# Contrôle optimal par réduction de modèle POD et méthode à région de confiance du sillage laminaire d'un cylindre circulaire.

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#### **Introduction** Configuration and numerical method

- Two dimensional flow around a circular cylinder at Re = 200
- Viscous, incompressible and Newtonian fluid
- Cylinder oscillation with a tangential sinusoidal velocity  $\gamma(t)$

$$\gamma(t) = \frac{V_T}{U_\infty} = \mathbf{A}\sin(2\pi \mathbf{S} t_f t)$$



 $\Gamma_{inf}$ 

Find the control parameters  $c = (A, St_f)^T$  such that the mean drag coefficient is minimized 1.  $\int_{-1}^{T} \int_{-2\pi}^{2\pi} du = 1$ 

$$\langle C_D \rangle_T = \frac{1}{T} \int_0^T \int_0^{2\pi} 2 p \, n_x \, R \, d\theta \, dt - \frac{1}{T} \int_0^T \int_0^{2\pi} \frac{2}{Re} \left( \frac{\partial u}{\partial x} \, n_x + \frac{\partial u}{\partial y} \, n_y \right) \, R \, d\theta \, dt, \quad \left( \frac{\partial u}{\partial x} \, n_x + \frac{\partial u}{\partial y} \, n_y \right) \, R \, d\theta \, dt, \quad \left( \frac{\partial u}{\partial x} \, n_x + \frac{\partial u}{\partial y} \, n_y \right) \, R \, d\theta \, dt,$$

#### Introduction Parametric study









#### **Introduction** *Mean drag coefficient* & steady unstable base flow



**Fig.** : Variation with the Reynolds number of the mean drag coefficient. Contributions and corresponding flow patterns of the base flow and unsteady flow.



Protas, B. et Wesfreid, J.E. (2002) : Drag force in the open-loop control of the cylinder wake in the laminar regime. *Phys. Fluids*, **14**(2), pp. 810-826.

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### **Reduced Order Model (ROM) and optimization problems**







- ► Introduced in turbulence by Lumley (1967).
- ► Method of information compression

► Look for a realization  $\Phi(X)$  which is closer, in an average sense, to realizations u(X). ( $X = (x, t) \in D = \Omega \times \mathbb{R}^+$ )

 $\blacktriangleright \Phi({\pmb X})$  solution of the problem :

$$\max_{\mathbf{\Phi}} \langle |(\boldsymbol{u}, \boldsymbol{\Phi})|^2 \rangle \quad \text{s.t.} \quad \|\boldsymbol{\Phi}\|^2 = 1.$$

► Snapshots method, Sirovich (1987) :

$$\int_T C(t,t')a^{(n)}(t')\,dt' = \lambda^{(n)}a^{(n)}(t).$$

- ENA
- ▶ Optimal convergence in  $L^2$  norm (energy) of  $\Phi(X)$ ⇒ Dynamical order reduction is possible
- $\Rightarrow$  Dynamical order reduction is possible.





#### **ROM** Parameter sampling in an optimization setting



![](_page_6_Picture_2.jpeg)

Discussion of parameter sampling in an optimization setting (from Gunzburger, 2004).
— path to optimizer using full system, □ initial values, ■ optimal values, and ● parameter values used for snapshot generation.

![](_page_6_Picture_4.jpeg)

#### **ROM** A simple configuration, a rich dynamical behavior

![](_page_7_Figure_1.jpeg)

![](_page_7_Picture_2.jpeg)

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► Necessity for a given reference flow to introduce new modes : either new operating conditions or shift-modes

![](_page_8_Figure_2.jpeg)

![](_page_8_Picture_3.jpeg)

![](_page_8_Picture_4.jpeg)

Fig. : Schematic representation of a dynamical transition with a non-equilibrium mode

► POD approximations consistent with our approach :

$$U(\boldsymbol{x}, t) = (u, v, p)^{T} = \underbrace{\sum_{i=0}^{N} a_{i}(t) \phi_{i}(\boldsymbol{x})}_{\text{Galerkin modes}} + \underbrace{\sum_{i=N+1}^{N+M} a_{i}(t) \phi_{i}(\boldsymbol{x})}_{\text{non-equilibrium modes}} + \underbrace{\gamma(\boldsymbol{c}, t) \boldsymbol{U}_{c}(\boldsymbol{x})}_{\text{control function}}$$

