Optimal rotary control of the cylinder wake using POD Reduced Order Model

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Outline

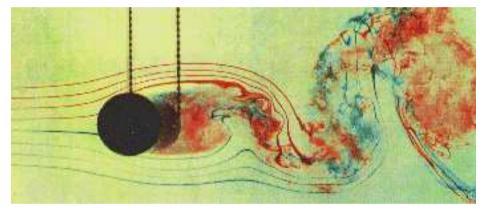
- I Flow configuration and numerical methods
- II Optimal control approach
- III Proper Orthogonal Decomposition (POD)
- IV Reduced Order Model of the cylinder wake (ROM)
- V Optimal control formulation applied to the ROM
- VI Results of POD ROM
- VII Nelder-Mead Simplex method
- Conclusions and perspectives



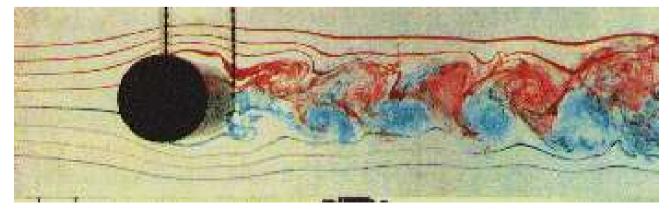


Motivations Cylinder wake flow?

- Prototype configuration of separated flow
- Experimental study of Tokumaru and Dimotakis (JFM 1991)Re=15000
 - ▶ Unforced flow



► Forced flow ⇒ 80% of drag reduction





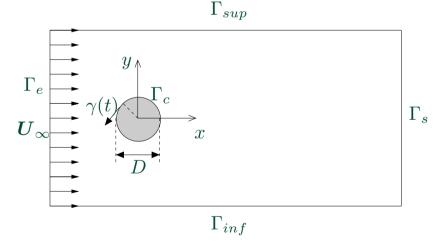


I - Configuration and numerical method

- Two dimensional flow around a circular cylinder at Re = 200
- Viscous, incompressible and Newtonian fluid

$$\nabla \cdot \boldsymbol{u} = 0$$
 ; $\frac{\partial \boldsymbol{u}}{\partial t} + (\boldsymbol{u} \cdot \nabla) \boldsymbol{u} = -\nabla P + \frac{1}{Re} \Delta \boldsymbol{u}$

• Cylinder oscillation with a tangential velocity $\gamma(t)$



Control parameter :

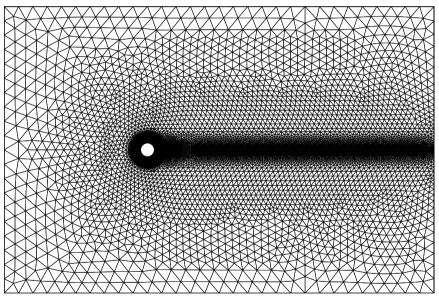


$$\alpha(t) = \frac{\gamma(t)}{U_{\infty}} = \frac{R\dot{\theta}(t)}{U_{\infty}} = \frac{\text{Tangential velocity}}{\text{Upstream velocity}}$$



I - Configuration and numerical method

- Fractional step method in time (pressure correction)
- Finite Element Method (FEM) in space (P_1, P_1)
 - Numerical domain $\Omega = \{-10 \le x \le 20 \, ; \, -10 \le y \le 10\}$; D=1
 - Mesh: 25042 triangles, 12686 vertices

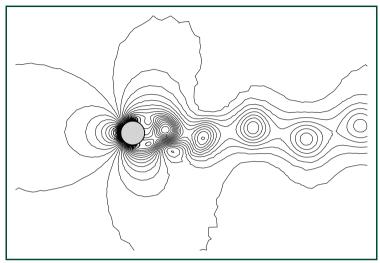




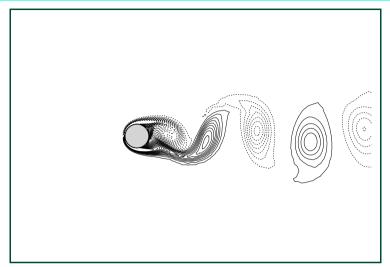
► Numerical code written by M.Braza (IMFT-EMT2) & D.Ruiz (ENSEEIHT)



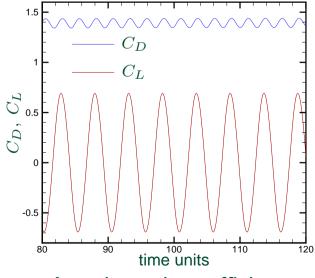
I - Configuration and numerical method



Iso pressure at t = 100.



