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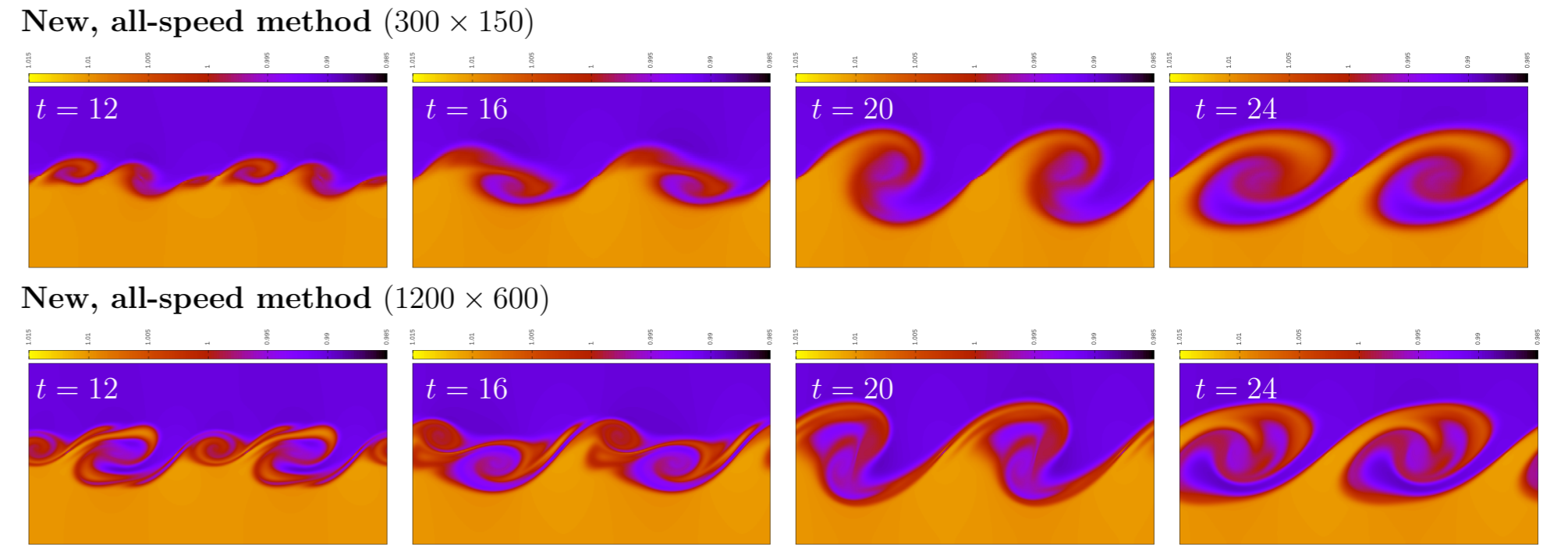
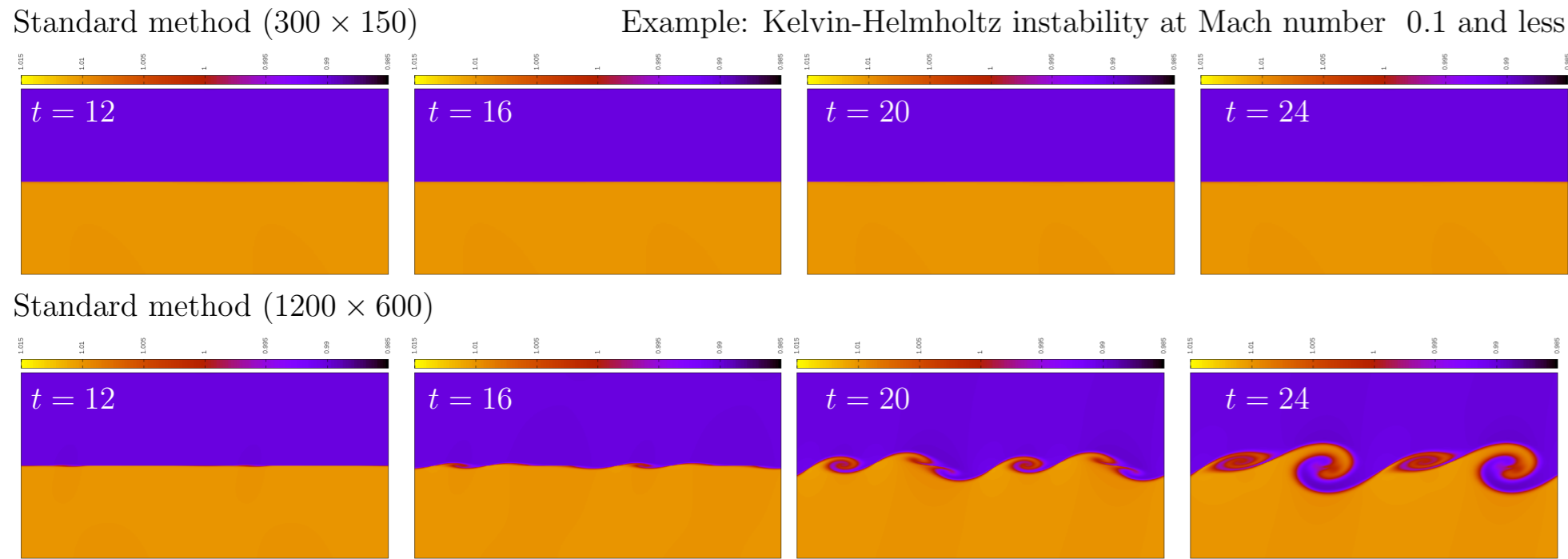