Physical aspects	Modes	Dynamical aspects
actuation mode	$oldsymbol{U}_{c}$	predetermined dynamics
mean flow mode	$oldsymbol{U}_m$ , $i=0$	$a_0 = 1$
<b>Galerkin modes</b> Dynamics of the reference flow <i>I</i>	i = 1	
	i=2	<b>POD ROM</b> Temporal dynamics of the
	i = N	modes (eventually, the
non-equilibrium modes	i = N + 1	mode $i = 0$ is solved then
Inclusion of new operating		$a_0 \equiv a_0(t)$ )
conditions $II, III, IV, \cdots$	i = N + M	

![](_page_9_Picture_4.jpeg)

► Galerkin projection of *NSE* onto the POD basis :

$$\left(\boldsymbol{\phi}_{i},\,\frac{\partial\boldsymbol{u}}{\partial t}+\boldsymbol{\nabla}\cdot\left(\boldsymbol{u}\otimes\boldsymbol{u}\right)\right)=\left(\boldsymbol{\phi}_{i},\,-\boldsymbol{\nabla}p+\frac{1}{Re}\Delta\boldsymbol{u}\right).$$

► Reduced order dynamical system where only (N + M + 1) ( $\ll N_{POD}$ ) modes are retained (state equations) :

$$\frac{d a_i(t)}{d t} = \sum_{j=0}^{N+M} \mathcal{B}_{ij} a_j(t) + \sum_{j=0}^{N+M} \sum_{k=0}^{N+M} \mathcal{C}_{ijk} a_j(t) a_k(t) + \mathcal{D}_i \frac{d \gamma}{d t} + \left( \mathcal{E}_i + \sum_{j=0}^{N+M} \mathcal{F}_{ij} a_j(t) \right) \gamma(\mathbf{c}, t) + \mathcal{G}_i \gamma^2(\mathbf{c}, t), a_i(0) = (\mathbf{U}(\mathbf{x}, 0), \phi_i(\mathbf{x})).$$

![](_page_10_Picture_5.jpeg)

 $\mathcal{B}_{ij}, \mathcal{C}_{ijk}, \mathcal{D}_i, \mathcal{E}_i, \mathcal{F}_{ij} \text{ and } \mathcal{G}_i \text{ depend on } \phi_i, U_c \text{ and } Re.$ 

![](_page_10_Picture_7.jpeg)

# Surrogate drag function and model objective function Generalities

► Drag operator :

$$\mathcal{C}_{\mathcal{D}}: \mathbb{R}^{3} \to \mathbb{R}$$
$$\boldsymbol{u} \mapsto 2 \int_{0}^{2\pi} \left( u_{3} n_{x} - \frac{1}{Re} \frac{\partial u_{1}}{\partial x} n_{x} - \frac{1}{Re} \frac{\partial u_{1}}{\partial y} n_{y} \right) R \, d\theta, \tag{1}$$

► Surrogate drag function :

$$\widetilde{C_D}(t) = \underbrace{a_0(t)N_0 + \sum_{i=N+1}^{N+M} a_i(t)N_i}_{\text{evolution of the mean drag}} + \underbrace{\sum_{i=1}^{N} a_i(t)N_i}_{\text{fluctuations } C'_D(t)} \text{ with } N_i = \mathcal{C}_{\mathcal{D}}(\phi_i).$$

► Model objective function :

$$m = \langle \widetilde{C}_D(t) \rangle_T = \frac{1}{T} \int_0^T \left( a_0(t) N_0 + \sum_{i=N+1}^{N+M} a_i(t) N_i \right) dt.$$

![](_page_11_Picture_7.jpeg)

![](_page_11_Picture_8.jpeg)

#### Surrogate drag function Test case A = 2 and St = 0.5

► Comparison of real drag coefficient  $C_D$  (symbols) and model function  $\widetilde{C_D}$  (lines) at the design parameters.

![](_page_12_Figure_2.jpeg)

![](_page_12_Picture_3.jpeg)

![](_page_12_Picture_4.jpeg)

#### **Robustness of the model objective function** *Test case* A = 2 *and* St = 0.5

![](_page_13_Figure_1.jpeg)

![](_page_13_Picture_2.jpeg)

Fig. : Comparison of the real and the model objective functions associated to the mean drag coefficient.

![](_page_13_Picture_4.jpeg)

#### Range of validity of the POD ROM restricted to a vicinity of the design parameters

Objective : Use ROMs to solve largescale optimization problems with assurance of :

- 1. Automatic restriction of the range of validity
- 2. Global convergence

![](_page_14_Figure_5.jpeg)

#### **Solution**

![](_page_14_Picture_7.jpeg)

Embed the POD technique into the concept of trust-region methods : Trust-Region Proper Orthogonal Decomposition (Fahl, 2000)

Conn, A.R., Gould, N.I.M. et Toint, P.L. (2000) : Trust-region methods. SIAM, Philadelphia.