Iso vorticity at t = 100.



Aerodynamic coefficients.

Authors	S_t	C_D
Braza <i>et al.</i> (1986)	0.2000	1.4000
Henderson et al. (1997)	0.1971	1.3412
He et al. (2000)	0.1978	1.3560
this study	0.1983	1.3972

Strouhal number and drag coefficient.





II - Optimal control Definition

Mathematical method allowing to determine without a priori knowledge a control law based on the optimization of a cost functional.

- State equations $\mathcal{F}(\phi,c)=0$; (Navier-Stokes + I.C. + B.C.)
- Control variables c; (Blowing/suction, design parameters ...)
- Cost functional $\mathcal{J}(\phi,c)$. (Drag, lift, target function, ...)



Find a control law c and state variables ϕ such that the cost functional

 $\mathcal{J}(\phi,c)$ reach an extremum under the constraint $\mathcal{F}(\phi,c)=0$.

II - Optimal control Lagrange multipliers

Constrained optimization ⇒ unconstrained optimization

- ▶ Introduction of Lagrange multipliers ξ (adjoint variables).
- ► Lagrange functional :

$$\mathcal{L}(\phi, c, \xi) = \mathcal{J}(\phi, c) - \langle \mathcal{F}(\phi, c), \xi \rangle$$

▶ Force \mathcal{L} to be stationary \Rightarrow look for $\delta \mathcal{L} = 0$:

$$\delta \mathcal{L} = \frac{\partial \mathcal{L}}{\partial \phi} \delta \phi + \frac{\partial \mathcal{L}}{\partial c} \delta c + \frac{\partial \mathcal{L}}{\partial \xi} \delta \xi = 0$$

▶ Hypothesis : ϕ , c and ξ assumed to be independent of each other :

$$\frac{\partial \mathcal{L}}{\partial \phi} \delta \phi = \frac{\partial \mathcal{L}}{\partial c} \delta c = \frac{\partial \mathcal{L}}{\partial \xi} \delta \xi = 0$$



where

$$\frac{\partial \mathcal{L}}{\partial x} = \lim_{\epsilon \to 0} \frac{\mathcal{L}(x + \epsilon \delta x) - \mathcal{L}(x)}{\epsilon} = 0 \quad \forall \delta x \quad \text{(Fréchet derivative)}$$



II - Optimal control Optimality system

- ▶ State equations ($\frac{\partial \mathcal{L}}{\partial \xi} \delta \xi = 0$) : $\boxed{\mathcal{F}(\phi, c) = 0}$
- ▶ Co-state (adjoint) equations ($\frac{\partial \mathcal{L}}{\partial \phi} \delta \phi = 0$) :

$$\left[\left(\frac{\partial \mathcal{F}}{\partial \phi} \right)^* \xi = \left(\frac{\partial \mathcal{J}}{\partial \phi} \right)^* \right]$$

▶ Optimality condition ($\frac{\partial \mathcal{L}}{\partial c}\delta c = 0$):

$$\left[\left(\frac{\partial \mathcal{J}}{\partial c} \right)^* = \left(\frac{\partial \mathcal{F}}{\partial c} \right)^* \xi \right]$$

⇒ Expensive method in CPU time and storage memory for large system!



