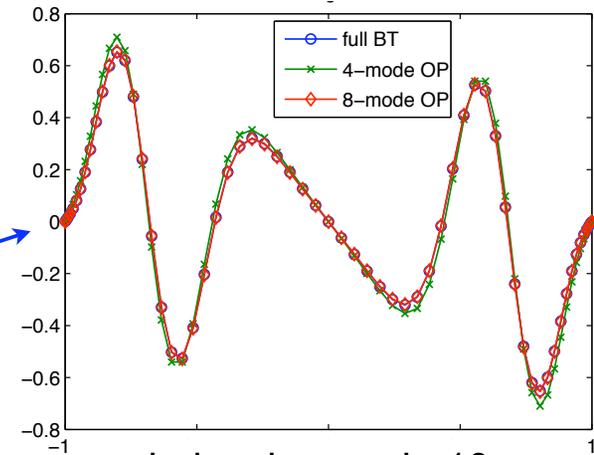
- System can be analyzed in 1-D so that full balanced truncation is tractable, allowing comparison with the BPOD approximation and POD
- Well-studied cases (Farrell, Henningson, Reddy, Schmid, Jovanovic, Bamieh)
- Case presented here $\alpha=1, \beta=1$ and exhibits rich dynamics



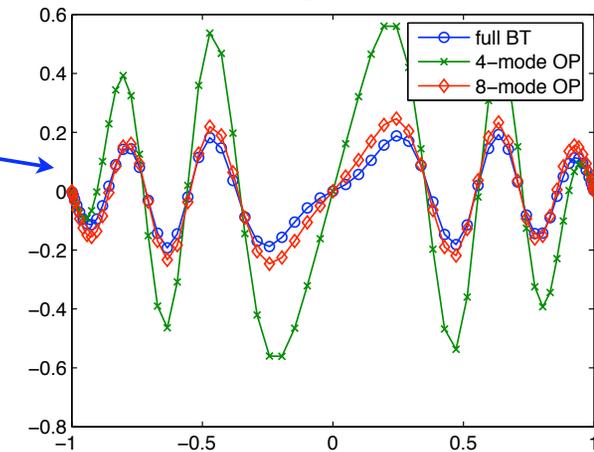
Modes and HSV - how good is BPOD?



balancing mode 6



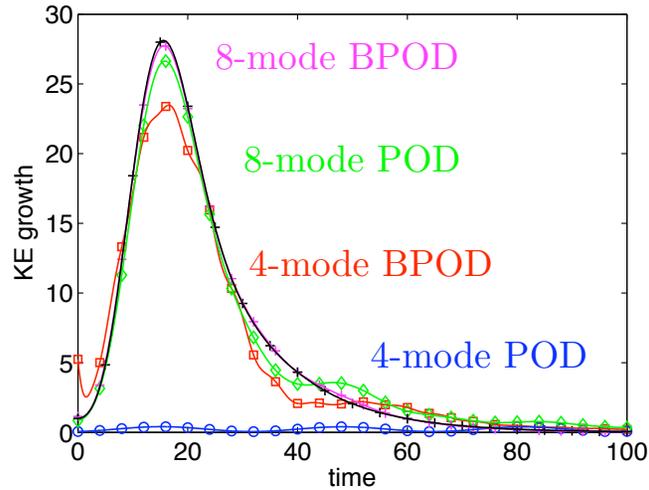
balancing mode 10



Both HSVs and balancing modes computed accurately up to approximately the rank of OP

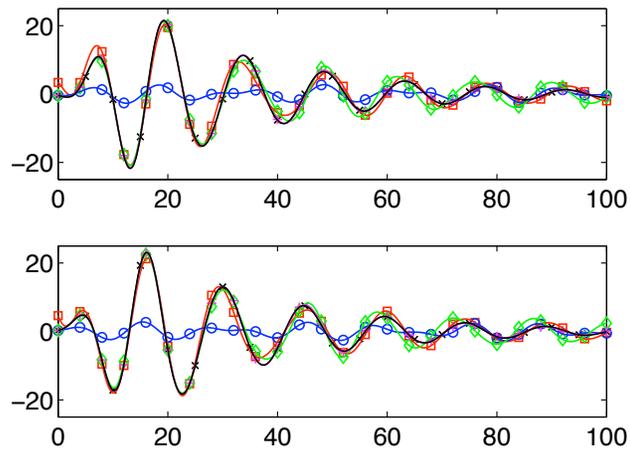
Single wavenumber - impulse response

Kinetic energy growth

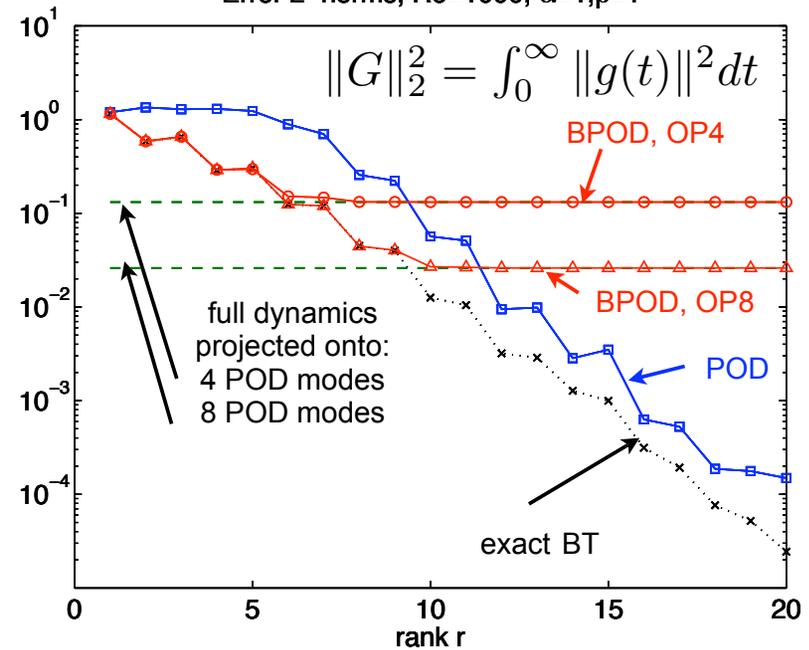


- Low-order POD models completely fail to capture energy growth
- BPOD model performance matches exact BT approximately up to the desired level of accuracy, determined by the output projection

First two outputs

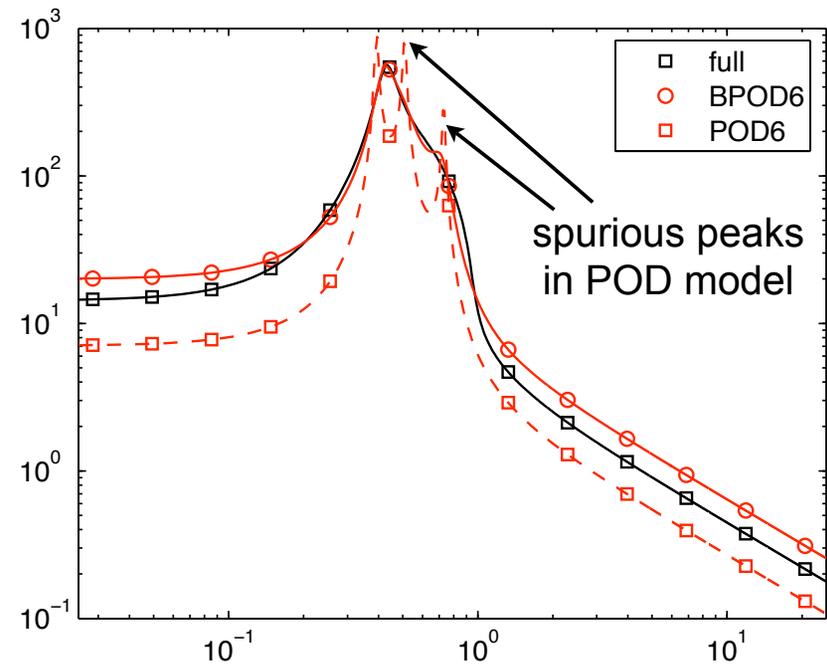
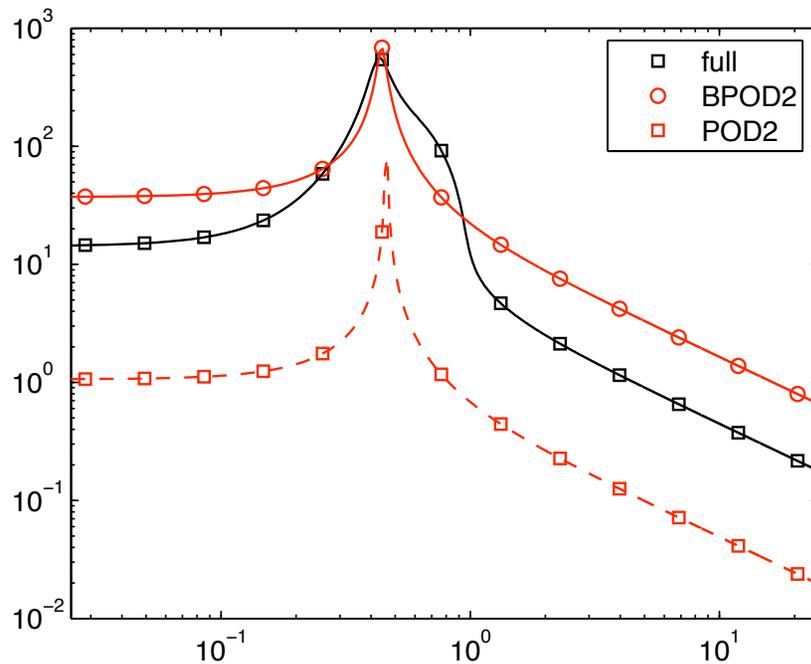


Error 2-norms, $Re=1000$, $\alpha=1, \beta=1$



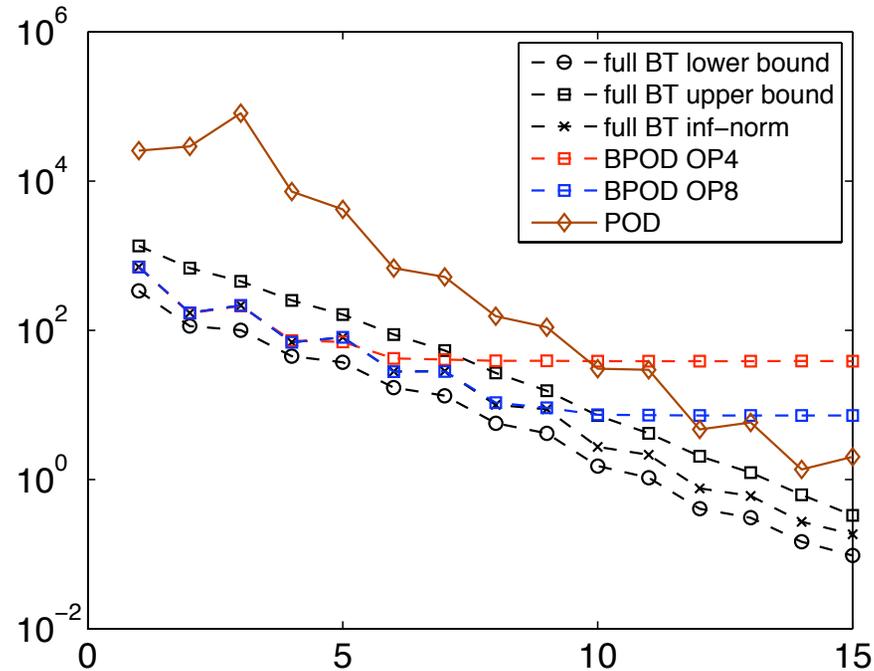
Single wavenumber - frequency response

- For a single wavenumber, frequency response can be computed exactly
- BPOD captures the resonant peak even at low order
- POD slowly catches up, but has **spurious peaks** due to eigenvalues near the imaginary axis



Single wavenumber - infinity norms

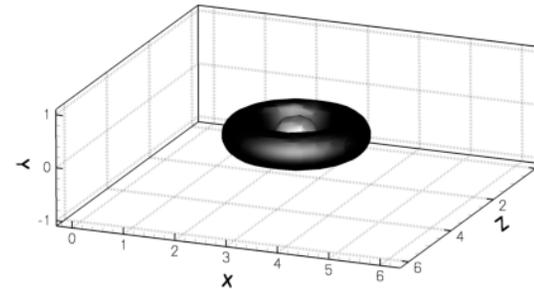
Infinity error norm bounds $\sigma_{r+1} \leq \|G - G_r\|_\infty \leq 2\sum_{j=r+1}^n \sigma_j$



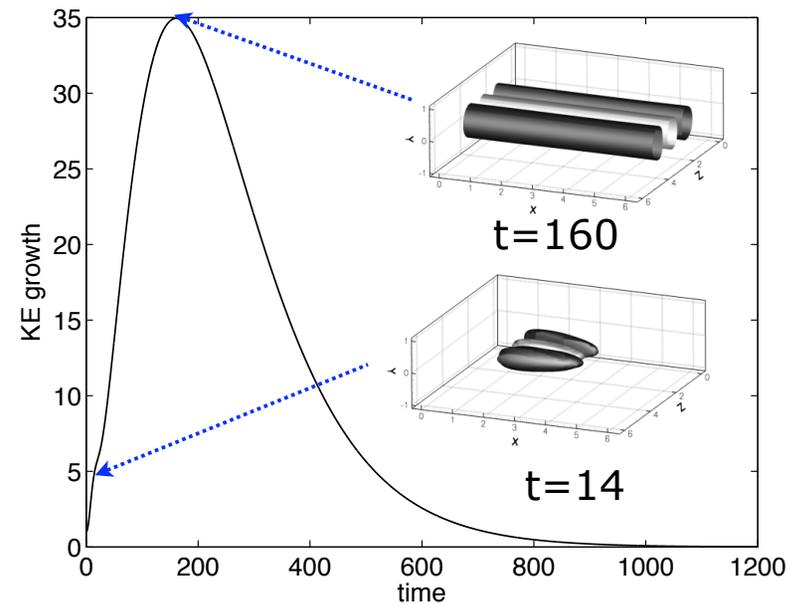
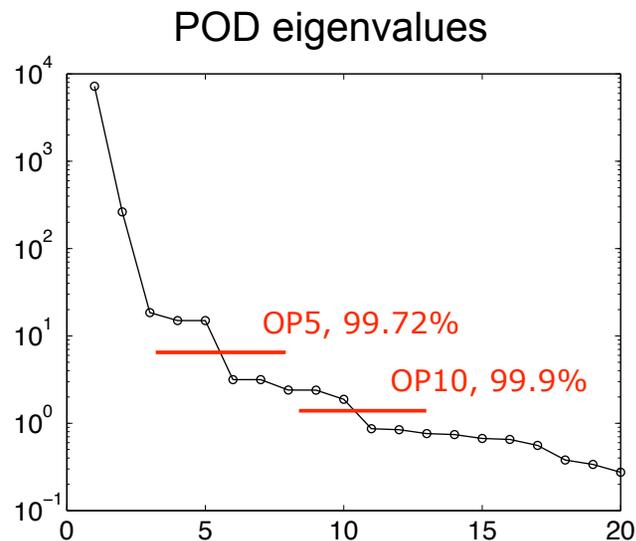
- Infinity norms of models also match those of exact BT up to approximately the rank of the output projection
- Again, POD 'catches up' only at a high rank

Localized actuator

- Periodic array of localized actuators in center of channel
- Large system (32x65x32), 133,120 states, exact BT intractable
- Impulse response snapshots obtained via linearized DNS, $Re=2000$
- Complex initial transient which develops into a streamwise-constant structure