![](_page_14_Figure_10.jpeg)

#### **Trust-Region Proper Orthogonal Decomposition (TRPOD)**Algorithm

![](_page_15_Figure_1.jpeg)

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Initial control parameters : A = 1.0 et St = 0.2

![](_page_16_Figure_2.jpeg)

Optimal control parameters : A = 4.25 et St = 0.74

Mean drag coefficient :  $\mathcal{J} = 0.993$ 

![](_page_16_Picture_6.jpeg)

![](_page_16_Picture_7.jpeg)

![](_page_17_Figure_1.jpeg)

![](_page_17_Figure_2.jpeg)

Optimal control parameters : A = 4.25 and St = 0.74

Mean drag coefficient :  $\mathcal{J} = 0.993$ 

![](_page_17_Picture_6.jpeg)

![](_page_17_Picture_7.jpeg)

![](_page_18_Figure_1.jpeg)

![](_page_18_Figure_2.jpeg)

Optimal control parameters : A = 4.25 and St = 0.74

Mean drag coefficient :  $\mathcal{J} = 0.993$ 

![](_page_18_Picture_6.jpeg)

![](_page_18_Picture_7.jpeg)

Initial control parameters : A = 6.0 et St = 1.0

![](_page_19_Figure_2.jpeg)

EMA

Mean drag coefficient :  $\mathcal{J} = 0.993$ 

![](_page_19_Picture_6.jpeg)

#### **TRPOD** Drag coefficient

► Optimal control law :  $\gamma_{opt}(t) = A \sin(2\pi St t)$  avec A = 4.25 et St = 0.74

![](_page_20_Figure_2.jpeg)

![](_page_20_Picture_3.jpeg)

![](_page_20_Picture_4.jpeg)

# **TRPOD** Vorticity contour plots

![](_page_21_Figure_1.jpeg)

Controlled flow,  $\gamma = \gamma_{opt}$ .

Fig. : Iso-values of vorticity  $\omega_z$ .

![](_page_21_Picture_4.jpeg)

Controlled flow : near wake strongly unsteady, far wake (after 5 diameters) steady and symmetric  $\rightarrow$  steady unstable base flow

![](_page_21_Picture_6.jpeg)

#### Numerical costs Discussion

- Optimal control of NSE by He *et al.* (2000) :  $\Rightarrow 30\%$  drag reduction for A = 3 and  $S_t = 0.75$ .
- Optimal control POD ROM by Bergmann et al. (2005) with no reactualization of the POD ROM :

 $\Rightarrow 25\%$  drag reduction for A = 2.2 and  $S_t = 0.53$ .

- Reduction costs compared to NSE :
  - CPU time : 100
  - Memory storage : 600

but no mathematical proof concerning the Navier-Stokes optimality.

- TRPOD (this study) :
  - $\Rightarrow$  More than 30% of drag reduction for A = 4.25 and  $S_t = 0.738$ .
  - Reduction costs compared to NSE :
    - CPU time : 4
    - Memory storage : 400

but global convergence.

![](_page_22_Picture_14.jpeg)

 $\hookrightarrow$  "Optimal" control of 3D flows becomes possible!

![](_page_22_Picture_16.jpeg)

# **Conclusions and perspectives**

- Conclusions on TRPOD
  - Important relative drag reduction : more than 30% of relative drag reduction
  - Global convergence : mathematical assurance that the solution is identical to the one of the high-fidelity model
  - TRPOD compared to NSE  $\Rightarrow$  important reduction of numerical costs :  $\hookrightarrow$  Reduction factor of the CPU time : 4
    - $\hookrightarrow$  Reduction factor of the memory storage : 400

"OPTIMAL" CONTROL OF 3D FLOWS POSSIBLE BY POD ROM

- Perspectives
  - Optimal control of the channel flow at  $Re_{\tau} = 180$
  - Test other reduced basis method than classical POD
    - Centroidal Voronoi Tessellations (Gunzburguer, 2004) : "intelligent" sampling in the control parameter space
    - Balanced POD (Rowley, 2004)
    - Model-based POD (Willcox, CDC-ECC 2005) : modify the definitions of the POD modes

![](_page_23_Picture_13.jpeg)

![](_page_23_Picture_15.jpeg)

# **Reverse von Karman flow**

Contrôle partiel (3 paramètres de contrôle)

 $\Rightarrow$  Effet propulsif, signe écoulement moyen inversé

![](_page_24_Picture_3.jpeg)

**Fig.** : Contrôle sur la partie amont :  $-120^{\circ} \le \theta \le 120^{\circ}$ 

![](_page_24_Picture_5.jpeg)

![](_page_24_Picture_6.jpeg)