⇒ Ensure only a local (generally not global) minimum



II - Optimal control Iterative method

- $ightharpoonup c^{(0)}$ given ; for $n=0,1,2,\ldots$ and while a convergence criterium is not satisfied, do :
 - 1. From t=0 to t=T solve the state equations with $c^{(n)}$; \hookrightarrow state variables $\phi^{(n)}$
 - 2. From t=T to t=0 solve the co-state equations with $\phi^{(n)}$; \hookrightarrow co-state variables $\xi^{(n)}$
 - 3. Solve the optimality condition with $\phi^{(n)}$ and $\xi^{(n)}$; \hookrightarrow *objective gradient* $\delta c^{(n)}$
 - 4. New control law $\hookrightarrow c^{(n+1)} = c^{(n)} + \omega^{(n)} \delta c^{(n)}$



► End do.



II - Optimal control Reduced Order Model (ROM)

"without an inexpensive method for reducing the cost of flow computation, it is unlikely that the solution of optimization problems involving the three dimensional unsteady Navier-Stokes system will become routine"

M. Gunzburger, 2000 Initialization Recourse to detailed model (TRPOD) High-fidelity model f(x), grad f(x)a(x), grad a(x)Approximation model **Optimization** Optimization on simplified model

III - Proper Orthogonal Decomposition (POD)

- ▶ Introduced in fluid mechanics (turbulence context) by Lumley (1967).
- ▶ Look for a realization $\phi(X)$ which is closer, in an average sense, to the realizations u(X). $(X = (x, t) \in \mathcal{D} = \Omega \times \mathbb{R}^+)$
- $lackbox{\phi}(m{X})$ solution of the problem : $\max_{m{\phi}}\langle |(m{u},m{\phi}|^2
 angle \quad \text{s.t.} \quad \|m{\phi}\|^2=1.$
- ➤ Snapshots method, Sirovich (1987):

$$\int_T C(t, t') a^{(n)}(t') dt' = \lambda^{(n)} a^{(n)}(t).$$

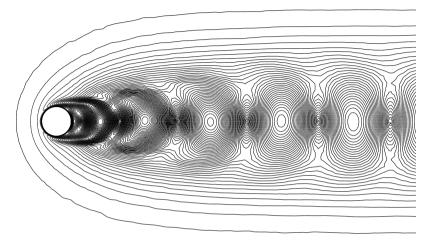
- ▶ Optimal convergence L^2 norm (energy) of $\phi(X)$ \Rightarrow Dynamical order reduction is possible.
- ► Decomposition of the velocity field :



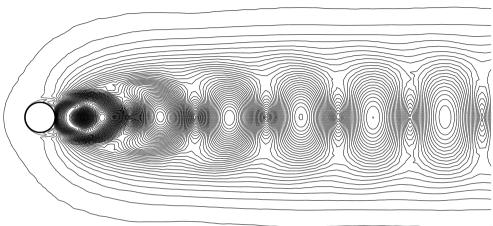
$$u(x,t) = \sum_{i=1}^{N_{POD}} a^{(i)}(t)\phi^{(i)}(x).$$



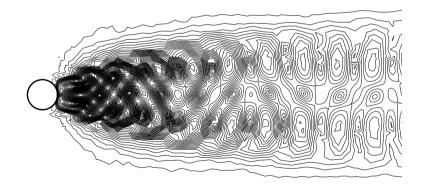
III - POD *POD* modes : uncontrolled flow ($\gamma = 0$)



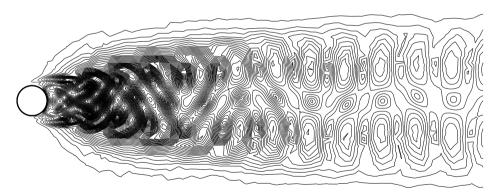
First POD mode.



Second POD mode.



Third POD mode.



Fourth POD mode.