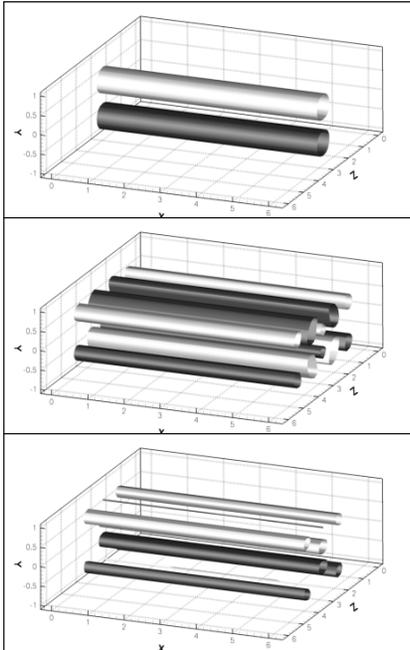


initial condition

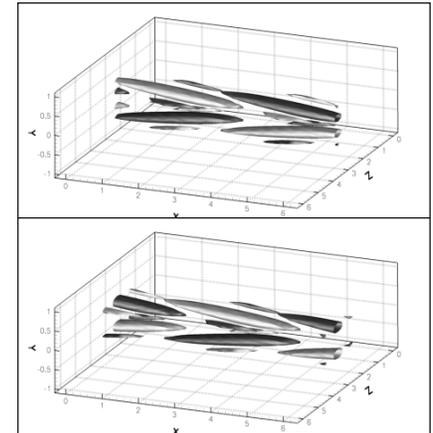


Localized actuator - POD model performance

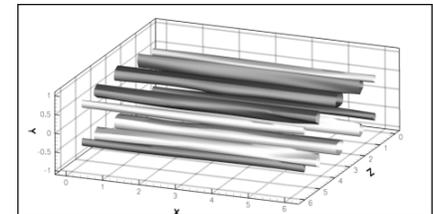
POD modes 1-3



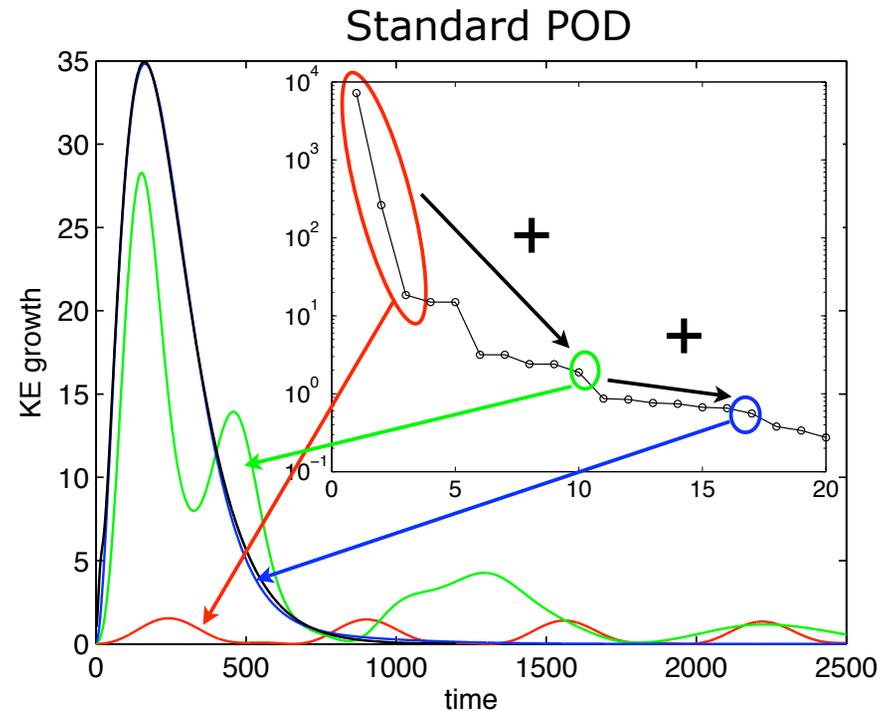
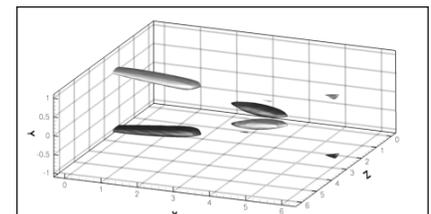
POD modes 4-5



POD mode 10

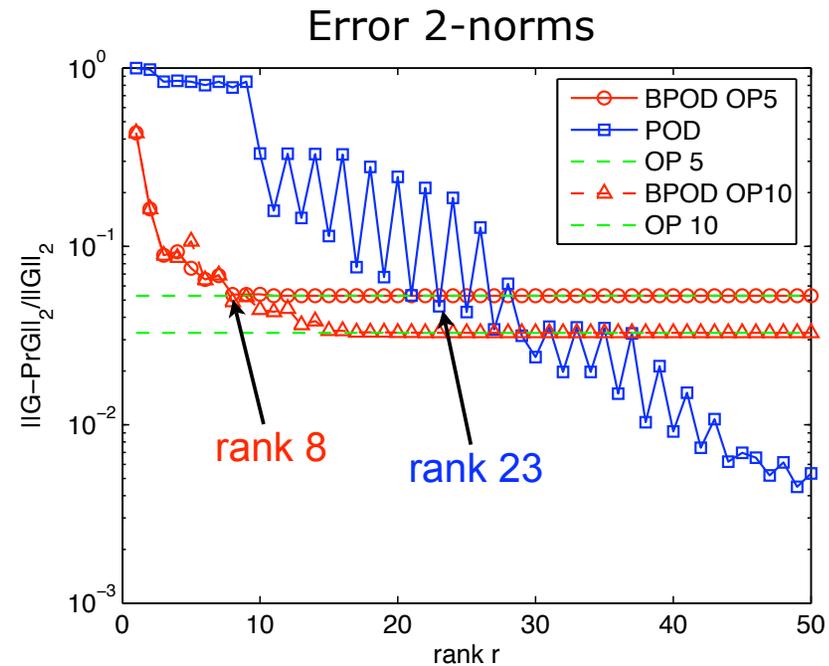
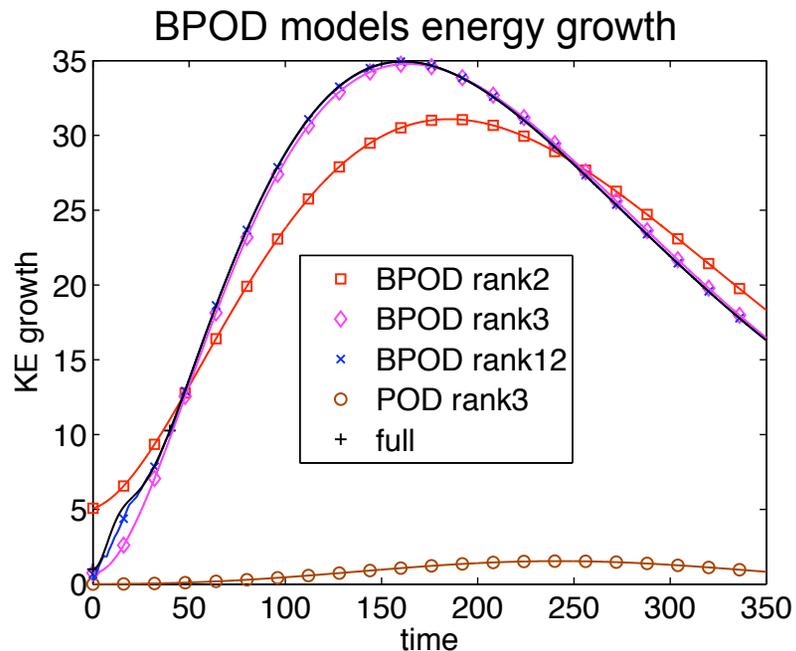


POD mode 17



- Some low-energy POD modes are very important for the system dynamics - can't naively use just the most energetic ones
- Pairs of modes corresponding to traveling structures not important for capturing energy growth
- For many POD low-order models, the output can have spurious oscillations due to the mode pairs

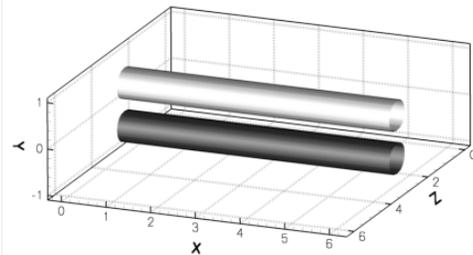
Localized actuator - BPOD impulse response



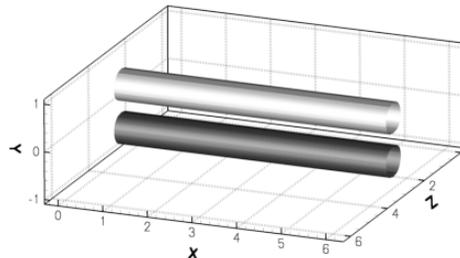
- Three-mode BPOD model excellent at capturing the energy growth
- Rank 8 BPOD model sufficient to correctly capture the dynamics of the first five POD modes, compared to at least 23 POD modes
- Inclusion of some POD modes significantly deteriorates performance (splitting of the pairs of oscillating modes)

Localized actuator - modes

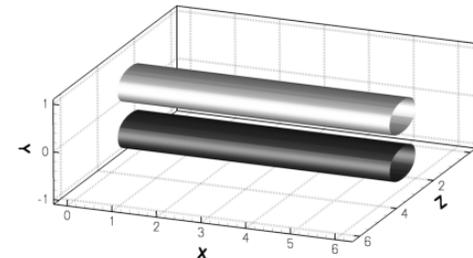
POD mode 1



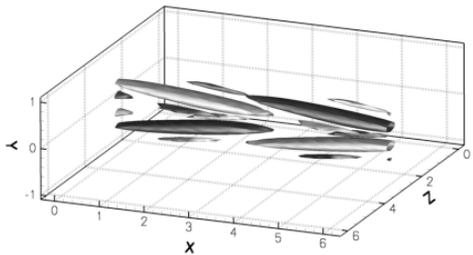
BPOD mode 1



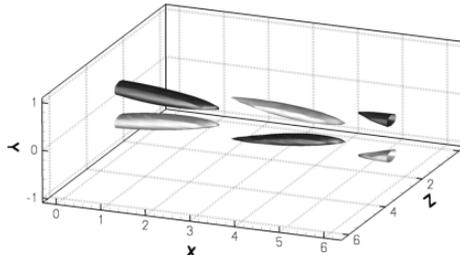
adjoint BPOD mode 1



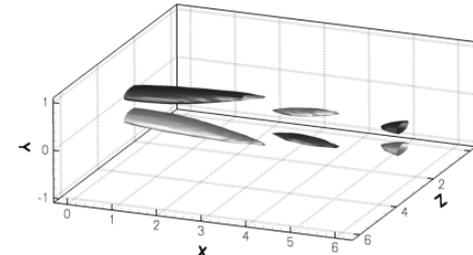
POD mode 4



BPOD mode 4



adjoint BPOD mode 4



BPOD and adjoint BPOD modes from OP5

Balancing modes and POD modes look similar but the adjoint modes are in general quite different => different dynamics of models

POD

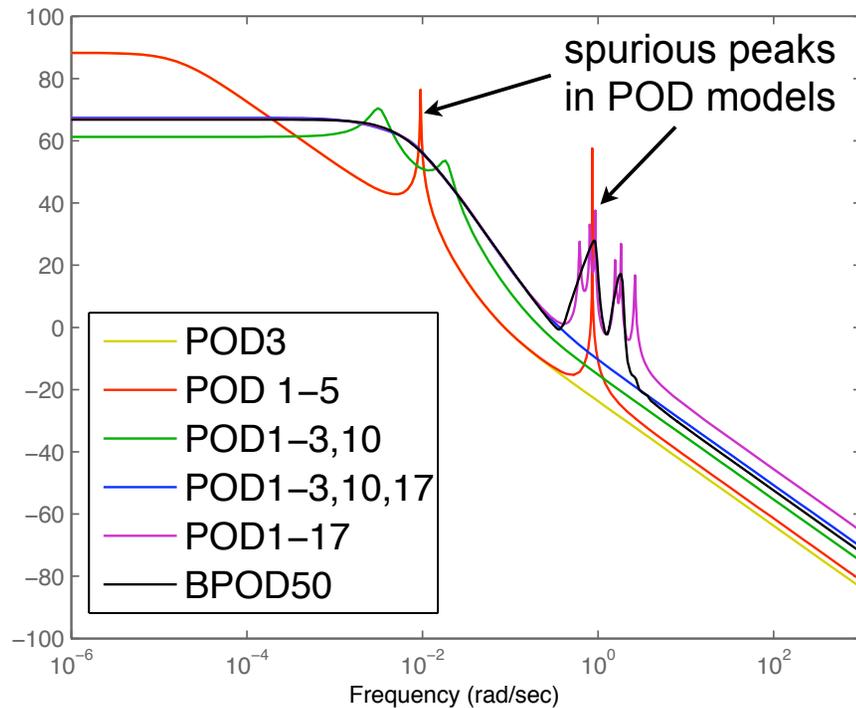
$$\dot{a}_j(t) = \langle \underline{\phi}_j, f(x) \rangle$$

BPOD

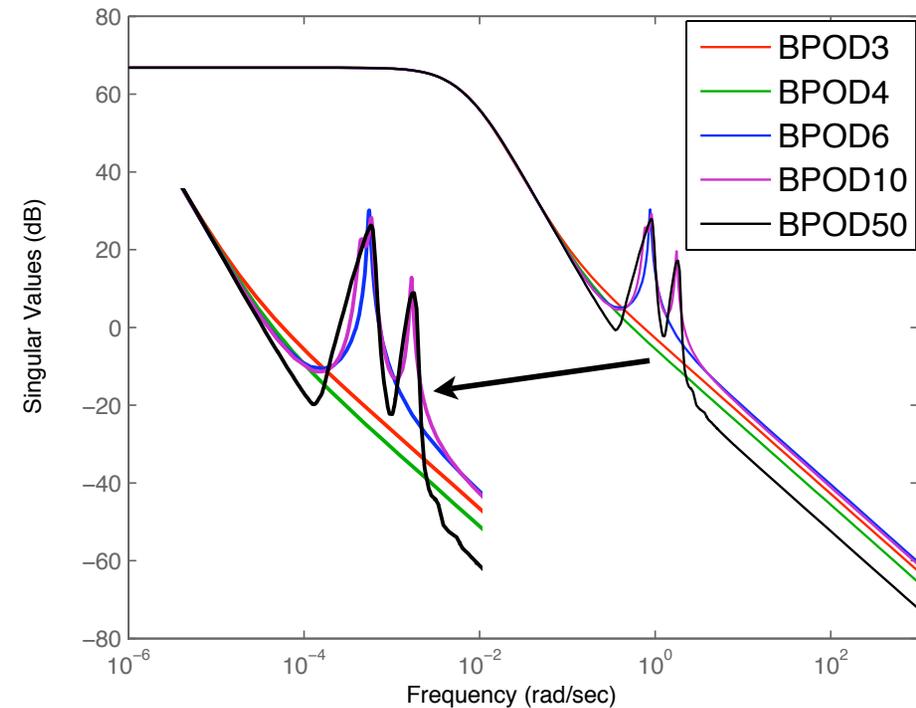
$$\dot{a}_j(t) = \langle \underline{\psi}_j, f(x) \rangle$$

Localized actuator - frequency response

POD singular value Bode plot

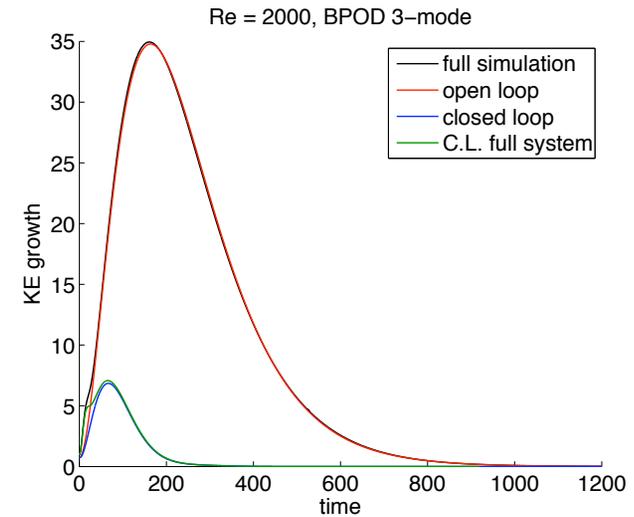
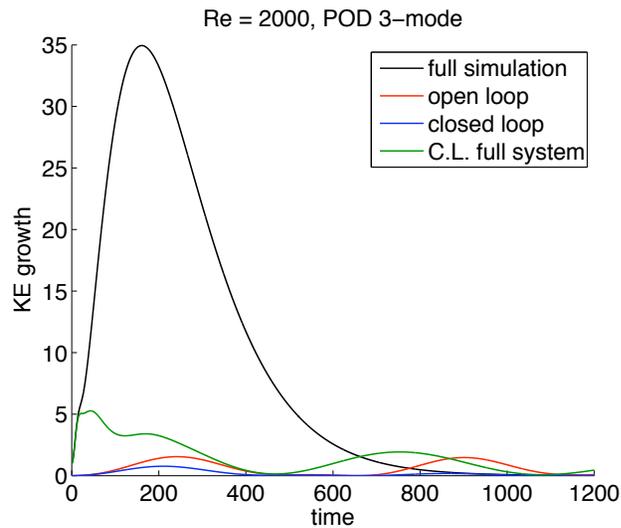


BPOD singular value Bode plot

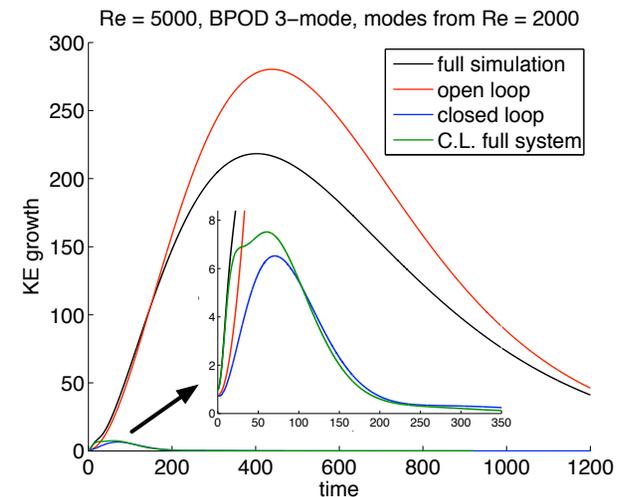


- BPOD 10-mode OP 50-mode model taken as 'full system'
- POD poorly captures low-pass behavior, spurious peaks
- Need pairs of BPOD modes to capture peaks