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TL;DR: At low Mach number, the solutions approach the incompressible regime with $\nabla \cdot \mathbf{v} = \partial_x u + \partial_y v \in \mathcal{O}(\epsilon)$. The aim is to achieve an all-speed (high + low Mach numbers) Finite Volume method. The problematic term at low Mach number is $\frac{1}{\epsilon} \partial_x^2 u$ in the numerical diffusion, but just removing it does not work for high Mach number flow. Here, a truly multi-dimensional way without ad-hoc fixes is presented: to complement the term to $\frac{1}{\epsilon} \partial_x (\partial_x u + \partial_y v) = \frac{1}{\epsilon} \partial_x (\nabla \cdot \mathbf{v})$. The stabilizing terms for high-Mach flow are still in place, but the term is automatically $\mathcal{O}(1)$ if Mach number is low; no switch required.



Simple way to think about the **spatial low Mach number problem**: A finite volume method that has an error $\sim \frac{\Delta x}{\epsilon} \rightarrow$ requires unreasonable grid refinement if ϵ is small. There is additionally a **temporal low Mach number problem** related to explicit time stepping: as $\epsilon \rightarrow 0$, due to the CFL condition $\Delta t \rightarrow 0$. A solution to this is implicit/IMEX methods. Here, let us insist on the ability to integrate explicitly in time and focus on the spatial low Mach number problem only!

Identifying the problems of standard methods:

Low Mach number limit

Take $\epsilon > 0$ and $M_{loc} = \frac{|\mathbf{v}|}{\sqrt{\gamma p/\rho}} \in \mathcal{O}(\epsilon)$ as $\epsilon \rightarrow 0$. This corresponds to solving

$$\begin{aligned} \partial_t \rho + \nabla \cdot (\rho \mathbf{v}) &= 0 \\ \partial_t (\rho \mathbf{v}) + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) + \frac{\nabla p}{\epsilon^2} &= 0 \\ \partial_t e + \nabla \cdot (\mathbf{v}(e + p)) &= 0 \end{aligned} \quad e = \frac{p}{\gamma - 1} + \frac{1}{2} \epsilon^2 \rho |\mathbf{v}|^2 \quad \mathbf{v} = (u, v)^T \text{ in 2d}$$

Formal asymptotic analysis:

$$\begin{aligned} \mathbf{v} &= \mathbf{v}^{(0)} + \mathbf{v}^{(1)} \epsilon + \mathbf{v}^{(2)} \epsilon^2 + \dots \\ p &= p^{(0)} + p^{(1)} \epsilon + p^{(2)} \epsilon^2 + \dots \end{aligned}$$

Limit equations: $\nabla p^{(0)} = \nabla p^{(1)} = 0, \nabla \cdot \mathbf{v}^{(0)} = 0$.

Asymptotic analysis

E.g. for the Roe solver, as $\epsilon \rightarrow 0$, the following numerical diffusion appears

$$\partial_t(\rho u) + \dots + \frac{\partial_x p}{\epsilon^2} \simeq \Delta x \partial_x \left(\frac{c \rho}{\epsilon} \partial_x u \right) + \Delta y \partial_y \left(\frac{c \rho}{\epsilon} \partial_y v \right) + \dots$$

As $\epsilon \rightarrow 0$, we have $\partial_x p^{(0)} = 0$ (good), and either

$$\partial_x p^{(1)} \neq 0 \quad \text{or} \quad \partial_x p^{(1)} = 0 \quad \text{and} \quad \partial_x (c^{(0)} \rho^{(0)} \partial_x u^{(0)}) + \dots = 0$$

the latter being an additional, artificial restriction on (u, v) . The discrete limit states are not discretizing **all** the limit states of the PDE: instead of all those with $\partial_x u^{(0)} + \partial_y v^{(0)} = 0$, the limit states of the scheme are (at best)

$$\partial_x u^{(0)} = 0 \quad \partial_y v^{(0)} = 0$$

The low Mach problem is **genuinely multi-dimensional**.