III - Reduced Order Model of the cylinder wake (ROM)

► Galerkin projection of *NSE* on the POD basis :

$$\left(\boldsymbol{\phi}^{(i)}, \frac{\partial \boldsymbol{u}}{\partial t} + (\boldsymbol{u} \cdot \boldsymbol{\nabla})\boldsymbol{u}\right) = \left(\boldsymbol{\phi}^{(i)}, -\boldsymbol{\nabla}p + \frac{1}{Re}\Delta\boldsymbol{u}\right).$$

► Integration by parts (Green's formula) leads :

$$\left(\boldsymbol{\phi}^{(i)}, \frac{\partial \boldsymbol{u}}{\partial t} + (\boldsymbol{u} \cdot \boldsymbol{\nabla})\boldsymbol{u}\right) = \left(p, \boldsymbol{\nabla} \cdot \boldsymbol{\phi}^{(i)}\right) - \frac{1}{Re} \left((\boldsymbol{\nabla} \otimes \boldsymbol{\phi}^{(i)})^T, \boldsymbol{\nabla} \otimes \boldsymbol{u}\right) - \left[p \, \boldsymbol{\phi}^{(i)}\right] + \frac{1}{Re} \left[(\boldsymbol{\nabla} \otimes \boldsymbol{u}) \boldsymbol{\phi}^{(i)}\right].$$



with
$$[a] = \int_{\Gamma} a \cdot n \, d\Gamma$$
 and $(A, B) = \int_{\Omega} A : B \, d\Omega = \sum_{i,j} \int_{\Omega} A_{ij} B_{ji} \, d\Omega$.



III - Reduced Order Model of the cylinder wake (ROM)

 \blacktriangleright Velocity decomposition with N_{POD} modes :

$$\boldsymbol{u}(\boldsymbol{x},t) = \boldsymbol{u}_m(\boldsymbol{x}) + \gamma(t) \, \boldsymbol{u}_c(\boldsymbol{x}) + \sum_{k=1}^{N_{POD}} a^{(k)}(t) \boldsymbol{\phi}^{(k)}(\boldsymbol{x}).$$

▶ Reduced order dynamical system where only N_{gal} ($\ll N_{POD}$) modes are retained (state equations):

$$\begin{split} \frac{d\,a^{(i)}(t)}{d\,t} = & \mathcal{A}_i + \sum_{j=1}^{N_{gal}} \mathcal{B}_{ij}\,a^{(j)}(t) + \sum_{j=1}^{N_{gal}} \sum_{k=1}^{N_{gal}} \mathcal{C}_{ijk}\,a^{(j)}(t)a^{(k)}(t) \\ & + \mathcal{D}_i\,\frac{d\,\gamma}{d\,t} + \left(\mathcal{E}_i + \sum_{j=1}^{N_{gal}} \mathcal{F}_{ij}\,a^{(j)}(t)\right)\gamma + \mathcal{G}_i\gamma^2 \\ a^{(i)}(0) = (\boldsymbol{u}(\boldsymbol{x},\,0),\,\boldsymbol{\phi}^{(i)}(\boldsymbol{x})). \end{split}$$

$$\mathcal{A}_i,\,\mathcal{B}_{ij},\,\mathcal{C}_{ijk},\,\mathcal{D}_i,\,\mathcal{E}_i,\,\mathcal{F}_{ij} \text{ and } \mathcal{G}_i \text{ depend on } \boldsymbol{\phi},\,\boldsymbol{u}_m,\,\boldsymbol{u}_c \text{ and } Re. \end{split}$$

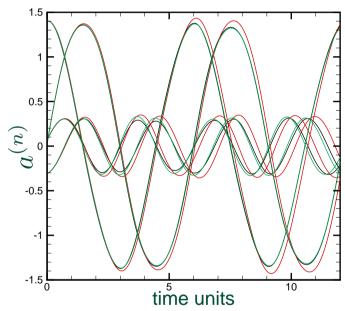




IV - Reduced Order Model of the cylinder wake Stabilization

Integration and "optimal" stabilization of the POD ROM for $\gamma = A \sin(2\pi S_t t)$, A=2 and $S_t=0.5$.

POD reconstruction errors ⇒ temporal modes amplification



Temporal evolution of the first 6 POD temporal modes.