Closed-loop control - localized actuator

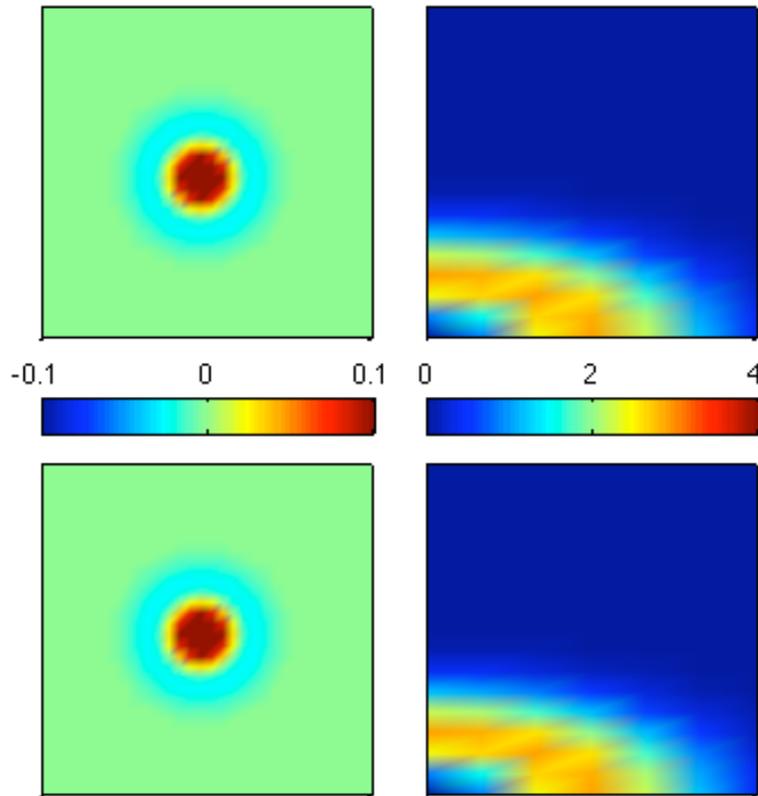


- Using the localized actuator to control a disturbance in channel center
- Standard LQR controller
- Using control gains from a 3-mode BPOD model reduces energy growth by a factor of 5
- BPOD works well in closed-loop at off-design condition (Re=5000 with modes from Re=2000)



Nonlinear Evolution of the Localized Perturbation

linear evolution of wall-normal velocity

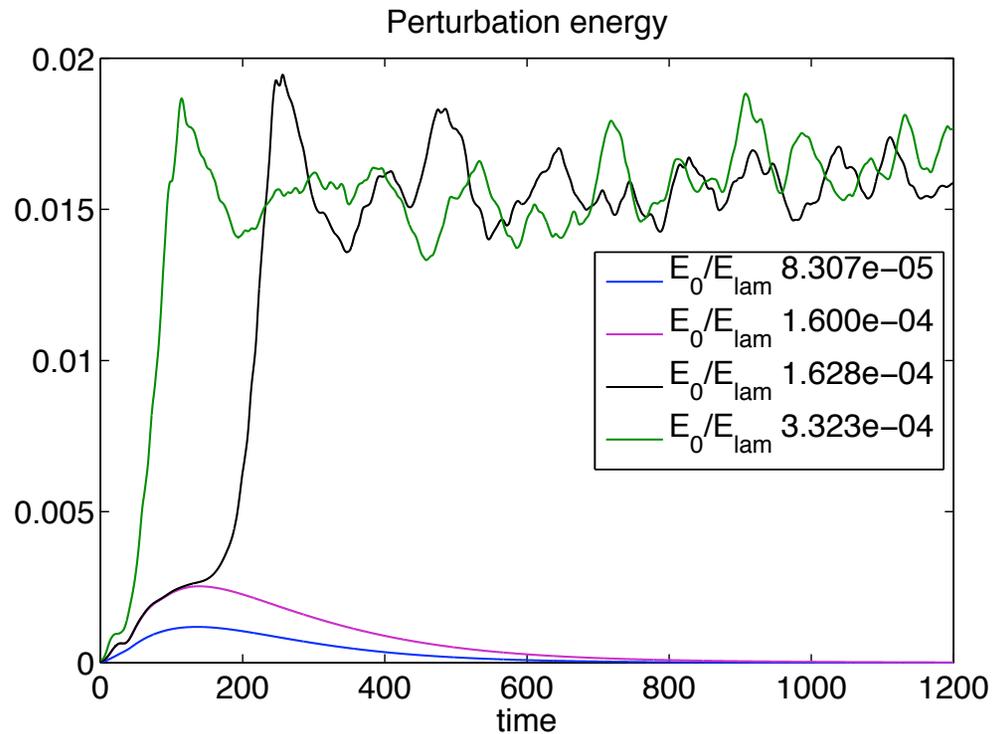


nonlinear evolution at $E_0/E_{lam} = 3.323 \times 10^{-4}$

- The spatial Fourier transform of the x,z plane at y=0 illustrates the perturbation evolution
- In the linear case the wavenumbers decay independently after the large transient growth
- $E_{lam} = 0.2667$ is the energy density of the mean laminar flow
- Transition for very small values of initial energy E_0
- The so-called β -cascade [Henningson et al, 1993] is observed in the nonlinear evolution - higher spanwise wavenumbers are introduced rapidly

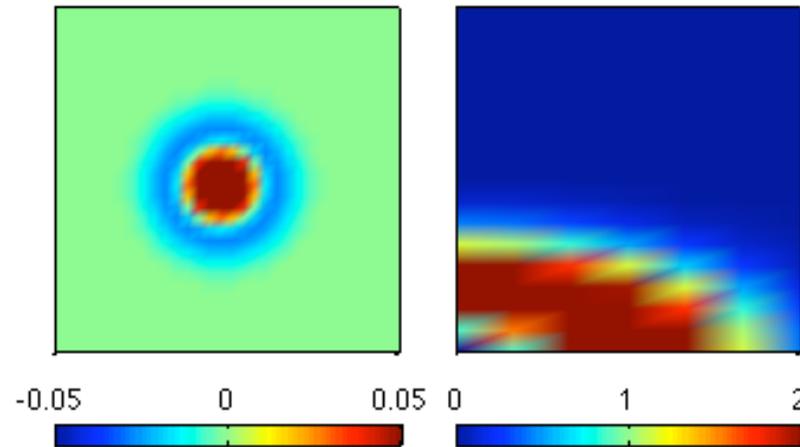
Delaying Transition Using Feedback Control

- Try to increase the transition threshold of a localized perturbation (after Reddy et al)
- The threshold is defined as the energy density of the initial perturbation above which the flow transitions to turbulence
- Threshold found to be at $E_0 = 1.614 \times 10^{-4}$ of the mean flow energy of the laminar profile, $E_{\text{lam}} = 0.2667$



Closed-Loop Control

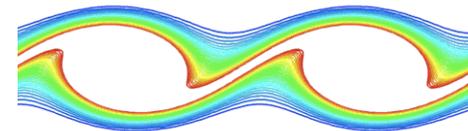
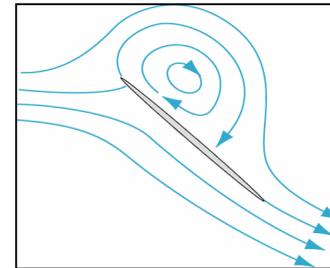
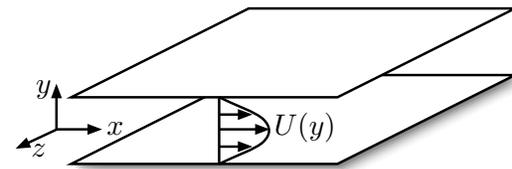
- The feedback gains computed using LQR for the linear system are used in a full nonlinear simulation with $E_0/E_{lam} = 3.323 \times 10^{-4}$
- An 'aggressive' controller ($R=0.1$ in LQR) manages to suppress the disturbance



- Explanation: the BPOD modes do not have components at high β , and are not able to suppress high betas once they arise, but the 'aggressive' controller suppresses low β wavenumbers so that the higher β 's emerge at very low amplitudes and decay linearly
- Transition threshold increased by a factor of 17 for $R=0.01$
- Work in progress: see how projection of full N-S equations onto linear BPOD modes will model the perturbation evolution, and possibly design a nonlinear controller

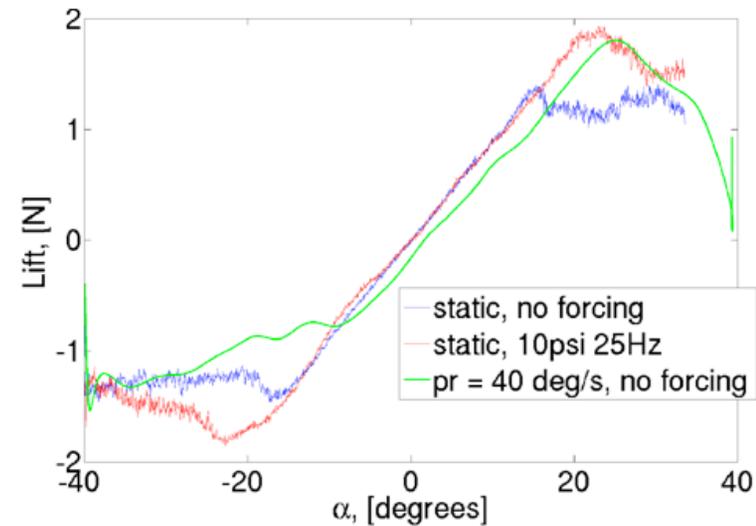
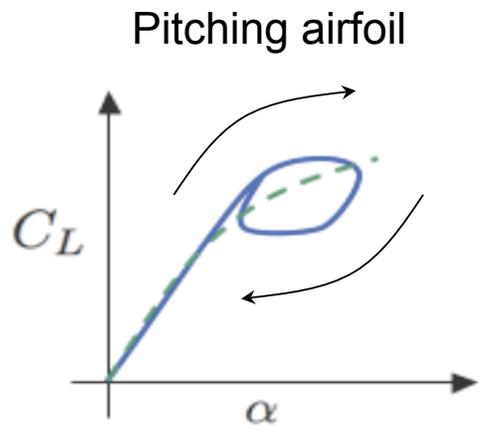
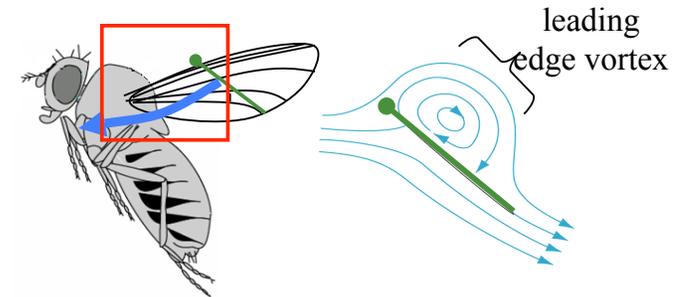
Outline

- Approximate balanced truncation using POD
 - Importance of inner product for Galerkin projection
 - Balanced truncation
 - Method of snapshots
- Applications
 - Linearized channel flow
 - Separating flow past an airfoil
- Dynamically scaling POD modes
 - Free shear layer
 - Scaled basis functions
 - Template fitting
 - Equations for the shear layer thickness



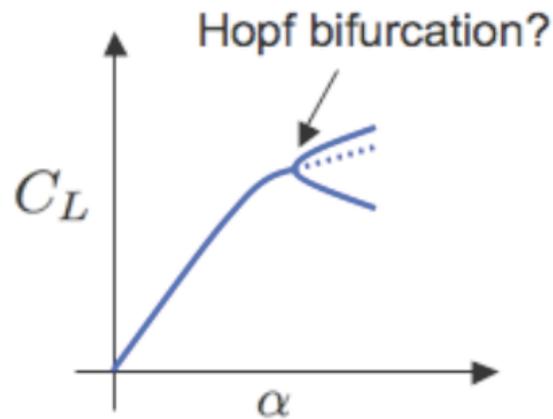
Motivation

- Leading edge vortices sometime provide high lift
- MURI goal: Stabilize these LEVs using feedback control
- High transient lift in pitching airfoils due to dynamic stall vortex

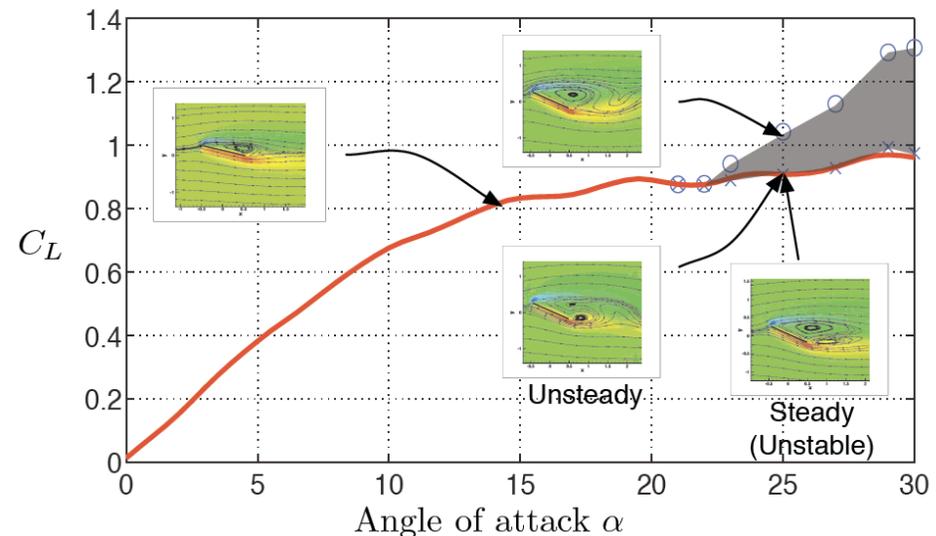


Dynamical behavior

- With increasing AoA, flow undergoes a Hopf bifurcation
- Reduced order models to stabilize unstable steady states at high AoAs



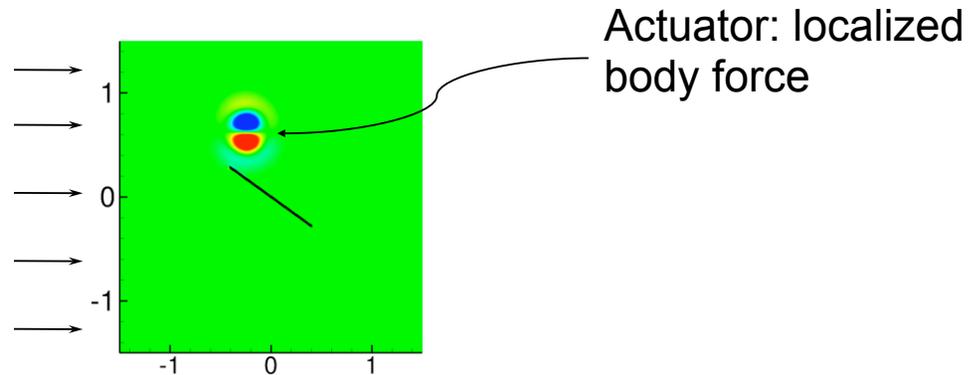
C_L at steady state



- Are there high-lift unstable steady states in low aspect ratio airfoils?

Model problem

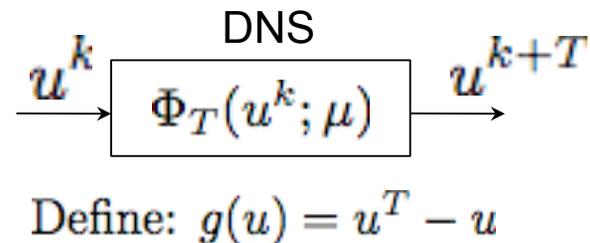
$Re = 100$
 $AoA = 25$ or 35



- A fast null-space based immersed boundary scheme for numerical simulations (T. Colonius and K. Taira, CMAME, 2007)

Steady state analysis

- Compute steady states using a wrapper around the DNS

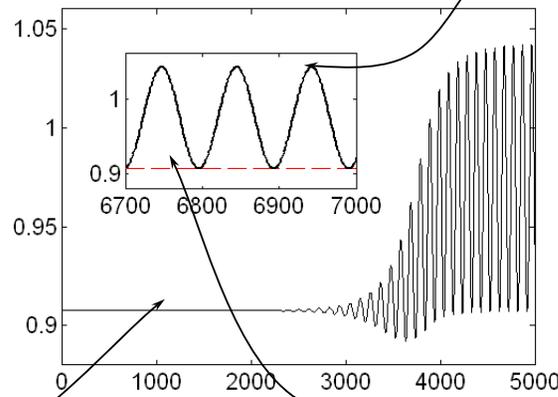
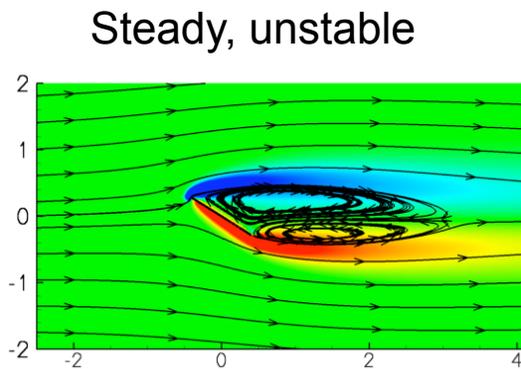


- Solve for zeroes of $g(u)$ using Newton-GMRES

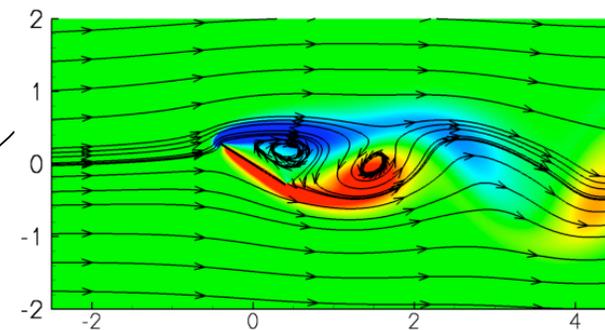
Barkley and Tuckerman,'99, Kelley, Kevrekidis, and Qiao,'02, Ahuja et al., '07