Directional splitting

For the conservation law $\partial_t q + \nabla \cdot \mathbf{f}(q) = 0$ (e.g. in 2d)

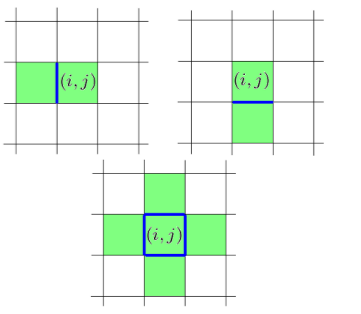
$$\partial_t q + \partial_x f^x(q) + \partial_y f^y(q) = 0$$

the **finite volume method** on Cartesian grids reads

$$\partial_t q + \frac{f_{i+\frac{1}{2},j}^x - f_{i-\frac{1}{2},j}^x}{\Delta x} + \frac{f_{i,j+\frac{1}{2}}^y - f_{i,j-\frac{1}{2}}^y}{\Delta y} = 0$$

Directional splitting: make the x-flux $f_{i+\frac{1}{2},j}^x$ only depend on $(q_{ij}, q_{i+1,j})$ and $f_{i,j+\frac{1}{2}}^y$ only on $(q_{ij}, q_{i,j+1})$. (This is a simplification of life!)

Directional splitting allows to immediately extend a 1-dimensional method to multi-d, but generally not in a way that takes into account subtle balances between different directions.



Solution (usual path):

- Remove the $\frac{1}{\epsilon}$ -scaling from the diffusion by multiplying those terms with $f \in \mathcal{O}(\epsilon)$ (e.g. $f = \min(\epsilon v/c, 1)$) (**low Mach fix**)

$$\partial_x \left(\frac{c \rho}{\epsilon} \partial_x u \right) \mapsto \partial_x \left(\frac{c \rho}{\epsilon} \partial_x u \right) \text{ with } f \in \mathcal{O}(\epsilon)$$

Li and Gu [2008], Thornber and Drikakis [2008], Dellacherie [2010], Rieper [2011], Li and Gu [2013], Chalons et al. [2016], Öbwald et al. [2016], Barsukow et al. [2017],...

- Remove all the diffusion in the acoustic operator \rightarrow central derivatives (and retain it only for advection) (**implicit/IMEX methods**) Degond et al. [2007], Degond and Tang [2011], Cordier et al. [2012], Haack et al. [2012], Bispen et al. [2017], Boscheri et al. [2020],...

Low Mach fixes come at the disadvantage of being **ad hoc**, and **reducing** stability (Birken and Meister [2005], Barsukow et al. [2017]). Implicit methods can be **expensive** for an **all-speed** regime. The actual question really is: What's wrong with Riemann solvers?

A little bit of background:

Acoustics and vorticity preservation

$$\begin{aligned} \partial_t \mathbf{v} + \nabla p &= 0 & \mathbf{v} : \mathbb{R}_0^+ \times \mathbb{R}^d \rightarrow \mathbb{R}^d \\ \partial_t p + \nabla \cdot \mathbf{v} &= 0 & p : \mathbb{R}_0^+ \times \mathbb{R}^d \rightarrow \mathbb{R} \end{aligned}$$

Vorticity $\nabla \times \mathbf{v}$ is stationary: $\partial_t (\nabla \times \mathbf{v}) = 0$. Write $\mathbf{v} = (u, v)$ in 2d.

Modified equation of a **standard scheme** that is inspired by 1d-arguments:

$$\begin{aligned} \partial_t u + \partial_x p &= \frac{1}{2} \Delta x \partial_x^2 u + \mathcal{O}(\Delta x^2) \\ \partial_t v + \partial_y p &= \frac{1}{2} \Delta x \partial_y^2 v + \mathcal{O}(\Delta x^2) \\ \partial_t p + (\partial_x u + \partial_y v) &= \frac{1}{2} \Delta x (\partial_x^2 p + \partial_y^2 p) + \mathcal{O}(\Delta x^2) \end{aligned}$$