► Reasons:

- Extraction by POD only of the large energetic eddies
- Dissipation takes place in small eddies

► Solution :

Addition of an optimal artificial viscosity on each POD mode



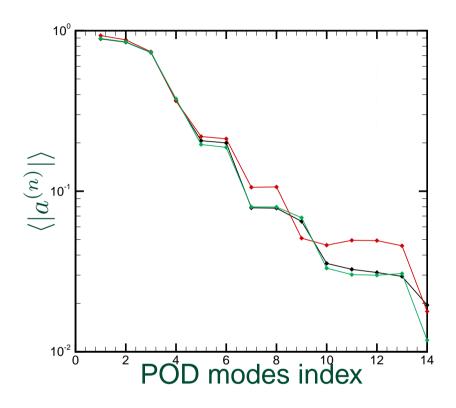
projection (Navier-Stokes)

prediction before stabilization (POD ROM)

prediction after stabilization (POD ROM).



IV - Reduced Order Model of the cylinder wake Stabilization



0.006 (u) b 0.004 0.003 (u) b 0.002 0.001 POD modes index

Comparison of energetic spectrum.

Comparison of absolute errors.

- ► Good agreements between POD ROM spectrum and DNS spectrum
- ► Reduction of the reconstruction error between predicted (POD ROM) and projected (DNS) modes





V - Optimal control formulation based on ROM

► Objective functional:

$$\mathcal{J}(\boldsymbol{a}, \gamma(t)) = \int_0^T J(\boldsymbol{a}, \gamma(t)) dt = \int_0^T \left(\sum_{i=1}^{N_{gal}} a^{(i)^2} + \frac{\alpha}{2} \gamma(t)^2 \right) dt.$$

 α : regularization parameter (penalization).

► Co-state equations :

$$\frac{d\xi^{(i)}(t)}{dt} = -\sum_{j=1}^{N_{gal}} \left(\mathcal{B}_{ji} + \gamma \mathcal{F}_{ji} + \sum_{k=1}^{N_{gal}} \left(\mathcal{C}_{jik} + \mathcal{C}_{jki} \right) a^{(k)} \right) \xi^{(j)}(t) - 2a^{(i)}$$
$$\xi^{(i)}(T) = 0.$$



Optimality condition (gradient):

$$\delta\gamma(t) = -\sum_{i=1}^{N_{gal}} \mathcal{D}_i \frac{d\xi^{(i)}}{dt} + \sum_{i=1}^{N_{gal}} \left(\mathcal{E}_i + \sum_{j=1}^{N_{gal}} \mathcal{F}_{ij} a^{(j)} + 2\mathcal{G}_i \gamma(t) \right) \xi^{(i)} + \alpha \gamma$$

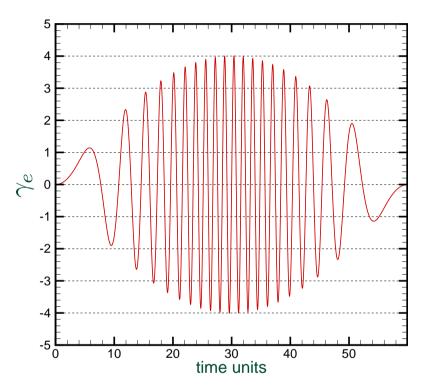
VI - Results of POD ROM Generalities

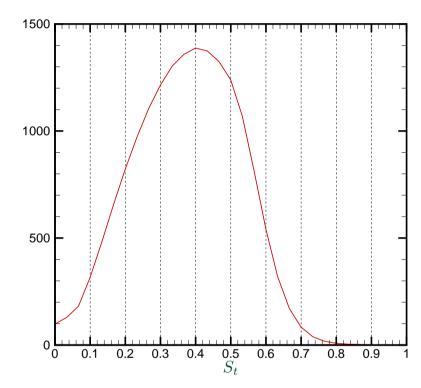
- ▶ No reactualization of the POD basis.
- ► The energetic representativity is *a priori* different to the dynamical one :
- → possible inconvenient for control,
- → a POD dynamical system represents a priori only the dynamics (and its vicinity) used to build the low dynamical model.
- ► Construction of a POD basis representative of a large range of dynamics :
- \hookrightarrow excitation of a great number of degrees of freedom scanning $\gamma(t)$ in amplitudes and frequencies.