Unstable steady state, $AoA = 35$

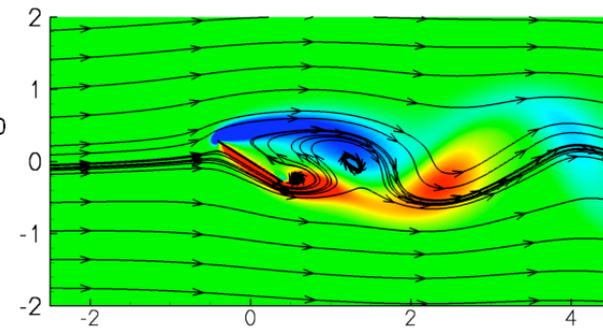
- Steady state lift close to the min. lift of the unsteady case
- No leading edge vortex
- Trailing edge vortex causes reduced lift



Unsteady, max lift

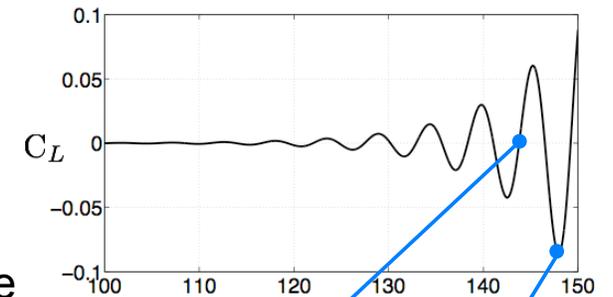


Unsteady, min lift

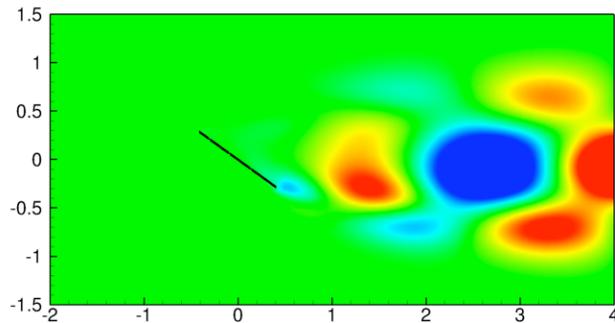
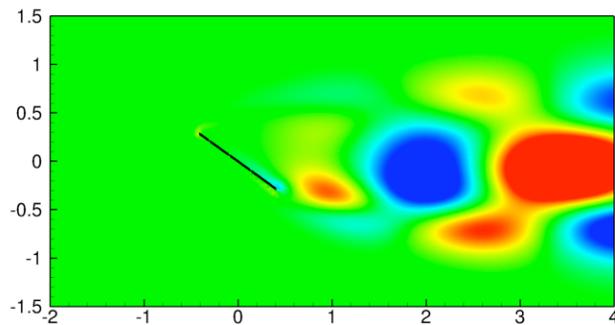


Linear stability analysis

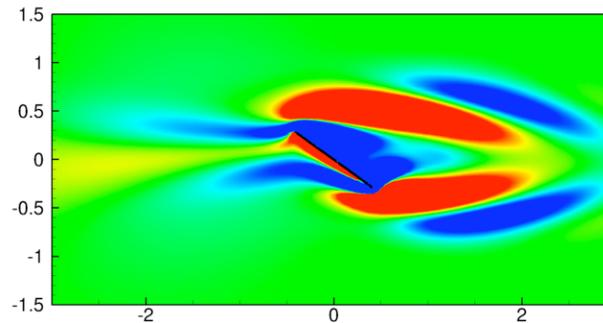
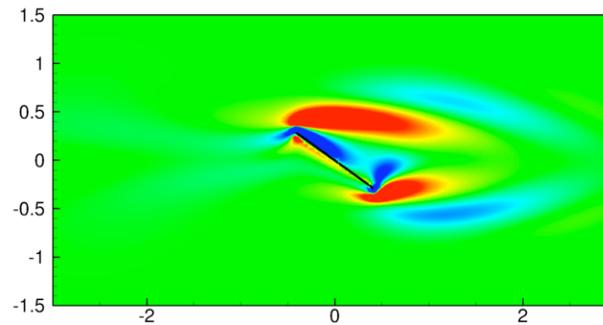
- Find the basis spanning the unstable eigenspace of the linearized and adjoint flows
- Run the linear simulations with a zero initial condition + 10^{-8} random noise



Right eigen-space



Left eigen-space

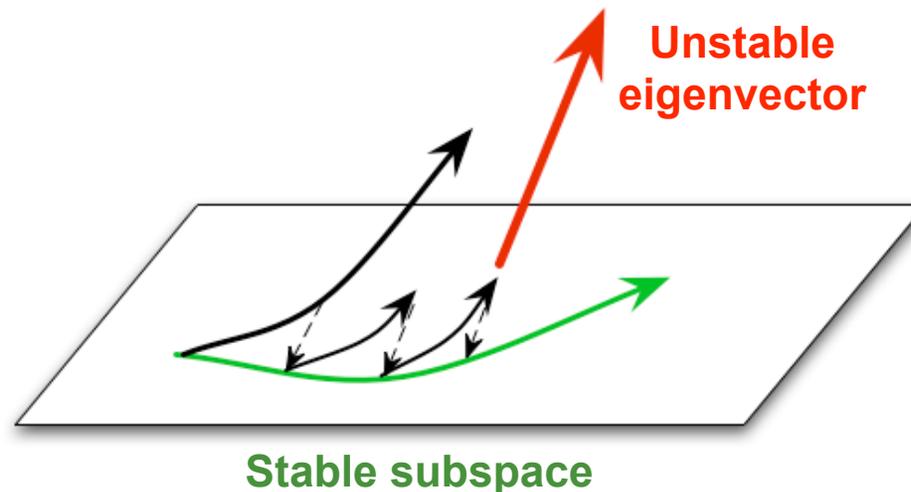


Reduced-order models for unstable systems

- Decouple stable and unstable subspaces
- Obtain balancing transformation for the stable subspace

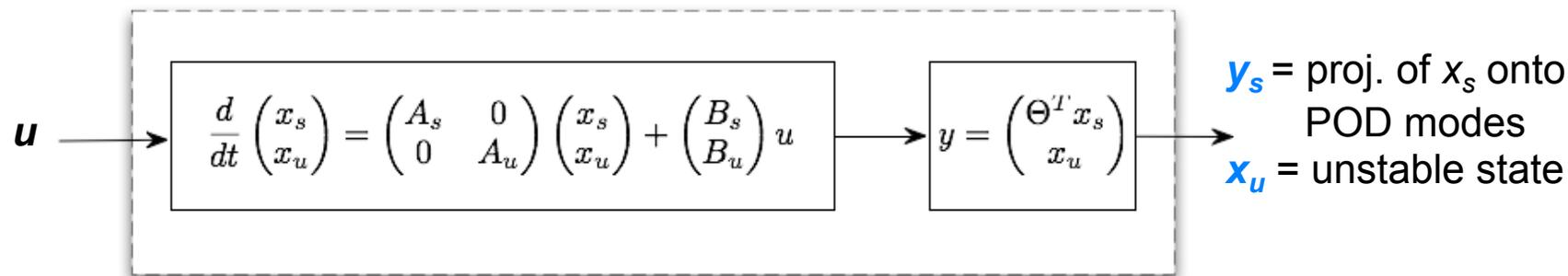
$$\frac{d}{dt} \begin{pmatrix} x_s \\ x_u \end{pmatrix} = \begin{pmatrix} A_s & 0 \\ 0 & A_u \end{pmatrix} \begin{pmatrix} x_s \\ x_u \end{pmatrix} + \begin{pmatrix} B_s \\ B_u \end{pmatrix} u$$

- Snapshot based procedure: project out the unstable component at each time step



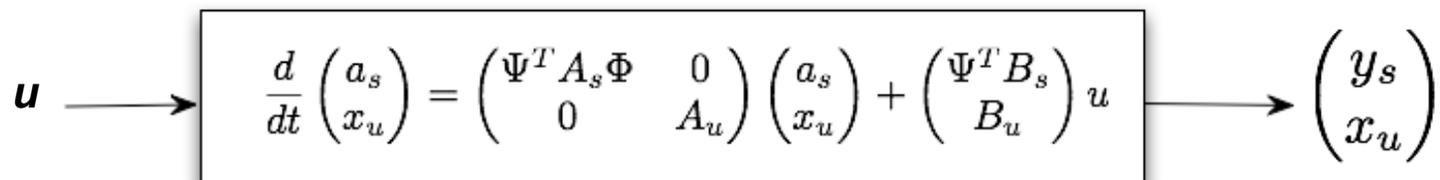
Model reduction: unstable system

Linearized NS eqns, 10^5



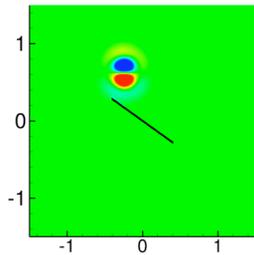
$$x_s = \Phi a_s$$
$$\Psi^T \Phi = I$$

Reduced order model, 10-50 eqns.

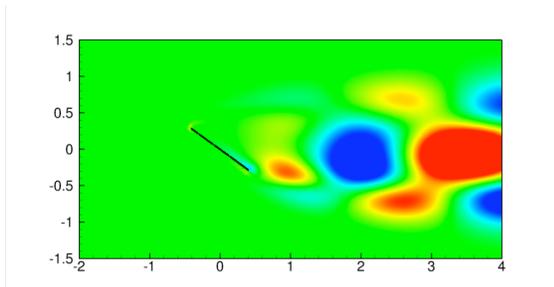


Impulse response: stable subspace

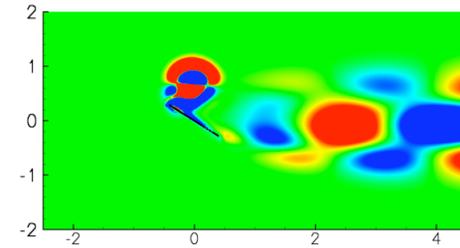
- Project out the unstable component from the initial condition



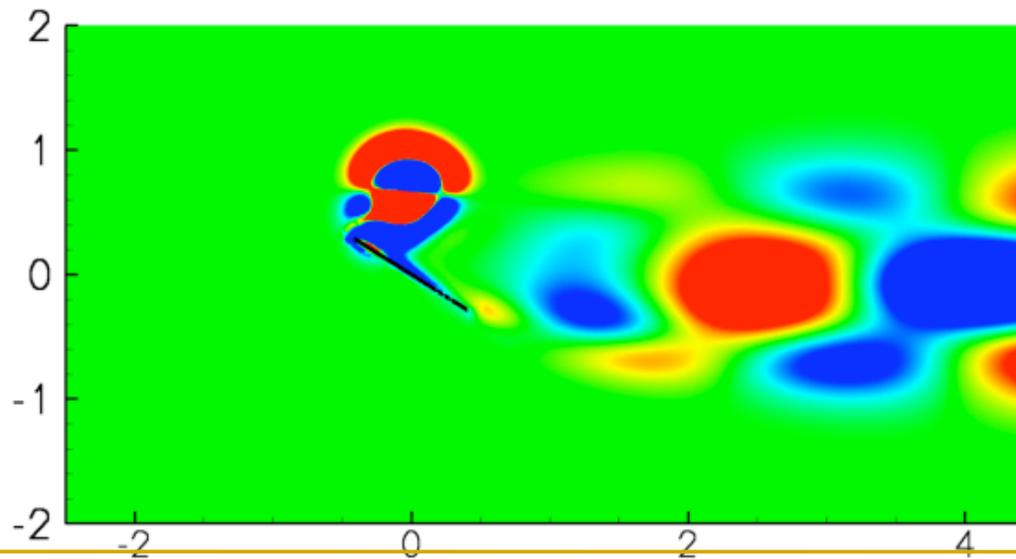
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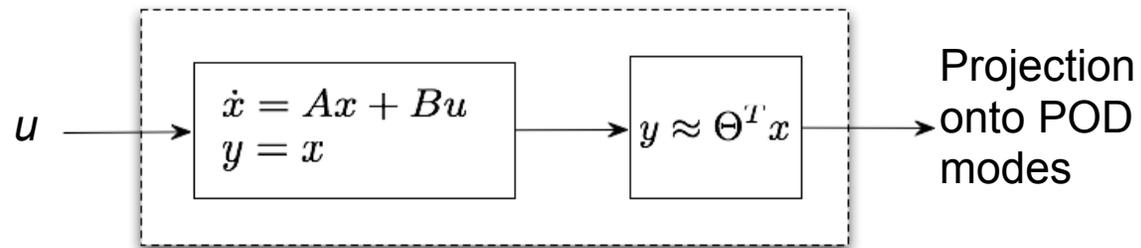
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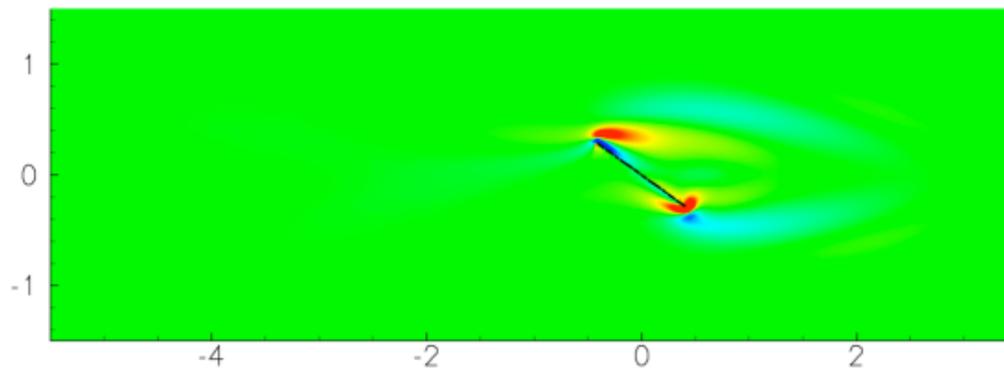
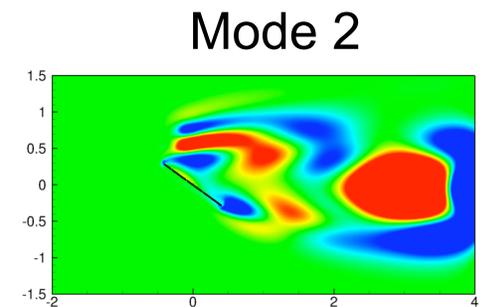
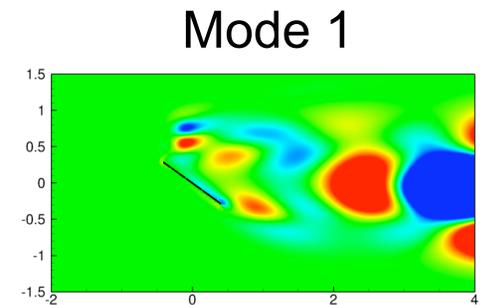
Vorticity contours:
Positive in red and
negative in blue



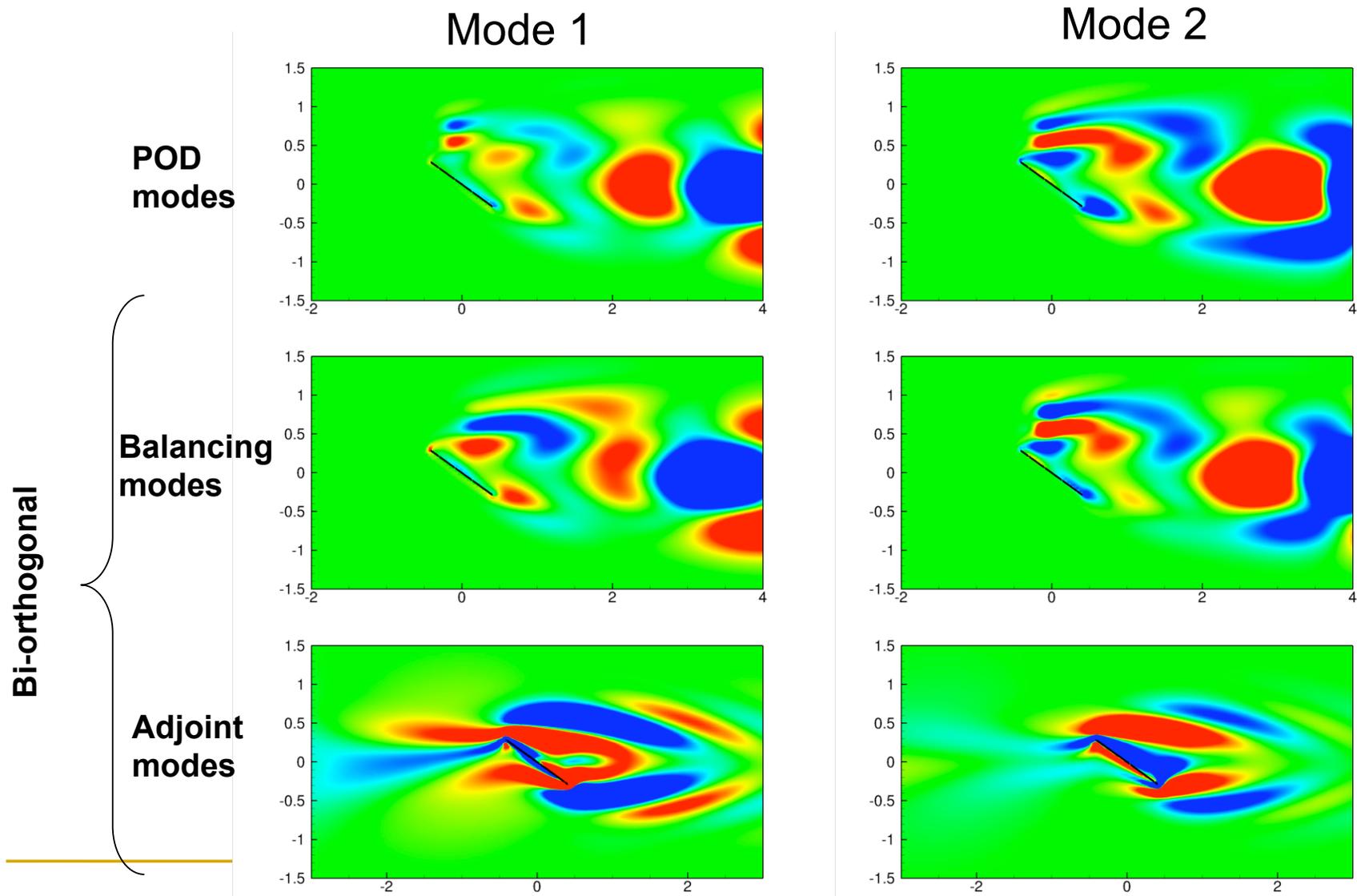
Adjoint impulse response



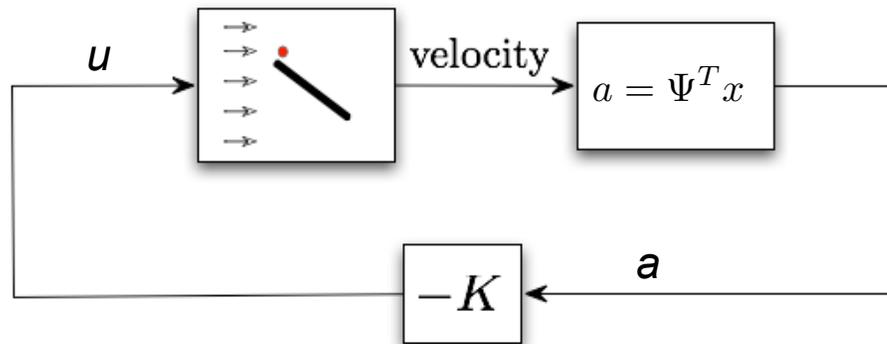
- Four POD modes capture 95% energy
- Adjoint solves with these POD modes as initial conditions