Vorticity is no longer stationary. A **vorticity preserving method** requires the numerical diffusion to be a gradient: Morton and Roe [2001], Sidilkover [2002], Jeltsch and Torrilhon [2006], Mishra and Tadmor [2009], Lung and Roe [2014], Barsukow [2019]

$$\begin{aligned} \partial_t u + \partial_x p &= \frac{1}{2} \Delta x \partial_x (\partial_x u + \partial_y v) + \mathcal{O}(\Delta x^2) \\ \partial_t v + \partial_y p &= \frac{1}{2} \Delta x \partial_y (\partial_x u + \partial_y v) + \mathcal{O}(\Delta x^2) \\ \partial_t p + (\partial_x u + \partial_y v) &= \frac{1}{2} \Delta x (\partial_x^2 p + \partial_y^2 p) + \mathcal{O}(\Delta x^2) \end{aligned}$$

The discrete version of $\partial_x u + \partial_y v = 0 \Rightarrow \partial_{xx} u + \partial_{yy} v = 0$ is not true for all discretizations of $\partial_x, \partial_y, \partial_{xx}, \partial_{yy}$, because $D_{xx} \neq D_x^2$ in general. However, the following finite differences have this property (2d Cartesian grids):

$$\begin{array}{ccc|ccc} -1 & 1 & 1 & 2 & 1 & 1 & -2 & 1 & -1 & 1 \\ -2 & 2 & 1 & -2 & 1 & 1 & 2 & -4 & 2 & 1 \\ -1 & 1 & 1 & -1 & -2 & -1 & 1 & -2 & 1 & -1 \end{array} \quad \begin{array}{ccc} \partial_x u + \partial_y v & & \\ \partial_{xx} u + \partial_{yy} v & & \end{array}$$

Summary and outlook:

- Multi-dimensional extension ensures that the numerical diffusion depends on derivatives of the **divergence** in the limit of low Mach number
- Leaves the 1d method untouched: conceptually pleasing and surely a reason for the **good stability** properties
- Surprisingly **easy to achieve** all-speed property on Cartesian grids; strategy has been applied to a relaxation solver and a Lagrange-Projection scheme [Barsukow \[2021\]](#)
- No ad-hoc parameters**
- Automatically **all-speed**, no switches

Future work:

- Extension to high-order and unstructured grids (see [Barsukow et al. \[2023\]](#))
- Showing an entropy inequality

Solution (new approach):

Change the diffusive terms to $\partial_x \left(\frac{c \rho}{\epsilon} \partial_x (u + \partial_y v) \right) + \partial_y \left(\frac{c \rho}{\epsilon} \partial_y (\partial_x u + v) \right)$ thus making appear the divergence $\partial_x u + \partial_y v$ each time.

This is a **low-dissipation scheme** for (asymptotically) divergence-free flows. The procedure is the opposite of low Mach fixes and central discretizations: They **remove** troublesome terms, we **complement** them to make the result disappear in the limit.

The only "magic" is how to do this discretely.

From vorticity preserving methods (see on the bottom left) for linear acoustics, we know how to discretize divergences $\partial_x u + \partial_y v$ and gradients of the divergence (e.g. $\partial_{xx} u + \partial_{yy} v$). The only part required for nonlinear problems is a discretization of $\partial_x (A(\partial_x u + \partial_y v))$, where A is any

function of the dependent variables. Observe that we can rewrite $\partial_x (A(\partial_x u + \partial_y v)) = A(\partial_{xx} u + \partial_{yy} v) + (\partial_x A)(\partial_x u + \partial_y v)$.

Here is a discrete counterpart of this Leibniz rule (bracket-notation explained below):

$$\left[A_{i,j} \left(\frac{1}{4} \{ \{ u \} \}_{j \pm \frac{1}{2}} + \frac{1}{4} \{ \{ v \} \}_{i \pm \frac{1}{2}} \right) \right]_{i \pm \frac{1}{2}} = \frac{1}{2} \{ A \}_{i \pm \frac{1}{2}, j} \left(\left[\frac{1}{4} \{ \{ u \} \}_{j \pm \frac{1}{2}} \right]_{i \pm \frac{1}{2}} + \frac{1}{4} \{ \{ v \} \}_{i \pm \frac{1}{2}, j} \right) + \frac{1}{2} \{ A \}_{i \pm \frac{1}{2}, j} \left(\frac{1}{4} \{ \{ u \} \}_{j \pm \frac{1}{2}} \right)_{i \pm 1} + \left[\frac{1}{4} \{ \{ v \} \}_{i \pm \frac{1}{2}} \right]_{j \pm 1}$$

Example:

Relaxation solver

Following Bouchut [2004], Chalons et al. [2010]:

$$Q \left(\frac{x}{t}; q_i, q_{i+1} \right) := \begin{cases} q_i & \frac{x}{t} \leq u_{i+\frac{1}{2}}^* - c_{i+\frac{1}{2}}^* \\ q_{i+\frac{1}{2},L}^* & u_{i+\frac{1}{2}}^* - c_{i+\frac{1}{2}}^* \leq \frac{x}{t} < u_{i+\frac{1}{2}}^* \\ q_{i+\frac{1}{2},R}^* & u_{i+\frac{1}{2}}^* \leq \frac{x}{t} < u_{i+\frac{1}{2}}^* + c_{i+\frac{1}{2}}^* \\ q_{i+1} & \frac{x}{t} \geq u_{i+\frac{1}{2}}^* + c_{i+\frac{1}{2}}^* \end{cases} \quad \text{with the intermediate states} \quad \begin{aligned} u_{i+\frac{1}{2}}^* &= \frac{\{u\}_{i+\frac{1}{2}}}{2} - \frac{1}{2a\epsilon} [p]_{i+\frac{1}{2}} \\ p_{i+\frac{1}{2}}^* &= \frac{\{p\}_{i+\frac{1}{2}}}{2} - \frac{a\epsilon}{2} [u]_{i+\frac{1}{2}} \quad \text{etc.} \end{aligned}$$

and thus $\frac{\partial_x p^*}{\epsilon^2} \simeq \frac{\partial_x p}{\epsilon^2} - \partial_x \left(\frac{a}{\epsilon} \partial_x (u + \partial_y v) \right)$. To this end, define

$$\begin{aligned} u_{i+\frac{1}{2},j}^* &= \frac{\{ \{ u \}_{i+\frac{1}{2}} \}_{j \pm \frac{1}{2}}}{8} - \frac{1}{2a\epsilon} \frac{\{ \{ p \}_{i+\frac{1}{2}} \}_{j \pm \frac{1}{2}}}{4} \\ p_{i+\frac{1}{2},j}^* &= \frac{\{ \{ p \}_{i+\frac{1}{2}} \}_{j \pm \frac{1}{2}}}{8} - \frac{a\epsilon}{2} \left(\frac{\{ \{ u \}_{i+\frac{1}{2}} \}_{j \pm \frac{1}{2}}}{4} + \frac{\{ \{ v \}_{i+\frac{1}{2}} \}_{j \pm 1}}{4} \right) \quad \text{etc.} \end{aligned}$$

Notation: Square brackets are differences and curly brackets are sums. For example, $[a]_{i+\frac{1}{2}} = a_{i+1} - a_i$, and $\{ \{ a \} \}_{i \pm \frac{1}{2}} = \{ a \}_{i+\frac{1}{2}} + \{ a \}_{i-\frac{1}{2}} = a_{i+1} + 2a_i + a_{i-1}$. For all details, see Barsukow [2019].

Define then (the divergence in the denominator is not necessary, but seems natural)

$$\rho_{i+\frac{1}{2},j,L} = \frac{\rho_{ij}}{1 + \frac{\rho_{ij}}{2a} \left(\frac{\{ \{ u \}_{i+\frac{1}{2}} \}_{j \pm \frac{1}{2}}}{4} + \frac{\{ \{ v \}_{i+\frac{1}{2}} \}_{j \pm 1}}{4} \right) - \frac{\rho_{ij}}{2a^2} \frac{\{ \{ p \}_{i+\frac{1}{2}} \}_{j \pm \frac{1}{2}}}{4}} \quad E_{i+\frac{1}{2},j,L} = \rho_{i+\frac{1}{2},j,L} \left(\frac{E_{ij}}{\rho_{ij}} - \frac{p_{i+\frac{1}{2},j}^* u_{i+\frac{1}{2},j}^* - p_{ij} u_{ij}}{a} \right) \quad \text{etc.}$$

Finally $f_{i+\frac{1}{2},j}^x := \left(\rho_{i+\frac{1}{2},j,L}^* u_{i+\frac{1}{2},j}^*, \rho_{i+\frac{1}{2},j,L}^* \left(u_{i+\frac{1}{2},j}^* \right)^2, \rho_{i+\frac{1}{2},j,L}^* u_{i+\frac{1}{2},j}^* v_{ij}, E_{i+\frac{1}{2},j,L} u_{i+\frac{1}{2},j}^* \right)$ if $u_{i+\frac{1}{2},j}^* > 0$, etc.

The method is all-speed, i.e. it is able to resolve **both low Mach number flow and shocks** in a stable way. For more details and numerical examples see Barsukow [2021].

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