VI - Results of POD ROM Excitation





$$\gamma_e(t) = A_1 \sin(2\pi S_{t1} t) \times \sin(2\pi S_{t2} t - A_2 \sin(2\pi S_{t3} t))$$

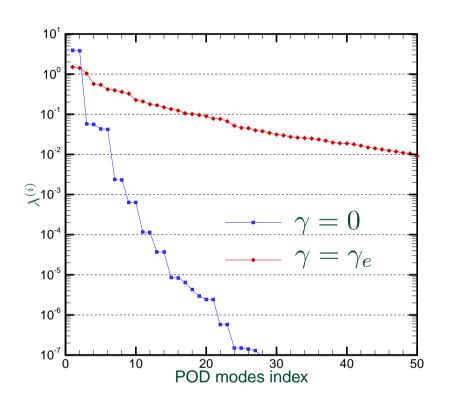
with
$$A_1 = 4$$
, $A_2 = 18$, $S_{t1} = 1/120$, $S_{t2} = 1/3$ and $S_{t3} = 1/60$.

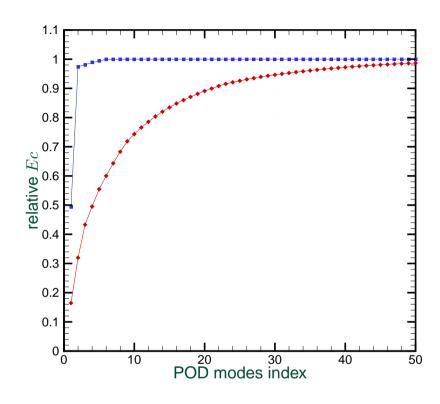


- ▶ $0 \le \text{amplitudes} \le 4 \text{ and Fourier analysis} \Rightarrow 0 \le \text{frequencies} \le 0.65$
- $ightharpoonup \gamma_e$ initial control law in the iterative process.



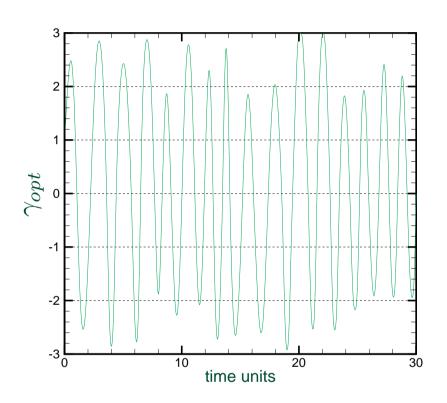
VI - Results of POD ROM Energy

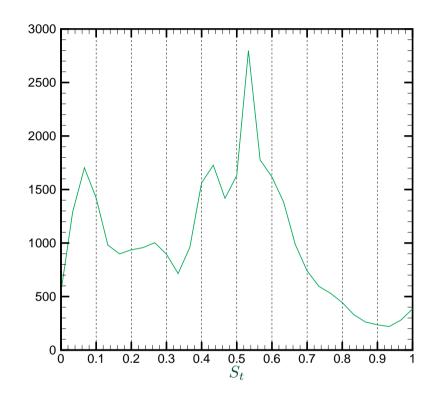




- ▶ Stationary cylinder $\gamma = 0 : \hookrightarrow 2$ modes out of 100 are sufficient to restore 97% of the kinetic energy.
- ▶ Controlled cylinder $\gamma = \gamma_e : \hookrightarrow 40$ modes out of 100 are then necessary to restore 97% of the kinetic energy
- ⇒ Improvement of the POD ROM robustness to dynamical evolution

VI - Results of POD ROM Optimal control





▶ Reduction of the wake instationarity. $\gamma_{opt} \simeq A \sin(2\pi S_t t)$ with A=2.2 and $S_t=0.53$