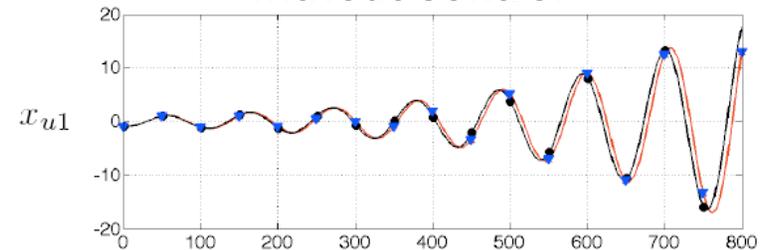
Balancing modes: stable subspace



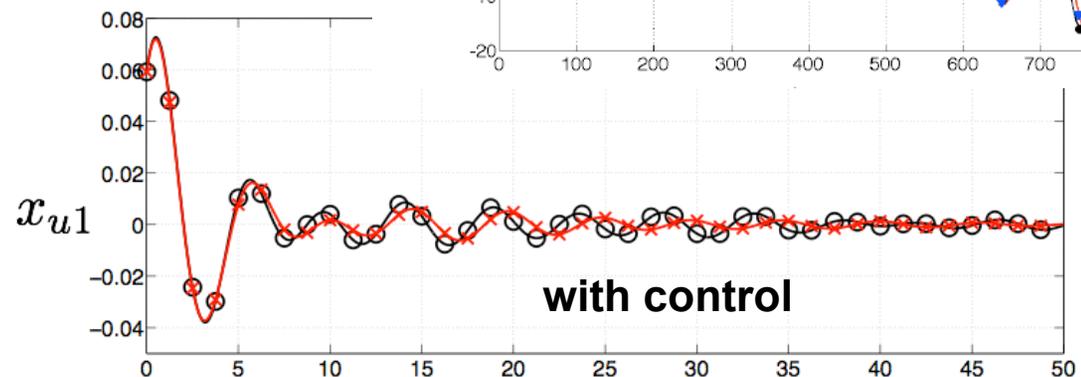
Model results: controlled case



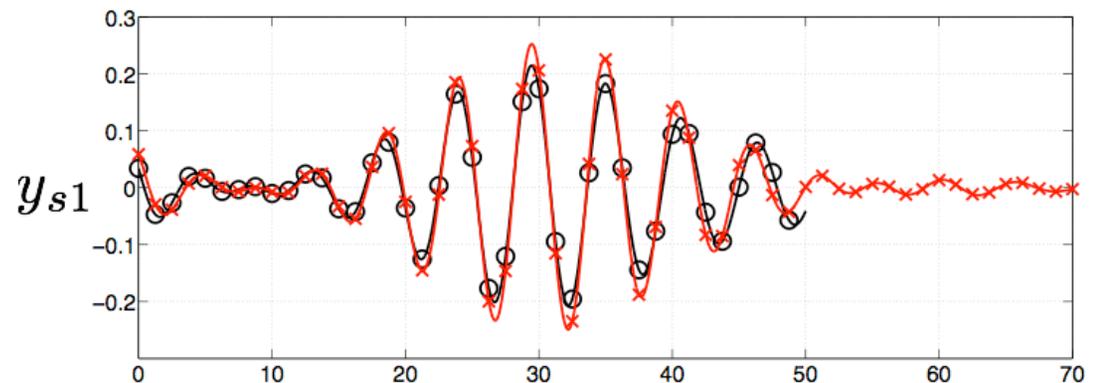
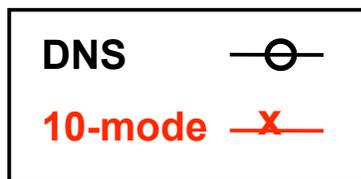
without control



- Control based on a 10-mode model
- Gain K using LQR

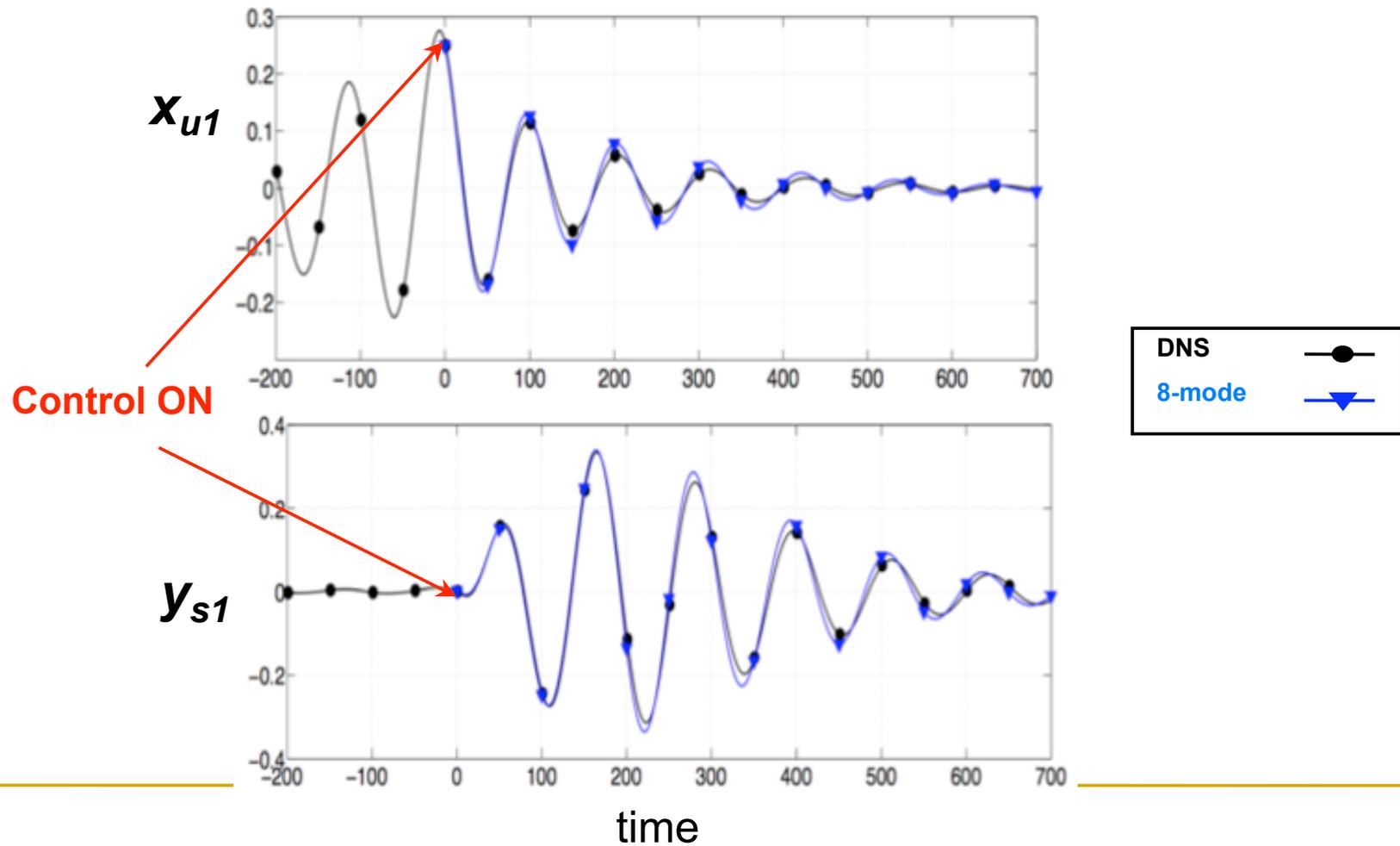


with control



Control in full nonlinear system: close to steady state

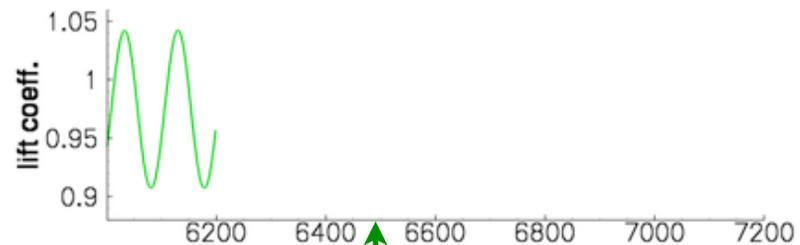
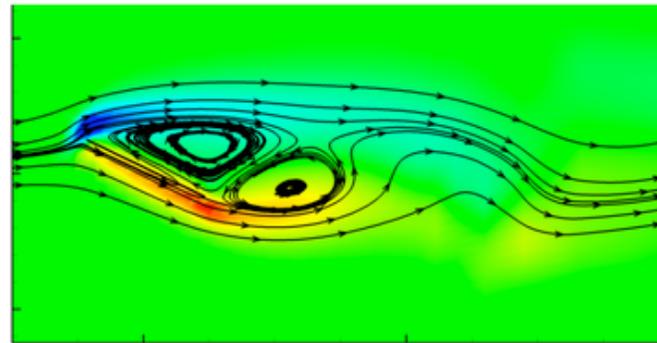
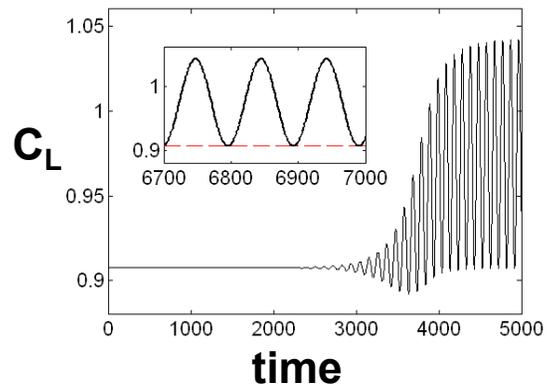
Results of an 8-mode model



Feedback stabilization at $AoA=25$

- Full state feedback
- Large domain of attraction even in the full NL system
- Controller suppresses the vortex shedding

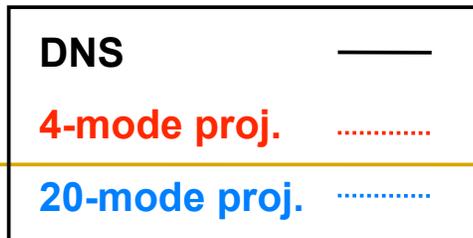
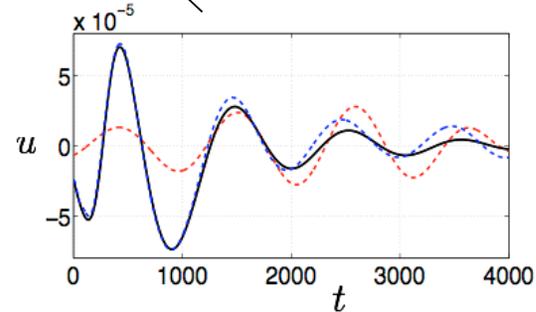
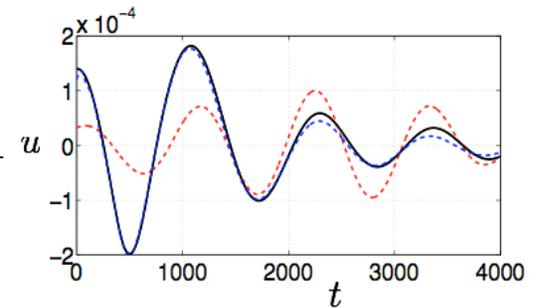
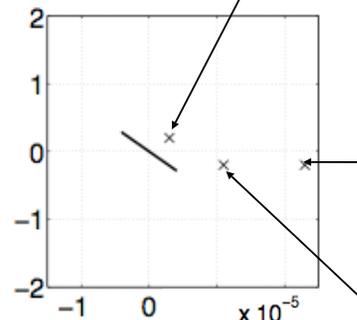
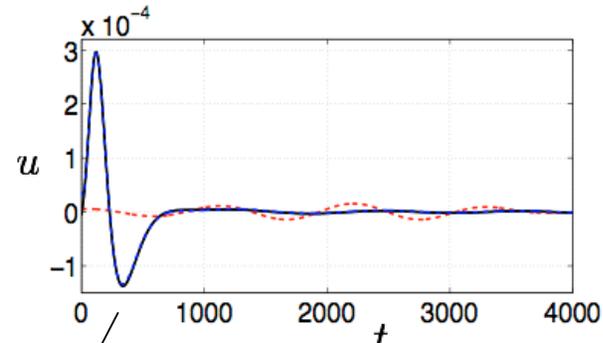
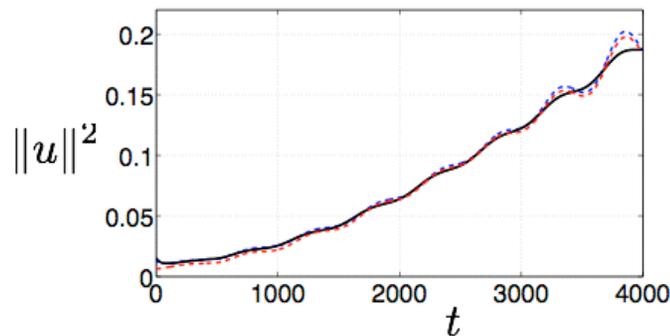
No control



Control on

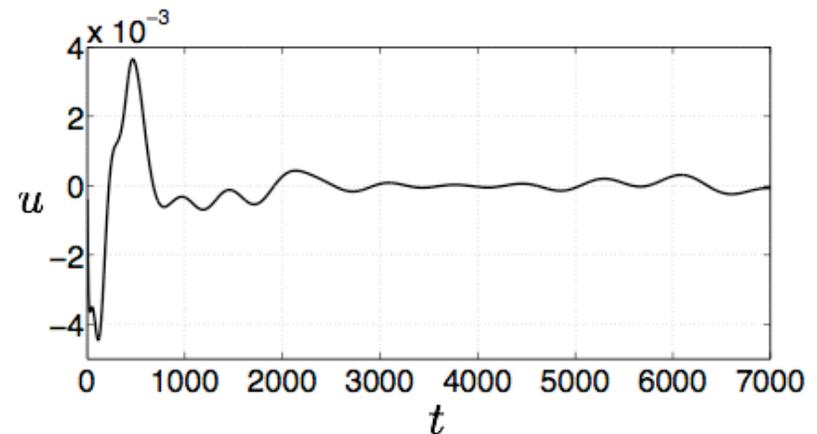
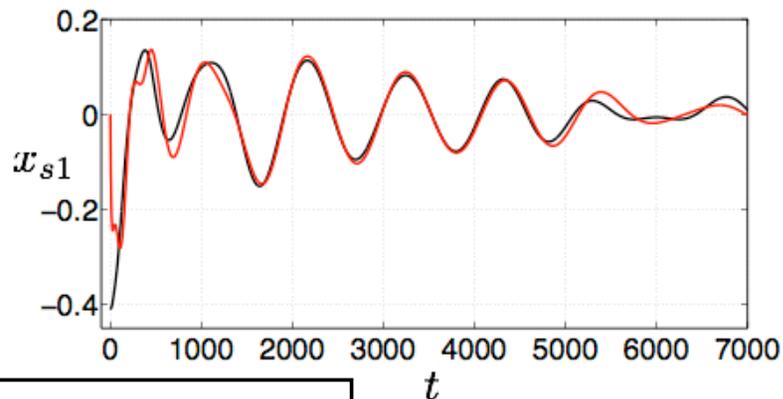
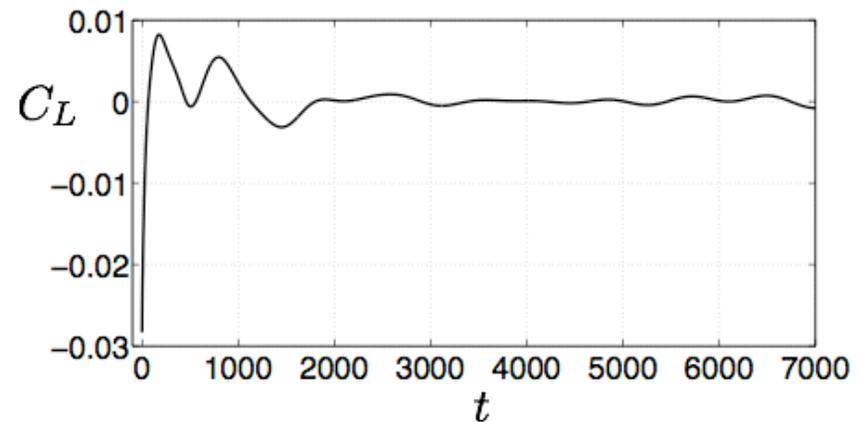
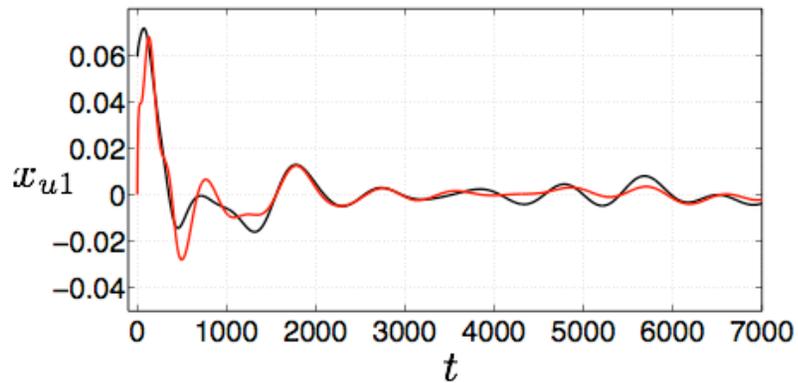
Observer design: velocity sensors