$$\mathcal{J}(\gamma_e) = 9.81 \implies \mathcal{J}(\gamma_{opt}) = 5.63.$$

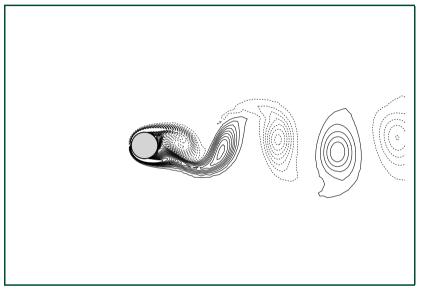


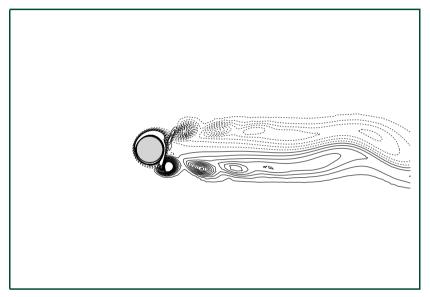
- ▶ The control is optimal for the reduced order model based on POD.
- ▶ Is it also optimal for the Navier-Stokes model?



VI - Results of POD ROM Comparison of wakes' organization

▶ No mathematical proof concerning the Navier Stokes optimality.





no control $\gamma = 0$

optimal control $\gamma = \gamma_{opt}$

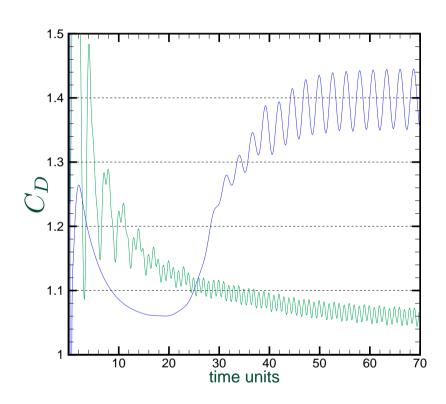
Isocontours of vorticity ω_z .

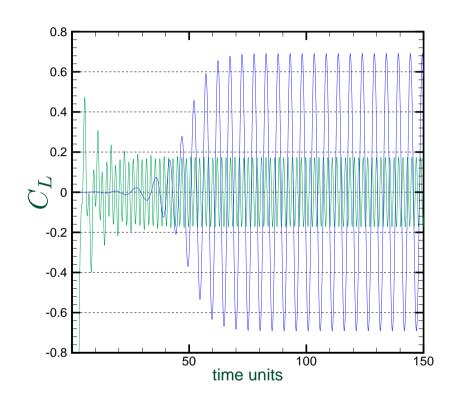
- ightharpoonup no control : $\gamma = 0 \Rightarrow$ Asymmetric flow.
 - \hookrightarrow Large and energetic eddies.
- ▶ optimal control : $\gamma = \gamma_{opt} \Rightarrow$ Symmetrization of the (near) wake.
 - \hookrightarrow Smaller and lower energetic eddies.





VI - Results of POD ROM Aerodynamic coefficients





► Important drag reduction :

$$C_{D0}=1.40$$
 for $\gamma=0$ and $C_D=1.04$ for $\gamma=\gamma_{opt}$ $C_D/C_{D0}=0.74\Rightarrow$ more than 25%.



▶ Decrease of the lift amplitude :

$$C_L=0.68$$
 for $\gamma=0$ and $C_L=0.13$ for $\gamma=\gamma_{opt}$.



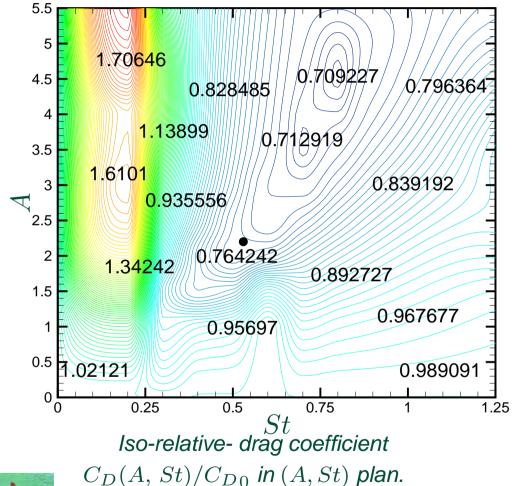
VI - Results of POD ROM Numerical costs

- ▶ Optimal control of NSE by He *et al.* (2000) : \hookrightarrow harmonic control law with A=3 and $S_t=0.75$. $\Rightarrow 30\%$ drag reduction.
- ▶ Optimal control POD ROM (this study) : \hookrightarrow harmonic control law with A=2.2 and $S_t=0.53$. $\Rightarrow 25\%$ drag reduction.
- Reduction costs using POD ROM compared to NSE :
 - CPU time : 100
 - Memory storage : 600
 - → "Optimal" control of 3D flows becomes possible!



▶ Does the POD ROM control law correspond to the global minimum?

VI - Results of POD ROM Numerical optimization



Observations

- ► Minimum is located in a smooth valley
- \hookrightarrow Global minimum : around A=4.4 and St=0.76
- ► Maximum is located in a sharp peak
- \hookrightarrow Global maximum : near St=0.2, the natural frequency : lock-on flow



Finding the global minimum with an optimization algorithm may be difficult due to the smooth valley



VI - Results of POD ROM Local versus global minimum

- ► POD ROM control law does not correspond to the global minimum
 - \hookrightarrow POD ROM parameters : A=2.2 and $St=0.53 \Rightarrow C_D=1.04$
 - \hookrightarrow Global minimum parameters : A=4.4 and $St=0.76\Rightarrow C_D=0.98$
- ightharpoonup Results in (A, St) quite different but not so far in terms of C_D
- ► Improvement : coupling to the POD ROM approach an efficient new optimization algorithm for smooth fonctions





VII - Nelder-Mead Simplex method Generalities

Advantages

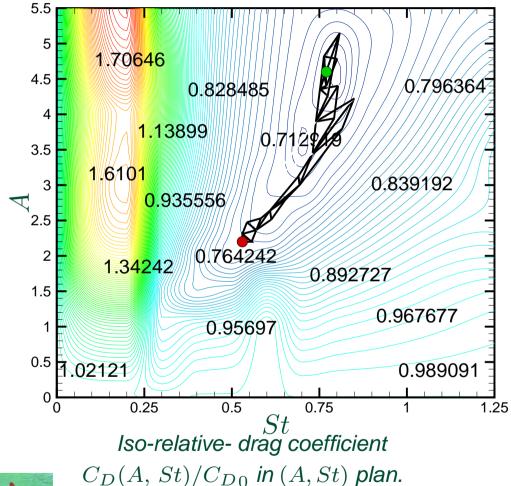
- ▶ Numerical simplicities
- ► Adaptive topology
- ► Free gradient optimization method
- ► Good results with smooth functions

Drawbacks

- ▶ No proof of optimality for simplex dimensions greater than two
- ▶ Need to fix free parameters
- ► Maybe more iterations than gradient based optimisation algorithms



VII - Nelder-Mead Simplex method Results



Observations

- ► Topology adaptation function of the curve of the valley
- ► Minimum found by the simplex method :

$$A = 4.5$$
 and $St = 0.76$

- Seems to be the global minimum
- ▶ 30 NSE resolutions $\Rightarrow 5\%$ additive drag reduction compared to POD ROM



Relative drag reduction by POD ROM : 25% (1 NSE resolution) Usefulness of coupling a new algorithm?



Conclusions

- Important drag reduction obtained by POD ROM : more than 25% of relative drag reduction
- This solution is not the global minimum of the drag function
- POD ROM compared to NSE ⇒ important reduction of numerical costs :
 - → Reduction factor of the CPU time: 100

"OPTIMAL" CONTROL OF 3D FLOWS POSSIBLE BY POD ROM

- Coupling POD ROM with the Nelder-Mead simplex method leads a priori to the global minimum of the drag function
- But the gain on the drag function is quite small (5%) compared to results obtained by POD ROM



Perspectives

- Improve the representativity of the POD ROM
 - \hookrightarrow "Optimize" the temporal excitation γ_e
- Look for harmonic control $\gamma(t) = A \sin(2\pi S_t t)$ with POD basis reactualization (closed loop on NSE and not only on POD ROM)
- Coupling the POD ROM approach with Trust Region POD method (TRPOD)
 - ⇒ proof of convergence under weak conditions
- Introducing the pressure into the POD dynamical system
 - \hookrightarrow pressure contribution to drag coefficient : 80%
- Optimal control of the Navier-Stokes equations