- 3 velocity sensors
- Compare projections onto 4 and 20 POD modes
- L2-norm looks similar, but the velocities at sensor locations are poorly captured by 4 POD modes



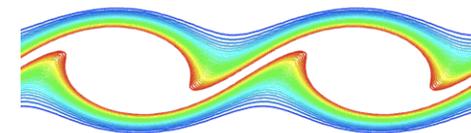
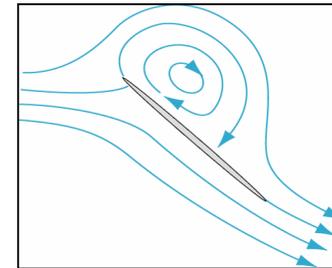
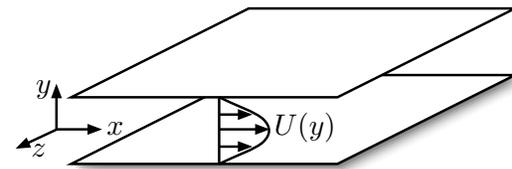
Observer based control

- Observer gain obtained using LQG
- Compensator stabilizes the steady state, but there is residual noise due to the errors in modeling the system and the measurements



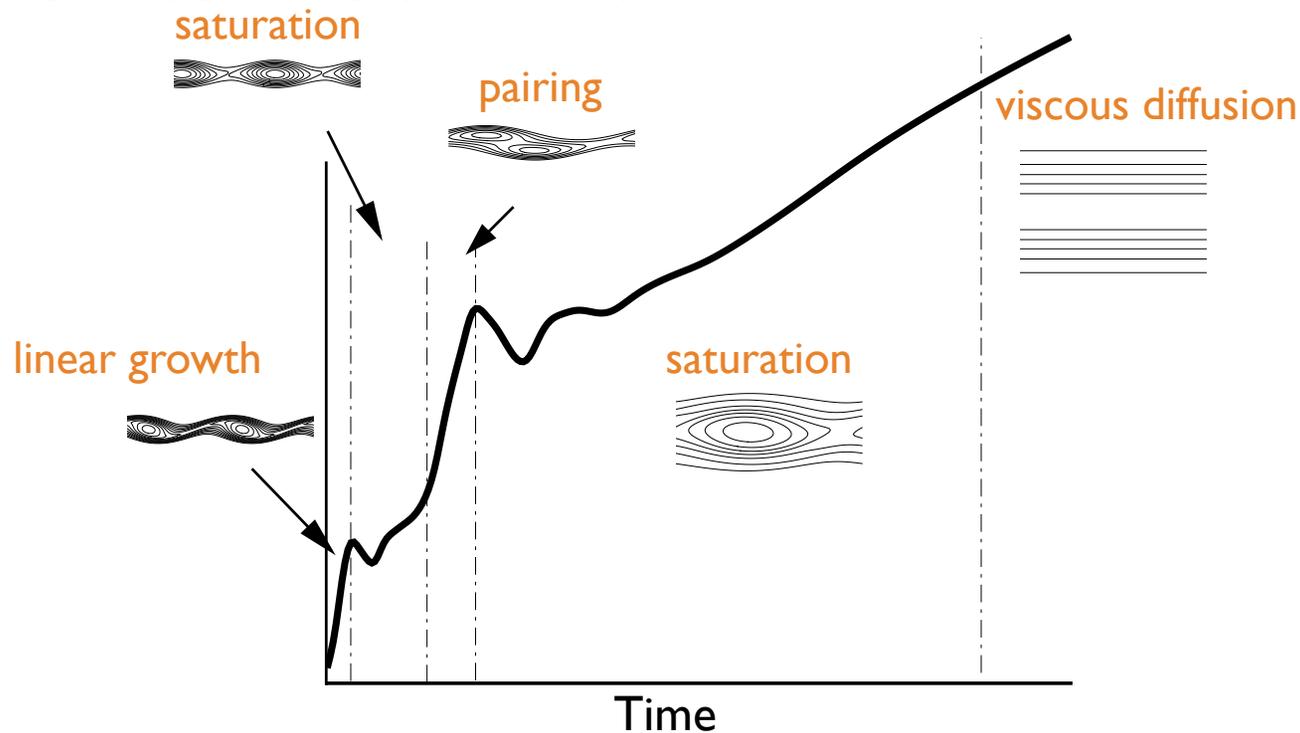
Outline

- Approximate balanced truncation using POD
 - Importance of inner product for Galerkin projection
 - Balanced truncation
 - Method of snapshots
- Applications
 - Linearized channel flow
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- Dynamically scaling POD modes
 - Free shear layer
 - Scaled basis functions
 - Template fitting
 - Equations for the shear layer thickness



Modeling free shear layers

- Evolution history of thickness for temporal shear layer (spatially periodic):



- Model initial linear growth, saturation, pairing, and eventual viscous diffusion



Methodology

- Scale POD modes dynamically in y direction to account for shear layer spreading
- Scaling invariants:
 - divergence of velocity field
 - inner product
- Key idea: template fitting
- Main result: an equation for the shear layer spreading rate:
 - as usual, also get equations for time coefficients of POD modes



Scaling basis functions

- Write solution in scaled reference frame

$$\mathbf{q} = (u, v)$$

$$\mathbf{q}(x, y, t) = G(g)\tilde{\mathbf{q}}(x, g(t)y, t)$$

- Choose $G(g) = \begin{bmatrix} 1 & 0 \\ 0 & 1/g \end{bmatrix}$: $\text{div } \mathbf{q} = \text{div } \tilde{\mathbf{q}}$
- Expand scaled variable $\tilde{\mathbf{q}}$ in terms of POD modes

$$\tilde{\mathbf{q}}(x, y, t) = \mathbf{u}_0(y) + \sum_{j=1}^n a_j(t)\varphi_j(x, y)$$

- Advantage of the scaling: capture similar-looking structures as shear layer spreads
- Advantage of divergence-invariant mapping: auto-satisfy continuity equation; simplify pressure term



Template fitting

- How do we choose the scaling $g(t)$?
 - Choose $g(t)$ so that $\tilde{\mathbf{q}}(x, y, t)$ lines up best with a preselected **template** (here, the base flow):

$$\frac{d}{ds} \Big|_{s=0} \|\tilde{\mathbf{q}}(x, y, t) - \mathbf{u}_0(x, h(s)y)\|^2 = 0$$

for any curve $h(s) > 0$ with $h(0) = 1$

- This means the scaled solution $\tilde{\mathbf{q}}(x, y, t)$ satisfies

$$\left\langle y \frac{\partial \mathbf{u}_0}{\partial y}, \tilde{\mathbf{q}} - \mathbf{u}_0 \right\rangle = 0$$

- Geometrically, the set of all “properly scaled” functions $\tilde{\mathbf{q}}$ is an affine space through \mathbf{u}_0 and orthogonal to $y \partial_y \mathbf{u}_0$
- This enables one to write dynamics for how the thickness $g(t)$ evolves

$$\frac{\dot{g}}{g} = \frac{\langle f_g^1(\tilde{u}), y \partial_y u_0 \rangle}{\langle y \partial_y \tilde{u}, y \partial_y u_0 \rangle}$$



Equation for evolution of the thickness

- How does $g(t)$ evolve in time?
 - We have a constraint ($\tilde{\mathbf{q}}(x, y, t)$ lines up best with template \mathbf{u}_0):

$$\left\langle y \frac{\partial \mathbf{u}_0}{\partial y}, \tilde{\mathbf{q}} - \mathbf{u}_0 \right\rangle = 0$$

- Differentiate:

$$\left\langle y \frac{\partial \mathbf{u}_0}{\partial y}, \frac{\partial \tilde{\mathbf{q}}}{\partial t} \right\rangle = 0$$

- Use equations of motion

$$\frac{\partial \tilde{\mathbf{q}}}{\partial t} = f_g(\tilde{\mathbf{q}}) - \frac{\dot{g}}{g} y \frac{\partial \tilde{\mathbf{q}}}{\partial y} - G(1/g) \dot{G}(g, \dot{g}) \tilde{\mathbf{q}}(x, y, t)$$

- This gives an equation for g :

$$\frac{\dot{g}}{g} = \frac{\langle f_g^1(\tilde{u}), y \partial_y u_0 \rangle}{\langle y \partial_y \tilde{u}, y \partial_y u_0 \rangle}$$



Galerkin equations for the shear layer

- Equation for the POD mode coefficients:

- retain only modes $k=1, n=1$ and 2 :

$$\dot{a}_{1,1} = \frac{g^2 c_{11g} + c_{11}}{g^2 n_{1g} + n_1} a_{1,1} + \frac{g^2 c_{12g} + c_{12}}{g^2 n_{1g} + n_1} a_{1,2} + \frac{1}{\text{Re}} \left[-\left(\frac{2\pi}{L}\right)^2 + \frac{g^2 d_{1g} + d_1}{g^2 n_{1g} + n_1} g^2 \right] a_{1,1} + \frac{g^2 e_{1g} + e_1}{g^2 n_{1g} + n_1} \frac{\dot{g}}{g} a_{1,1},$$

$$\dot{a}_{1,2} = \frac{g^2 c_{21g} + c_{21}}{g^2 n_{2g} + n_2} a_{1,1} + \frac{g^2 c_{22g} + c_{22}}{g^2 n_{2g} + n_2} a_{1,2} + \frac{1}{\text{Re}} \left[-\left(\frac{2\pi}{L}\right)^2 + \frac{g^2 d_{2g} + d_2}{g^2 n_{2g} + n_2} g^2 \right] a_{1,2} + \frac{g^2 e_{2g} + e_2}{g^2 n_{2g} + n_2} \frac{\dot{g}}{g} a_{1,2},$$

- Equation for the scaling g :

$$\dot{g} = \frac{c_{01}}{n_0} a_{1,1} a_{1,1}^* g + \frac{c_{02}}{n_0} a_{1,2} a_{1,2}^* g + \frac{c_{03}}{n_0} a_{1,1} a_{1,2}^* g + \frac{c_{04}}{n_0} a_{1,2} a_{1,1}^* g + \frac{1}{\text{Re}} \frac{d_0}{n_0} g^3$$

- Retaining modes $k=1$ and $2, n=1$ and 2 also tractable, but messy
- Use inner product that is preserved under scaling:

$$\langle \tilde{\mathbf{q}}_1, \tilde{\mathbf{q}}_2 \rangle_g = \int_{\Omega} \left(\frac{1}{g} \tilde{u}_1 \tilde{u}_2 + \frac{1}{g^3} \tilde{v}_1 \tilde{v}_2 \right) dx dy$$



Results

- Base flow with small perturbation

- Base flow: $u_0 = U_c \operatorname{erfc}(\eta), \quad \eta = \frac{-y}{2g} \sqrt{\frac{\operatorname{Re}}{t_0}}$

- Perturbation is along the unstable eigenfunction of the linearized problem

- Consider three separate cases

- No perturbation: viscous growth
- Initial perturbation with $k=1$: vortex roll-up
- Initial perturbation with $k=2$:
 - vortex roll-up
 - pairing
 - $k=1$ mode arises through pairing

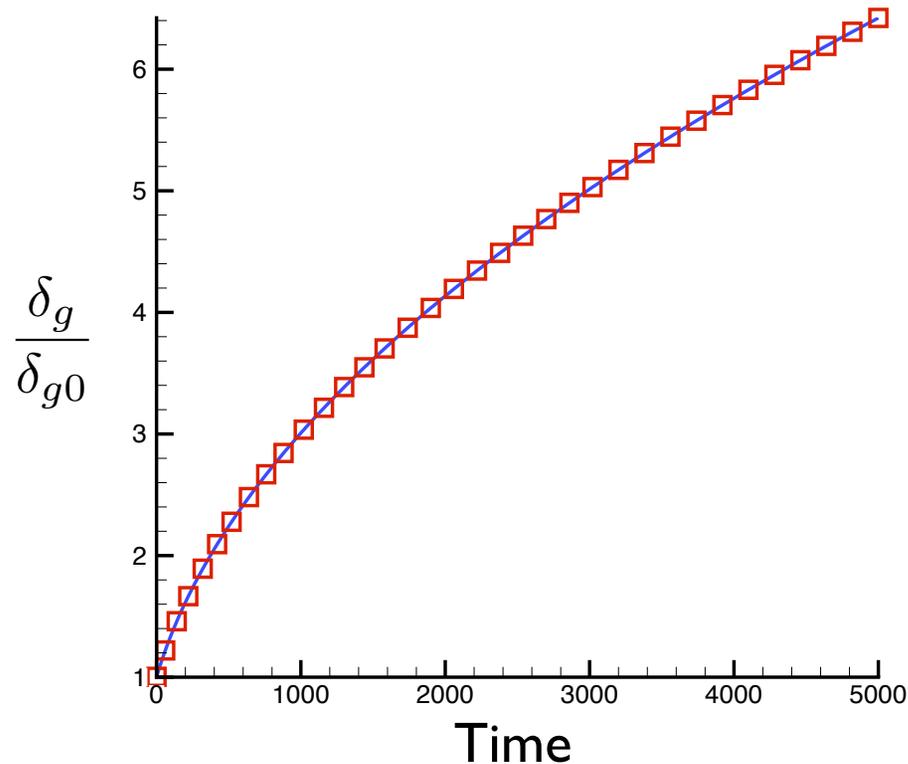


Model results: $k=0$

- Only one equation left for g :

$$\dot{g} = \frac{1}{\text{Re}} \frac{d_0}{n_0} g^3 \quad \Longrightarrow \quad \dot{g} = -\frac{g^3}{2t_0} \quad \Longrightarrow \quad g(t) = \sqrt{\frac{t_0}{t}}$$

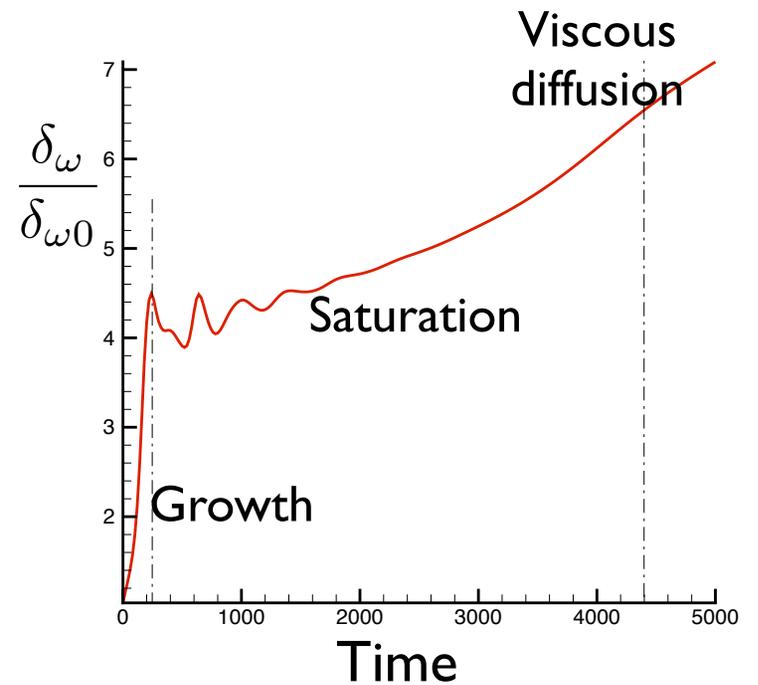
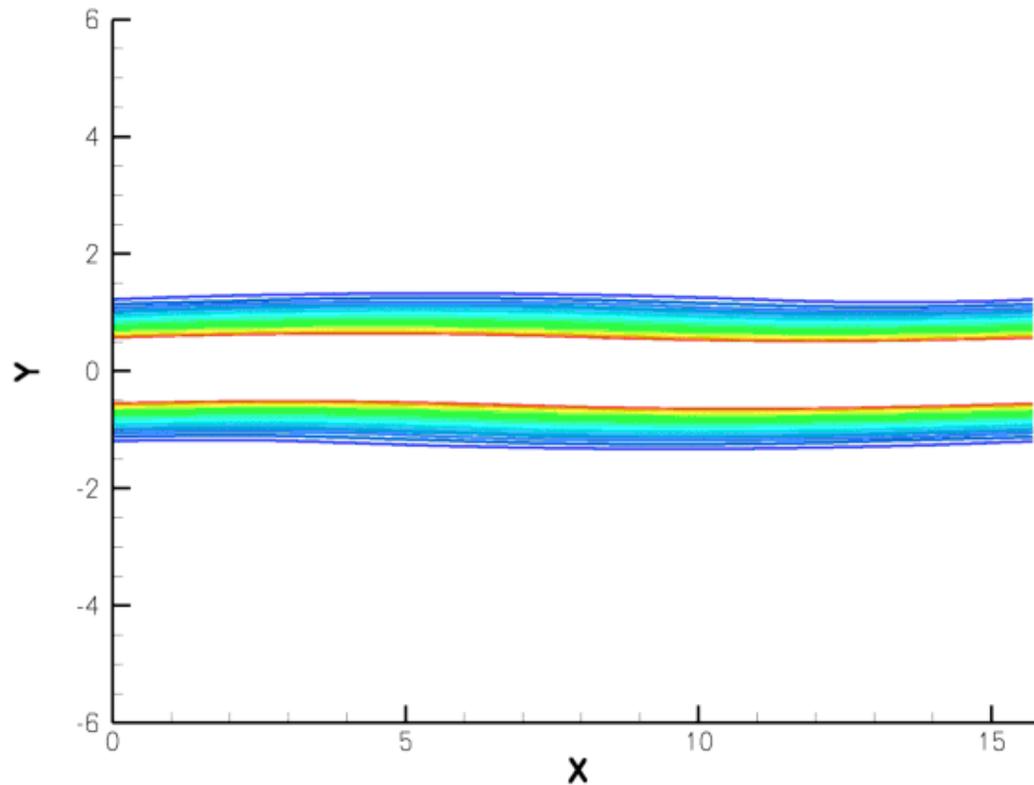
- Recovers exact theoretical growth rate for Stokes problem:



Movie of DNS

- Initial condition with $k=1$ ($Re = 200$)

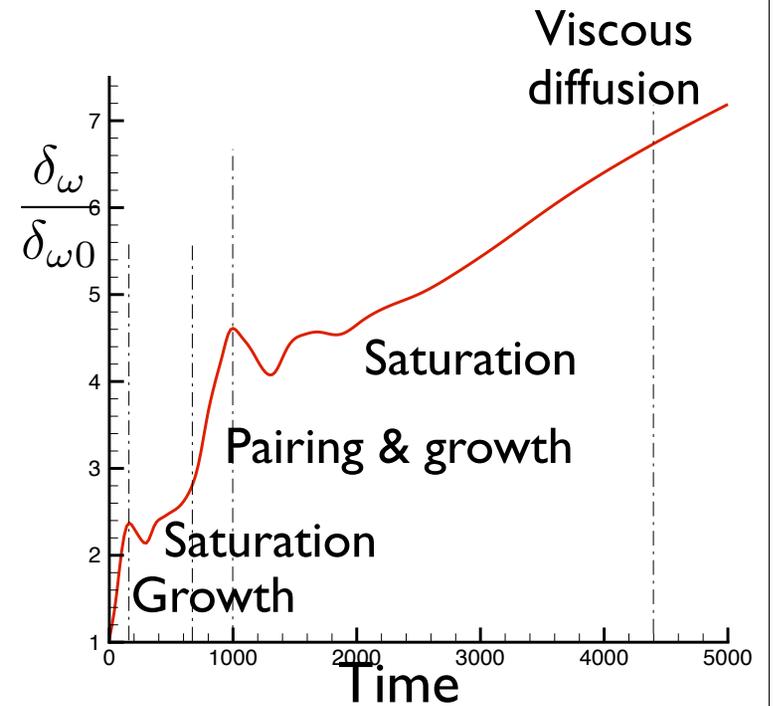
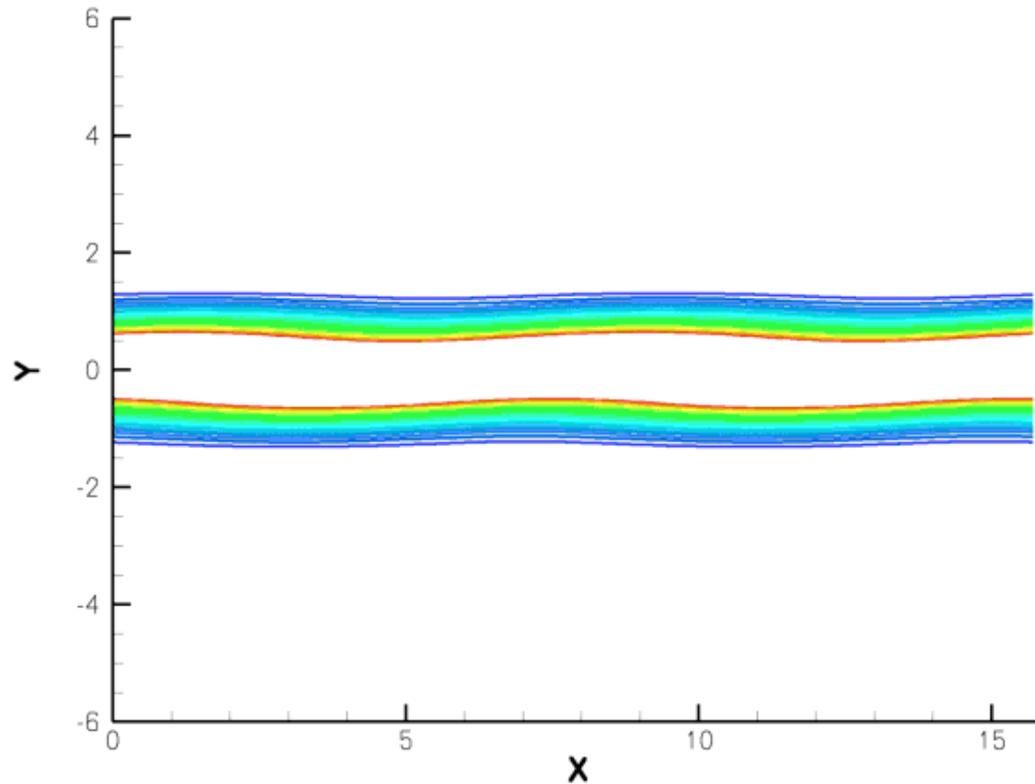
$k = 1$ simulation



Movie of DNS

- Initial condition with $k=2$ ($Re = 200$)

$k = 2$ simulation



POD modes

- Energy contained in modes ($k=1$ initial condition)

(k,n)	lambda	Energy (%)
(1,1)	130.3	91.0
(1,2)	6.8	4.8
(2,1)	4.5	3.1
all $k=0$		0.4

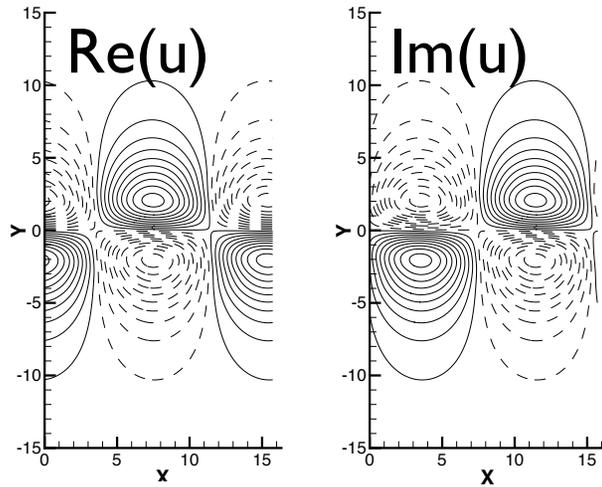
- Zero mode contains very little energy - scaling was effective at removing the mean spreading



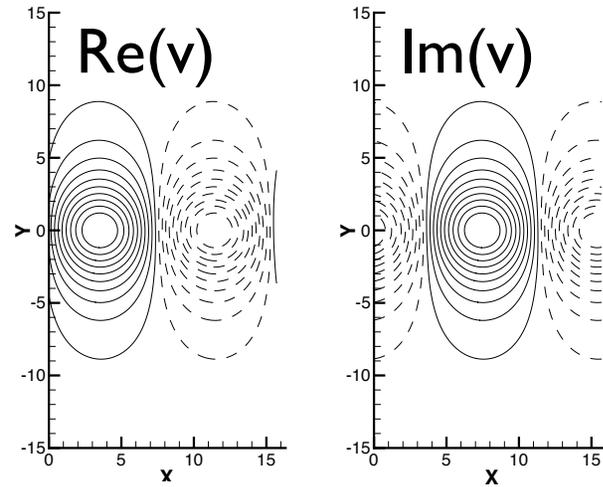
POD modes

- Initial condition with $k=1$

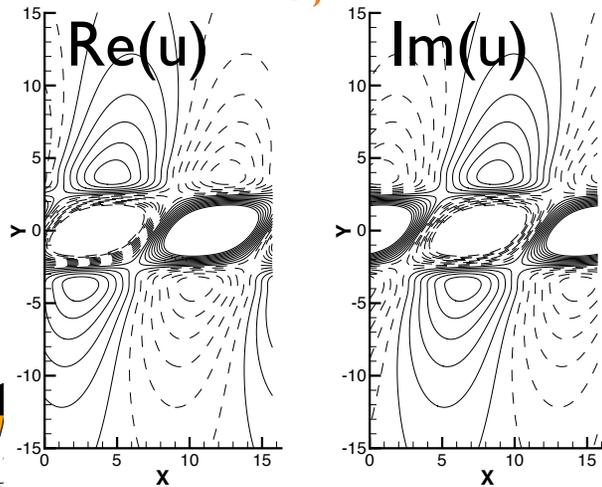
$k=1, n=1$



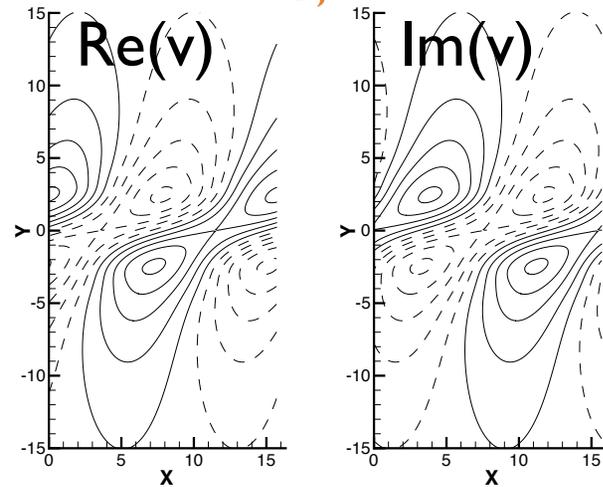
$k=1, n=1$



$k=1, n=2$



$k=1, n=2$



POD modes

- Energy contained in modes (k=2 initial condition)

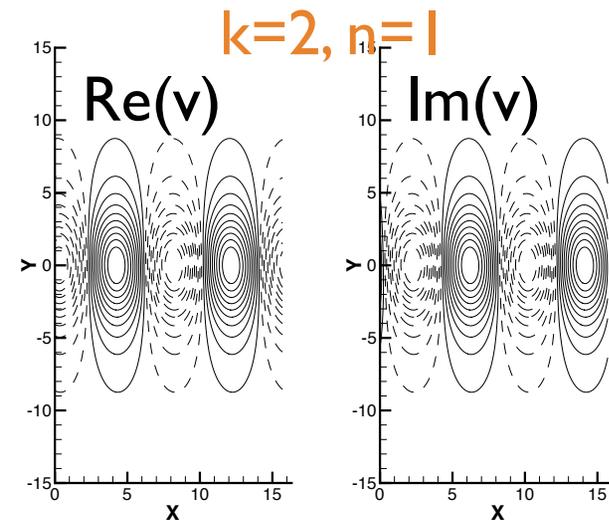
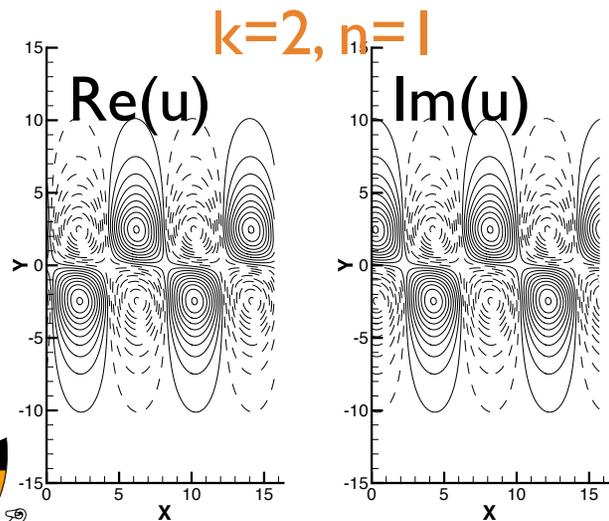
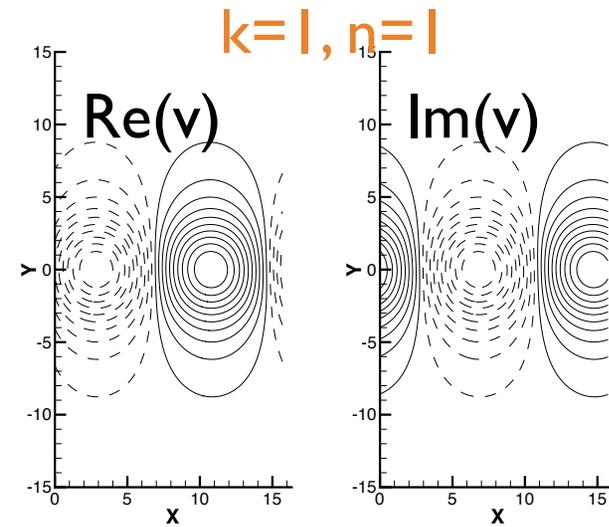
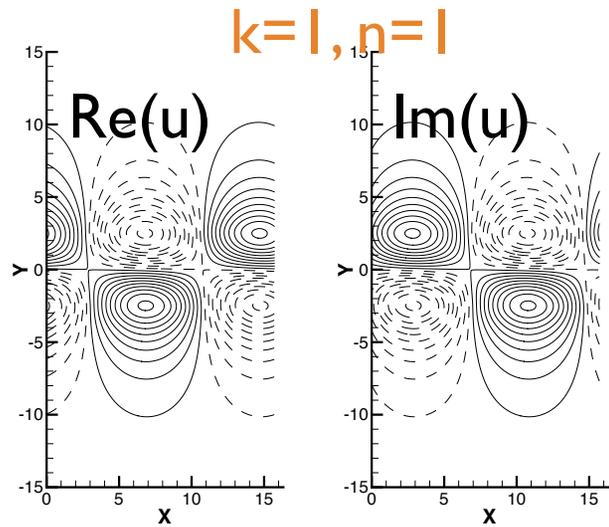
(k,n)	lambda	Energy (%)
(1,1)	27.5	40.1
(2,1)	37.9	55.2
(1,2)	0.9	1.3
(2,2)	1.6	2.3
all k=0		0.6

- Scaling still effective at removing the mean spreading (zero mode has small energy)



POD modes

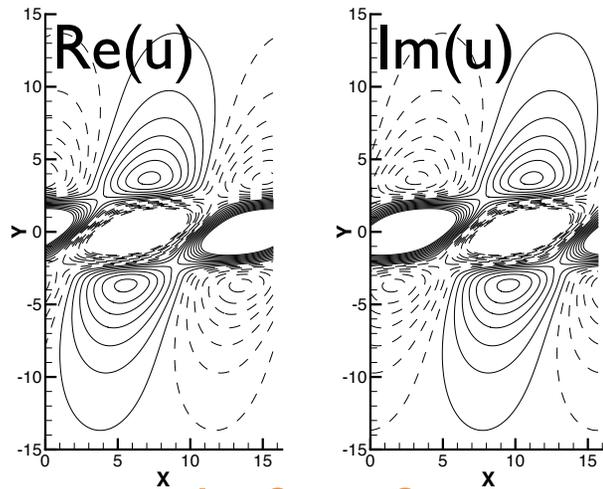
- Initial condition with $k=2$



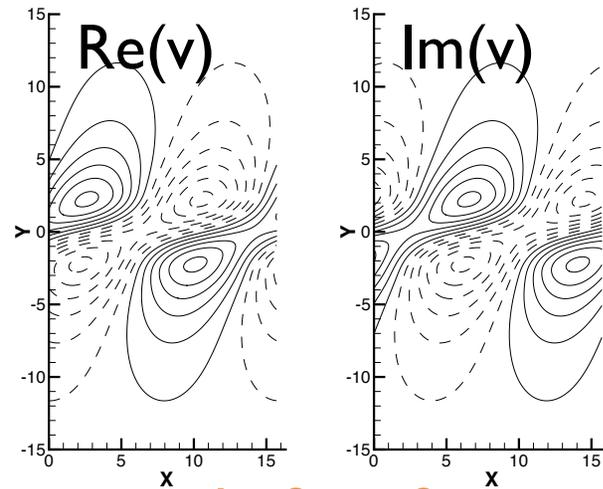
POD modes

- Initial condition with $k=2$

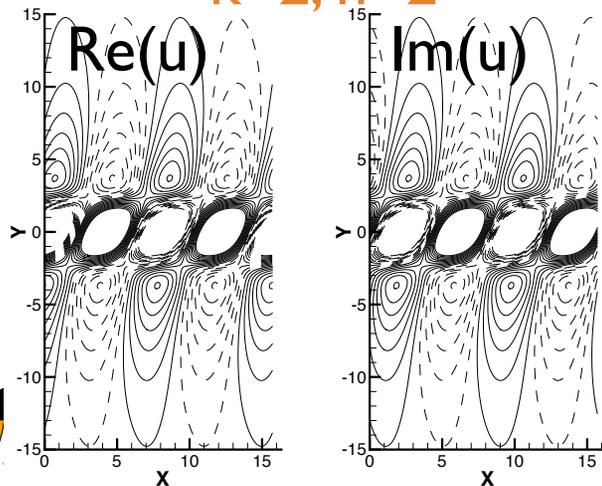
$k=1, n=2$



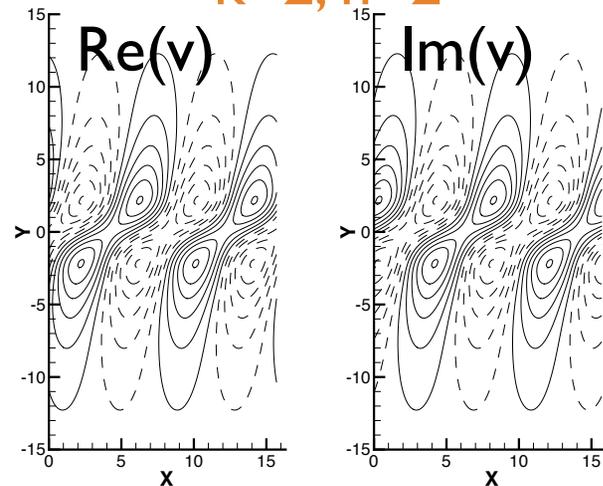
$k=1, n=2$



$k=2, n=2$

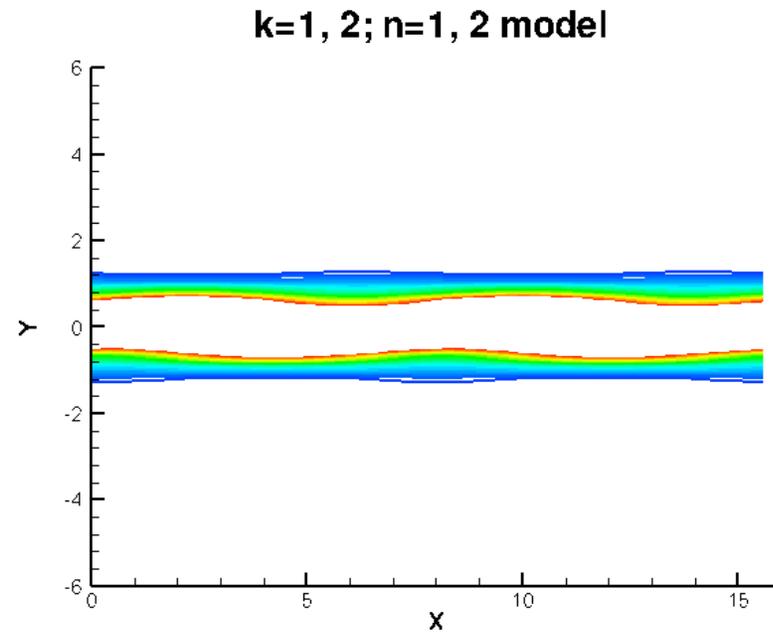
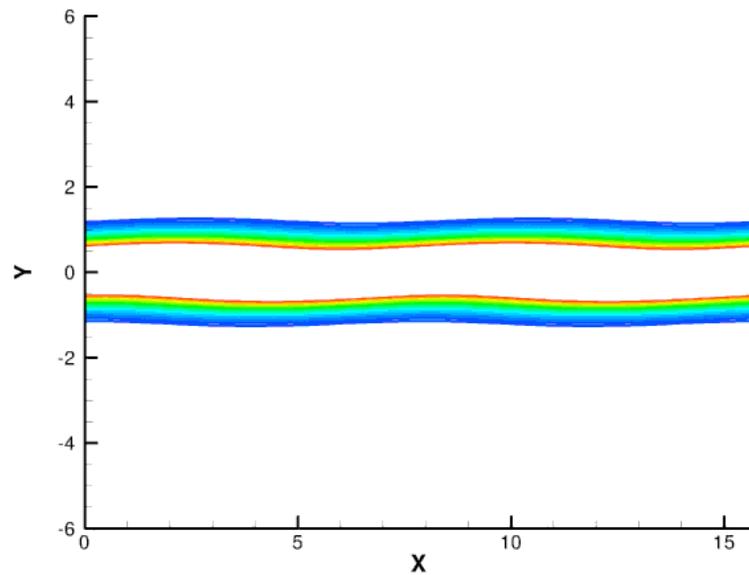


$k=2, n=2$



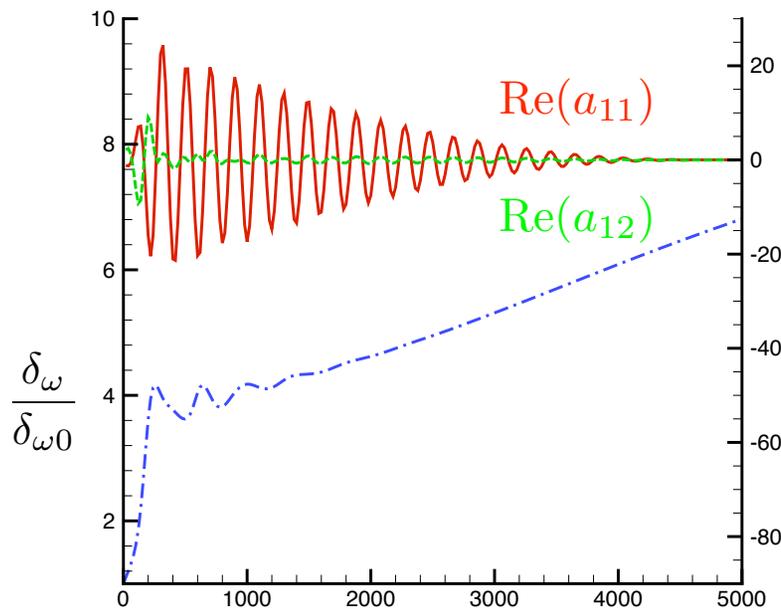
DNS v.s. Model

- Comparison of simulation and model results

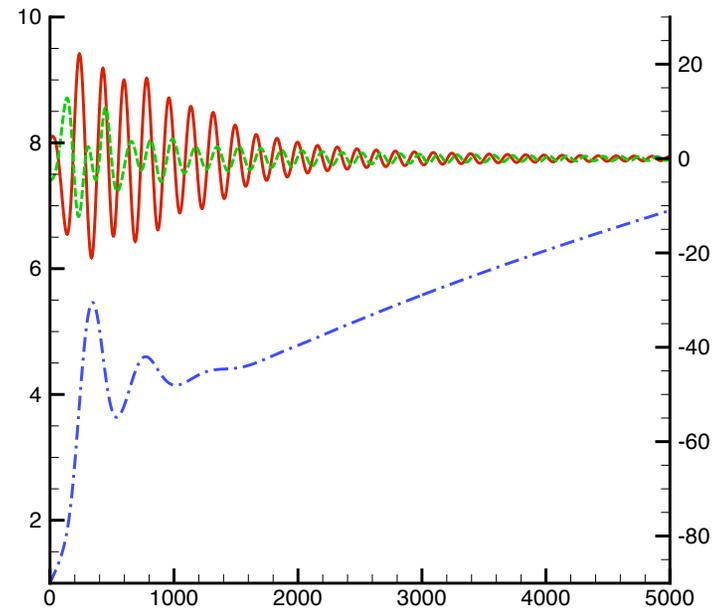


Model results: $k=1$

- Thickness and amplitude of POD modes for $k=1$ initial condition: projection of full simulation



- Thickness and amplitude of POD modes for $k=1$ initial condition: low-dimensional model

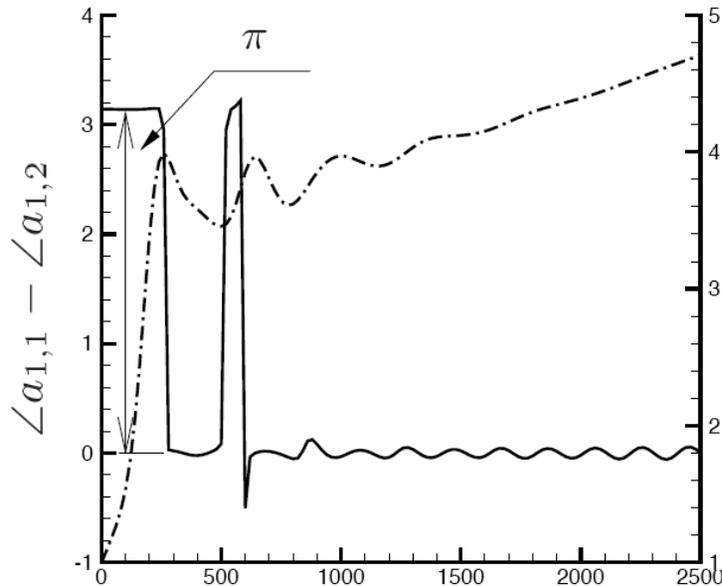


Phase shift phenomenon: Modes 1 and 2 are out of phase during linear growth, in phase after saturation

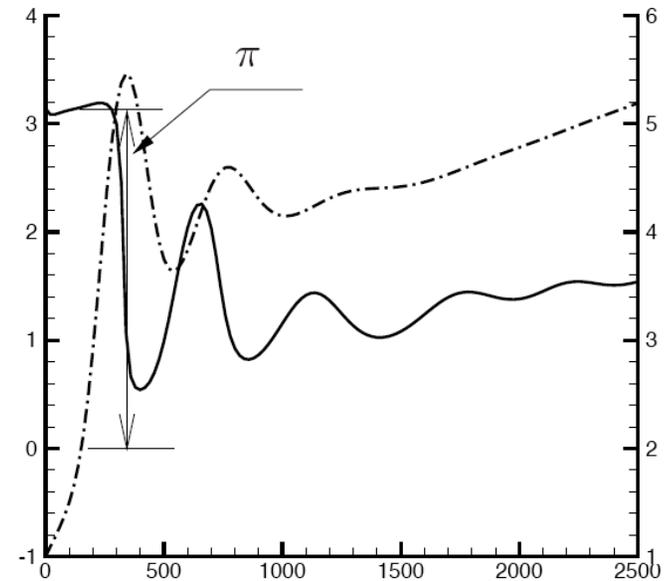


Model results: $k=1$

- Phase delay between the first 2 POD modes: **projection of full simulation**

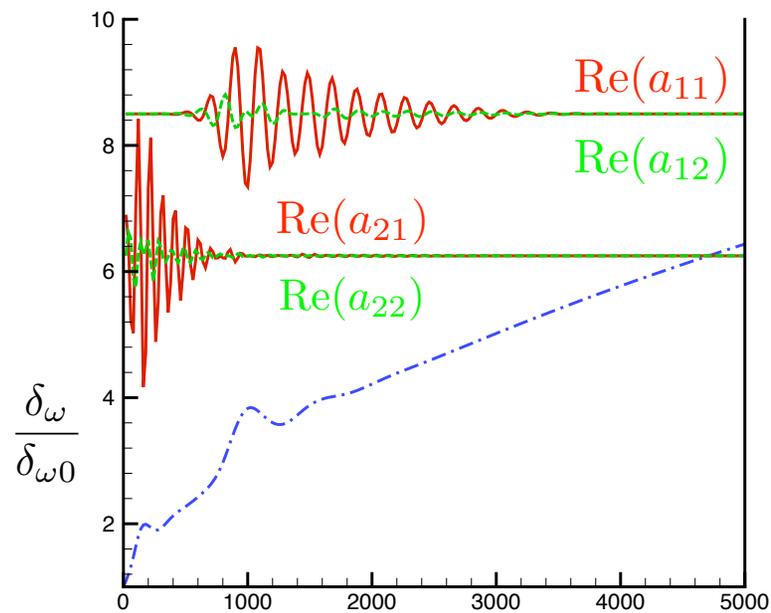


- Phase delay between the first 2 Pod modes: **low-dimensional model**

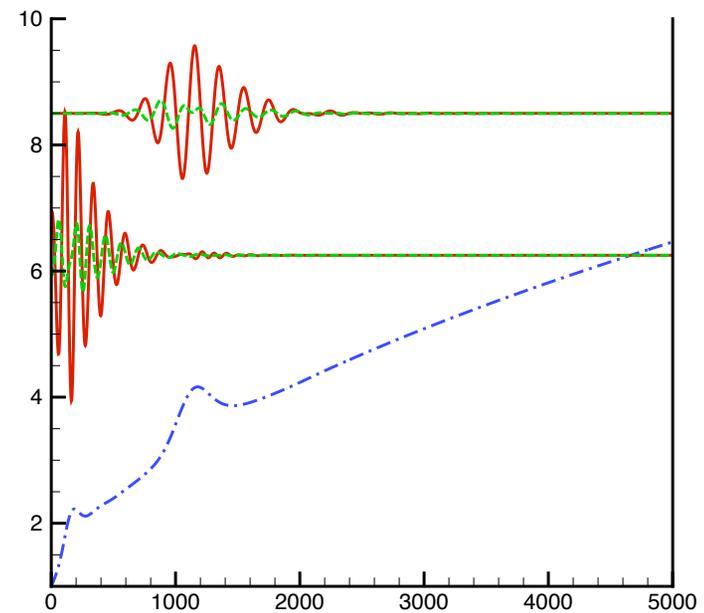


Model results: k=2

- Thickness and amplitude of POD modes for k=2 initial condition: **projection of full simulation**

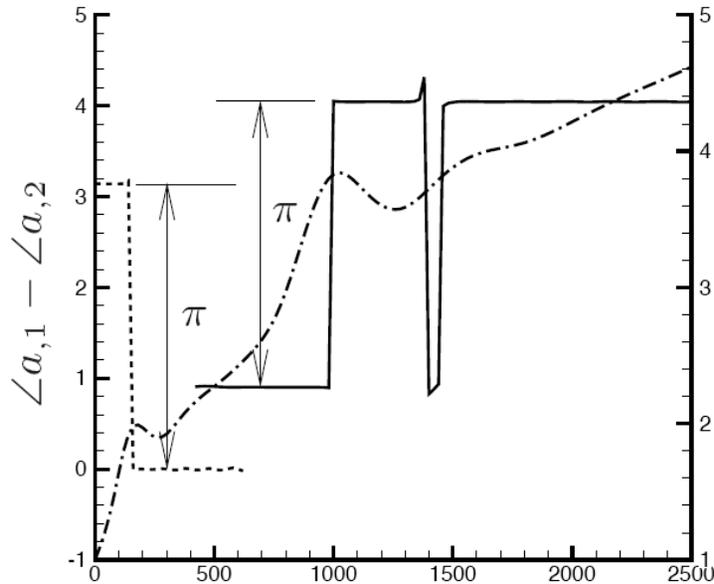


- Thickness and amplitude of POD modes for k=2 initial condition: **low-dimensional model**

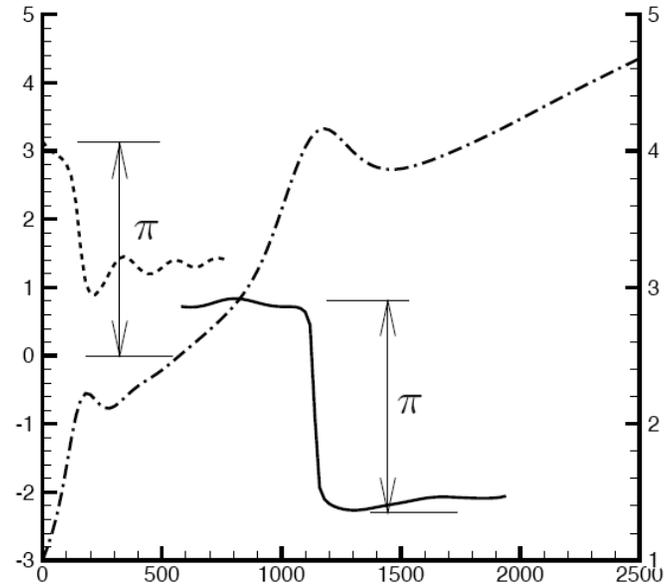


Model results: $k=2$

- Phase delay between the first 2 POD modes: projection of full simulation



- Phase delay between the first 2 Pod modes: low-dimensional model



Summary

- **Approximate balanced truncation**
 - Approximates exact balanced truncation to as high accuracy as desired, using snapshots from linearized and adjoint simulations
 - Computational cost similar to POD, once snapshots computed
 - For a given number of modes, transients and frequency response much more accurately captured than POD models of same order
 - Extension of basic approach to model unstable linear systems
 - Feedback controllers designed from these models perform well, even on full-order, nonlinear systems
 - Extensions to (weakly) nonlinear systems straightforward
- **Dynamically scaled POD modes**
 - For systems with self-similar behavior, dynamic scaling decreases number of modes required
 - Temporal shear layer dynamics modeled with 4 complex modes, including linear growth, saturation, pairing, and viscous diffusion



Outlook

- Outstanding challenges
 - Combining ideas from balanced truncation with results from experimental data, where adjoints are not available
 - Systematic approach for highly nonlinear systems (far from equilibrium)
 - Reduced-order models for messy, turbulent flows. Low-dimensional models are, strictly speaking, not possible, but one is not interested in all of the details
 - New control synthesis tools needed for these classes of nonlinear systems?

